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ADVANCED DECOY TECHNOLOGY PROGRAM ADTECH IV FINAL REPORT (U)

APPENDIX II USERS INVINUAL—OPTIMUM DECOY DESIGN PROGRAM

Parepared by

AUGO COPERBORNE PRODUCTS GROUP MESSILE SESTERS DEVISION 201 Locall Stroet Wilmington, Massachusetts 01887

AVMSD-0465-68-RR, APP. II Contract F04701-68-C-0012

June 1969

3ponsored by

Advanced Research Projects Agency
Department of Defense
ARPA Order No. 441, Amendment No. 12

JUN 1 8 1969

THIS DOCUMENT IS SUBJECT TO SPECIAL EXPORT CONTROLS AND EACH TRANSMITTAL TO FOREIGN GOVERNMENTS OR FOREIGN NATIONALS MAY BE MADE ONLY WITH PRIOR APPROVAL OF SPACE AND MISSILE SYSTEMS ORGANIZATION (***

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Prepared for

SPACE AND MISSILE SYSTEMS ORGANIZATION
DEPUTY FOR REENTRY SYSTEMS
AIR FORCE SYSTEMS COMMAND
MORTON Air Force Base, California 92409

FOR OFFICIAL USE ONLY

ADVANCED DECOY TECHNOLOGY PROGRAM ADTECH IV FINAL REPORT (U)

APPENDIX II USERS MANUAL--OPTIMUM DECOY DESIGN PROGRAM

Prepared by

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AVMSD-0465-68-RR, APP. II Contract F04701-68-C-0012

by

C. P. Russell, Jr.

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UNCLASSIFIED ABSTRACT

- (U) This technical report describes analyses and techniques used in the design and evaluation of advanced decoy concepts. The work described addresses both the design of specific penetration aid elements and the formulation of techniques for their evaluation. The three major technical areas covered in this report are:
 - 1. Investigation of a penetration aid technique that degrades the measurement capability of the radar sensor.
 - 2. The design of a computer program to solve the decoy design problem with flexibility in the selection of optimization criteria and constraints.
 - 3. Studies of the use of certain discrimination techniques for a hard point defense system.

This appendix to this report contains the input-output information for the optimum decoy design program.

TABLE OF CONTENTS

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1.0	Intr	oduction
2.0	Disc	sussion of Input Techniques
	2.1	Input Interrelationships II-2
	2.2	Special Restrictions on the Input II-4
	2.3	Stacked-Case Ground Rules
	2.4	Input Aids
3.0	Defi	nition of Input Symbols II-1
4.0	D1sc	ription of the Input Sheets
5.0	Disc	ription of the Sample Problem Inputs II-4
6.0	Desc	ription of the Output
	6.1	Trajectory Printout II-45
	6.2	Plotter Output
,	6.3	Classic Check Case Printout
		APPENDIX
	1	Master Input Sheets
	2	Input Sheets for Sample Problems II-103
	3	Printout from Sample Problems II-119
	4	Plots from Sample Problems 11-261

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1.0 Introduction

This manual for the ADTECH Optimum Decoy Design Program is designed for immediate reference by the person actually filling out the input sheets for the program. It contains discussions descriving the input interrelationships and a number of tables which are useful to the user. Each of the input symbols are defined in detail and suitable notes and comments are included along with the definitions. Blank input sheets are included which may be reproduced for production use of the program. Input sheets and a listing of the actual input card images are included for one long check problem and seven short ones. The total printout for the seven short problems and selected critical parts of the printout for the long problem are included along with a detailed description of the output. The plots produced by these check cases are also presented. These check problems and their descriptions provide a means for testing the operation of the program at other installations.

Flow charts of the subroutines of the overall program are shown in Figures 1 and 2 for reference.

2.0 Discussion of Input Techniques

The primary use of the Optimum Decoy Design Program involves the comparison of calculated deccy performance data with stored reentry vehicle data and subsequent adjustment of the decoy design to improve or optimize its performance. Secondary uses of the program include the evaluation and comparison of a single decoy with a reentry vehicle, calculation of the trajectory of a single object without comparisons, calculation of drag coefficients for flight or wind tunnel conditions without trajectory calculations, and classic check problems for research in optimization methodology. Since the requirements for decoy performance vary with mission objectives, considerable flexibility in stating the problem has been coded into the program. The requirements for flexibility lead to a significant number of input options and input quantities in the program; however, for current use, a large number of quantities have been "preset" to help relieve the user of the need to enter the same data over and over. A word of caution is necessary, however, to indicate that the user has the ultimate responsibility for judging whether the preset values of the input are suitable for his particular problem.

2.1 Input Interrelationships

Many of the input quantities are not used in a typical calculation. It is helpful to the user if he can concentrate his attention on those inputs which will be used. Figure 3 shows an Avco invented device for presenting the interrelationships between the inputs. This presentation allows the user to identify the input quantities which actually need attention during the preparation of the inputs.

It is essential that the user have an unambiguous and complete statement of his objectives and requirements clearly in mind before he II-2

prepares the input or attempts to use Figure 3. One of the most common problems with a large program comes about when the user attempts to change the objectives of his calculations after the preparation of the input sheets has been started. The wisdom of not changing horses in mid-stream applies to the preparation of these inputs.

Figure 3 is intended to be used in the following manner. With a clear statement of objectives in mind, the user starts at the top of the table and looks up the definitions of the quantities which are listed there in the Definition of Input Symbols section of this report. As values of the option codes are determined, associated symbols listed under these values in Figure 3 must be looked up and their .lues assigned. These associated symbols are listed only for the user's consideration and do not guarantee that the symbol will actually be used. In some cases the listed symbols may contain a few symbols which are not actually being used in the option being considered but were added to simplify the construction of the table. Tracing out a given problem will result in a path down through this table which provides a "road-map" of the inputs which should be reviewed for that problem. This path does not cross any vertical lines. A symbol with an equal sign indicates that the symbol should be input to have the indicated value. With the significant input symbols identified, the input sheets may be used to record the values of the desired inputs. Depending on the options selected, some input sheets will not be used at all.

For all problems which involve the comparison of a reentry vehicle with one or more decoys, at least two "cases" are required. A "case" is defined as the input quantities appearing before a "transfer card". The transfer card (See Sample Inputs) has a "1" in the first column to indicate that the program is to stop reading input data and start calculating. The first of the required cases accomplishes the process of storing the reentry vehicle data for later comparison (IREF = 1 or 3). The second case, with IREF = 2, involves the calculation of the performance of the decoys and their comparison with the stored reentry vehicle data. The influence coefficient calculations of the MØDE = 2 option require a third case to define the perturbation effects and influence coefficients. Single trajectory calculations, drag calculations, and classic check calculations require only one case. It is generally necessary to trace a path through Figure 3 for each case being submitted.

2.2 Special Restrictions on the Input

The drag calculations in this program were designed to apply to the following body parameters and flight conditions:

	PARAMETER	erimary rance	SECONDARY RANGE
1.	Cone half angle, 0, deg.	4. to 27.	4 to 40. degrees
2.	Vehicle length, inches	12. to 168.	3. to 168.
3-	Surface Temperature, 04	1000.to6000.	er tit besom utlangeb
4.	Bluntness ratio, RN/RB	0.0 to 0.6	:#000=*(annotes-pa-
5.	Altitude, ft.	0. to 400K	· «TELEGRAPHICALISM
6.	Free-stream Mach number	5. to 30.	
7.	Angle of Attack at 300K, deg	0.te 20. 0.te 9	***************************************
8.	Flight Path Angle, deg.	-90 to 0.0	etanarentanaren
Mair			

The primary range represents the region of most accurate calculations, while the secondary range shows the region where the program will produce results with perhaps fegraded accuracy. If the bluntness ratio changes from below 0.6 to above 0.6 during a trajectory calculation as the result of nose shape change, the calculations will be terminated. If the Mach number becomes less than 5.0 the calculations will also be terminated.

Provisions are made in the program to bypass the trajectory calculations temporarily if an optimizer attempts to evaluate a decoy outside the limits shown in Table 1. These provisions assume that the starting configuration (See OVECT, ALOW, UP, etc.) are within the limits shown in Table 1. The optimization processes may go unstable if this restriction is not complied with.

The restrictions on the number of entries in the input tables are indicated in the Definition of Input Symbols and in some cases on the input sheets. The restrictions on the order of the independent variable(s) for the tables are difficult to generalize. In many cases the table-lock-up subroutines are designed to interpolate the data in either ascending or descending monotonic order; however, this has not been confirmed by checkmuns. The preset deck and the sample check cases illustrate the conventional order for most of the tables. The remaining tables, unless otherwise noted in the Definition of Input Symbols, should be input in the order that the data is used in the calculations, independent of whether the independent variable is increasing or decreasing. Note, particularly that the definitions of the independent variables for the thrust variables, Independent or the independent variables.

These definitions allow a given thrust profile to be moved up or down the trajectory, by inputting different initiation parameters, without having to transform the independent variables for each case. The corridor altitudes, H, must start and stop at the same altitudes as the calculations, although the intermediate values do not have to correspond to the printout altitudes. The trajectory input entries in the ZPIØT table must correspond to all the printout altitudes.

Note that the heatshield material is described separately for the trajectory calculations and the wake calculations. There is no internal check on the consistency of the two sets of inputs.

2.3 Stacked Case Groundrules

A series of cases in a job are said to be "stacked". The inputs for cases stacked after the first need special consideration. This program has been designed to retain the most current input for use at the beginning of each stacked case (with the exception of the large table, HH). This means that once an input quantity has been input, it does not have to be input again in following cases if the same value is desired. The preset values listed in the Definition of Input Symbols do not apply to quantities which have been input in previous cases of the current job. Considerable coding effort has been required to maintain this system of stacking cases. As a general rule, the program is not allowed to permanently redefine the value of any input quantity. In cases where it is desirable to redefine an input quantity, the input value is stored, the quantity redefined temporarily, and then the stored value is put back for use in the next case, if no new input supersedes. This technque must be understood when interpretting "core dumps". The HH matrix was judged

too large to save, thus it is an exception to the above discussion.

If Davidon techniques using the input HH matrix are to be stacked, then
the HH matrix must be input for every case or else the final HH matrix
from the previous case will be used.

Trajectories must not be stacked after classic checkruns (ICOM(1) = 1) unless the values of the A vector which have been input for the classic check cases are reinput equal to the values in subroutine ZPRS.

Note that only the first IN values of the initial configuration table ØVECT are used. If IN is smaller in one case than it was in the previous case, care must be taken to enter the values of the remaining design variables under their own symbol names.

There are no serious restrictions on the options which may be stacked as long as the input is complete. The last of a series of reference reentry vehicle cases will be used for comparison with subsequent decoys. In the MØDE = 2 option where three cases are required, the last of a series of reference reentry vehicles and the last of a series of basic decoys will be used for comparison with the perturbation decoys. However, it is not considered wise to run perturbation decoys immediately after a reference reentry vehicle without at least one basic decoy in between.

Within a case, if a symbol is entered more than once, the last value entered will be used in the calculation.

2.4 Input Aids

The definitions of the TØP input matrix are contained in Table 2. The eight operations are listed across the top and the rine performance parameters are listed down the side. A value of zero deletes the operation and a value of one executes the operation if possible. Note that numbers 1 to 21 are preset to one and the others to zero.

The current list of design variables and design variable constraints is shown in Table 3. Although ZTURN is a design variable, it is not considered suitable for automatic optimization since it is only tested at printout altitudes and thus may not provide a continuous penalty function. It can of course be a parameter in a parametric study.

The storage locations (ØCCUR codes) for the calculated quantities which may need to be constrained are presented in Table 4. Up to 20 of these code numbers may be entered in the IDC constraint list to become part of the penalty equation.

The use of probability of discrimination (code 3962) can involve numerical troubles if the value of the probability of discrimination is very near zero or very near one. The error function used to define probability of discrimination (subroutine EFFECT) is coded to produce roughly 16 significant digits. Whenever more digits are needed the probability is set to either zero or one. This precludes any trend information for use by the optimizers and can lead to unstable results. An alternative is provided whereby the "difference in the means, "" (code 3965) can be constrained or minimized instead of the probability of discrimination and thus avoid the significant digits problem.

For searches in a narrow region where the range of values of the probability of discrimination are known in advance, either approach can be used.

Some of the possible input symbols are not used for typical operational problems. Table 5 presents a list of symbols which are not included on the input sheets.

The Euler angle sometem and the thrust orientation definitions are shown in Figures 4 and 5.

3.0 DEFINITION OF INPUT SYMBOLS

INPUT SYMBOLS

1819 (1911) Принципринци

SYMBOL	PRESET	COMMON	REMARKS
A(1-514)	ZPRS	301	Curve-fit constants used primarily in the traject- ory calculations. If $ICOM(1) = 1$, the first 26 can be used in CLASSC.
AA(1-3)	0.,0.,1.	CPCCUR	Coefficients of the altitude weighting function for the velocity corridor integral.
AA(4-6)	0.,0.,1.	CPCCUR	Coefficients of the altitude weighting function for the deceleration corridor integral.
AA(7+9)	0.,0.,1.	CPCCUR	Coefficients of the altitude weighting function for the ballistic coefficient corridor function.
AA(10-12)	0.,0.,1.	CPCCUR	Coefficients of the altitude weighting function for the first wake length corridor function.
AA(13-15)	0.,0.,1.	CPCCUR	Coefficients of the altitude weighting function for the second wake length corridor function.
AA(16-18)	0.,0.,1.	CPCCUR	Coefficients of the altitude weighting function for the third wake length corridor function.
AA(19-21)	0.,0.,1.	CPCCUR	Coefficients of the altitude weighting function for the first wake RCS corridor function.
AA(22-24)	0.,0.,1.	CPCCUR	Coefficients of the altitude weighting function for the second wake RCS corridor function.
AA(25-27)	0.,0.,1.	CPCCUR	Coefficients of the altitude weighting function for the third wake RCS corridor function.
ACØE(1-140)	0.0	CPCCUR	Polynomial coefficients used in MISC to define either free space radar cross section of the decoys as functions of three design variables or to compute any constraint which can be expressed as a polynomial function of up to three variables. The orders of the polynomial are controlled by ICOM (4-6).
acøn	1.0	NIMPUT	Exponent for scale factor, (CCØN) ACON, on trans- ition electron density, N _{et} , FLOWF.

SIMBOLS	PRESET	COMMON	REMARKS
AE	0.0	214	Thruster nozzle exit area for back pressure correction, ft.
AKW	50.0	NIMPOT	Heatshield conductivity for wake calculations, BTU/(ft-CR-hr).
ALPTBL (1-75)	0.0	3646	Input angle of attack table for use in drag calculations if INALPH is greater than zero, degrees.
AI&W(1-20)	0.0	Mad	Lower limits for independent (or design) variables in Fibonacci searches.
ALST	0.2	122	Value of the envelope angle of attack for switching from a rotational to a particle trajectory, degrees.
AMULA (1-20)	0.0	MIN	Multipliers for each term of the penalty equation. These should be selected to avoid very large or very small penalty terms. They may be used to "weight" various penalty terms if desired. These numbers influence the choice of ERR when IPROC is 3.
AWREIT	0.0	188	Reference area for the drag components in the table WCDTAB, ft2.
B(1-21)	ZPRS	823	Curve-fit coefficients in the trajectory calculations. These numbers are not normally input; however, it is possible to do so for research or debugging purposes.
BCB(1-40)	0.0	CPCCUR	Lower corridor limits for ballistic coefficient differences lb/ft2.
BOØN	1.0	nimpur	Exponent for scale factor, (CCØN) on the decay rate, b1, in FIØWF.
BCD(1-40)	0.0	CIPCCUR	lower corridor limits for deceleration differences,
BCV(1-40)	0.0	CPCCUR	Lower corridor limits for velocity differences, ft/sec.
BCWL1 (1-40)	0.0	CPCCUR	lower corridor limits for the first wake length differences, meters.
BCMT5 (1-70)	0.0	CPCCUR	Lower corridor limits for the second wake length differences, meters.

SYMBOL	PRESET	COMMON	REMARKS
BCWL3(1-40)	0.0	CPCCUR	Lower corridor limits for the third wake length differences, meters.
BCWR1(1-40)	0.0	CPCCUR	Lower corridor limits for the first wake RCS differences, m or db. depending on IDBL.
BCWR2(1-40)	0.0	CPCCUR	Lower corridor limits for the second wake RCS differences, m or db. depending on IDBL.
BCWR3(1-44)	0.0	CPCCUR	Lower corridor limits for the third wake RCS differences, m2 or db. depending on IDBL.
BETALL	0.0	152	Sublimation rate coefficient for initial heat- shield material if MATIN1 is 6, ft/sec - OR.
BETA12	0.0	171	Sublimation rate coefficient for heatshield material after ZTURN if MATIN2 is 6, ft/sec - OR.
BETA21	0.0	153	Sublimation rate coefficient for initial heat- shield material if MATINI is 6, ft/sec - PRETA31
BETA22	0.0	172	Sublimation rate coefficient for heatshield material after ZTURN if MATIN2 is 6, ft/sec - oRBETA32
BETA31	0.0	154	Order of reaction for initial heatshield material if MATIN1 is 6.
BETA32	0.0	173	Order of reaction for heatshield material after ZTURN if MATIN2 is 6.
BETA41	0.0	155	Activation temperature for initial heatshield material if MATIN1 is 6, or.
BETA42	0.0	174	Activation temperature for heatshield material after ZTURN if MATIN2 is 6, OR.
BETAPL (1-160)	0.0	CPCCUR	Table of ballistic coefficient inputs for reference R/V if IREF is 3, lb/ft2.
BETAZ(1-10)		CWAKE	Atmospheric density scale height for wake calcula- tions. The first two values are preset to 22.0 thousands of feet.

2. Programma o de la composición de la

SMEOT	FRESET	COMMON	REMARKS
BIWEN	1.0	DRCSEC	Scaling constant in RCSEC, b20.
HZERØ	5.8K-21	DRCSEC	Scaling constant in RCSEC, b.
BS	4.0E-10	DRCSEC	Scaling constant in RCSEC, b2.
В3	2.0	DRCSEC	Scaling constant in RCSEC, b.
B ₂ 1	1.0	NIMPUT	Scaling constant in FLØWF, b21.
1855	0.25	NIMPUT	Scaling constant in FIWWF, b22.
№ 3	0.0	NIMPUT	Scaling constant in FLØWF, b23.
B24	1.0E-26	DACSEC	Scaling constant in RCSEC, b24.
C	1.0	115	Multiplier on stagnation point heating in the nose blunting calculations which can be used to simulate a decoy having a nosecap made of a different material from the main body.
(1-50)	0.0	MIN	Lower bounds for constrained items in the penalty equation.
CAPG	32.21852	19	Graviational acceleration, ft/sec2.
CASE	0.0	128	Case number. Note that 0.001 will be added internally to the input number for each trajectory calculated when MØDE is 2 or 3.
COÓN	1.0	3963	Scale factor for transition electron density and decay rate in the wake calculations in FLØWF. This scale factor may be used as a design variable to roughly simulate wake-seeding concepts.
CDØWN (1-16)	1.0E-5	3549	Lower limits on accuracy of integrated quantities in the equations of motion. If the absolute value of the quantity being integrated is less than or equal to 1, the limits are equal to the inputs with the units listed below. If the absolute value is greater than 1, the limits are equal to the inputs (non-dimensional) times the absolute value of the quantity being integrated.
			1 Velocity, ft/sec.
p. f 1,77	SE VILL		2 Flight path angle, rad.
	1-1-		3 Time, sec.

CMPOT	DOCCOM	COMMON	REMARKS
SYMBOL	PRESET	COMMON	
CDØWN (1-16)	1.0E-5	3549	4 Range (downrange), ft.
7.00			5 Initial weight minus weight ablated, 1b.
			6 Nose radius, ft.
			7 Base radius, ft.
			8 Euler angle in yaw, rad.
			9 Euler angle in pitch, rad.
			10 Euler angle in roll, rad.
			11 Pitch rate, rad/sec.
			12 Yew rate, rad/sec.
			13 Roll rate, rad/sec.
			14 Side range, ft.
			15 Thrust direction, rad.
			16 Initial weight minus weight expelled by thruster, 1b.
CDTAB (1-75)	0.0	3383	Table of drag coefficients which will supercede the calculated drag coefficients if MAKCD is greater than zero.
снісн (1-16)	1.0E-4	3533	Upper bounds on accuracy of integrated quantities in the equations of motion. See CDOWN for identi- fication of the 16 items.
CMQIN1	0.0	124	Input pitch damping coefficient for initial configuration if IKONQ is 1.
CWOINS	0.0	125	Input pitch damping coefficient for configuration after ZTURN if IKCNQ is 1.

SYMBOL	PRESET	COMMON	REMARKS
CNE	0,0	DRCSEC	Transition electron density when non-linear production terms are considered in the turbulent wake. The item should be left at 0.0 in this model.
CNUMB (1-169)	-	NIMPUT	General constants used in the wake calculations. The quantities currently being utilized are preset in the preset deck.
CPG1	0.0	161	Specific heat of gas for initial heatshield material if MATIN1 is 6, BTU/lb R.
CPG2	0.0	180	Specific heat of gas for heatshield material after ZTURN if MATIN2 is 6, BTU/lb R.
CP21	0.0	160	Specific heat of solid for initial heatshield material if MATINI is 6, BTU/lbOR.
ass	0.0	179	Specific heat of solid for heatshigld material after ZTURN if MATIN2 is 6, BTU/lb R.
CREED	0.75	NIMPUT	Heatshield specific heat used in the wake calculations, BTU/1b R.
CTP(1-20)	0.0	MIIN	Upper bounds for constrained items in the penalty equation.
DATE	0.0	127	Date, for example: 814.68 means August 14, 1968.
DRIHC1	0.0	166	Heat of decomposition for initial heatshield material if MATINI is 6, BTU/lb.
DELEGS	0.0	185	Heat of decomposition for heatshield material after ZTURN 11 MATIN2 1s 6, BTU/lb.
DEUTN	-2000.	187	Maximum integration interval (altitude) for the initial integrations of the equations of motion,ft.
DELINU)	0.0	159	Difference between the virgin and char density of the initial heatshield material if MATIN1 is 6, lb/ft ³ .
DETASS.	0.0	178	Difference between the virgin and char density of the heatshield material after ZTURN if MATIN2 is 6, lb/ft ³ .

SYMBOL	PRESET	COMMON	REMARKS
DELTA	1.0	føpt	Estimate of the determinant of the initial H matrix (input as HH) in the Davidor method. If FAC is not zero, then input DELTA = HAC input is for printout only and does not affect the optimization process.
DELWH	0.01	nimput	Heatshield thickness for wake calculations, in.
DELX (1-20)	0.001	dørt	In the Davidon Method, finite difference increments; in the Rosenbrock Method, the initial step sizes. For use with the Davidon Method, these terms should be smaller than the anticipated minimum step size but large enough to obtain a meaningful gratient.
DELY	0.0	219	Linear component of thrust offset in the Y direction, in.
DELZ	0.0	550	Linear component of thrust offset in the Z direction, in.
DHCHEM	0.0	HIMPUT	Chemical enthalpy of the heatshield for the wake calculations, ft2/sec 2.
DNENDZ	0.0	248	lower altitude limit for tabular input atmosphere, ft.
DSB	0.0	DRCSEC	Additional wake relar cross section due to con- sideration of non-linear production terms in the turbulent wake. This item should be left at 0.0 in this model.
DTABL (11,5,4)	PRESET	TBLS12	Electron density as a function of normalized enthalpy, ratio of ablation to boundary layer air, and air density for 1000 PPM sodium seed for wake calculations, e/cm.
DVH(1-50)	0.0	CPCCUR	Input values of design variables for second perturba- tions of comparison decoys (MODE = 2).
DVL(1-50)	0.0	CPCCUR	Input values of design variables for first perturba- tions of comparison decoys (MODE = 2).
DX	50.	DRCSEC	Numerical step size used in finding the wake length, meters.

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SYMBOL	PRESET	COMMON	REMARKS
EMCTBL (1-12)	PRESET	TBLS12	Cone Mach number, independent variable for wake Mach number table, ETABL.
ENTABL (25,9)	PRESET	TBLS12	Table of electron density as a function of normalized enthalpy, HSTABL, and air density, RSTABL.
EPSILL	0.0	167	Coefficient of emission for initial heatshield material if MATIN1 is 6.
Epairs	0.0	186	Coefficient of emission for heatshield material after ZTURN if MATLN2 is 6.
ERNRTB (1-10)	PRESET	TBIS12	Air density, independent variable for normal shock electron density, ERNTBL, lb/ft ³ .
ERWIPL (8,10)	PRESET	TBLS12	Normal shock electron density as function of density, ERNUTB, and velocity, ERNUTB, e/cc.
ERNUTB (1-8)	PRESET	TBIS12	Velocity, independent variable for normal shock electron density table, ERNTBL, thousands of feet per second.
ERR	0.01	FØPT	If IPROC = 3 (Davidon), EnR is the stopping tolerance on the transformed gradient; if IPROC = 1 or 5 and LIMIT = 0, the accuracy requirement for the Fibonacci
			search in the physical units of the independent variables.
ETABL (12,11)	Preset	TBIS12	Mach number as a function of cone Mach number, EMCTBL and cone angle THITTBL.
PAC	1.0	FØPT	If non-zero, multiplier of the identity matrix to establish the initial H matrix in the Davidon Method for each sequential solution - finding operation; if zero, the input HH matrix will be used instead.
Posm	4.0	NALTEG	Multiplier on the maximum step size limit, f/gs, in the Davidon method.
PHQL	4.3528	CWAKE	First radar frequency for wake calculations, cps.
PRQ2	1.375E9	CWAKE	Second radar frequency for wake calculations, cps.
PRQ3	5,489	CWAKE	Third radar frequency for wake calculations, cps.
THE OWNER OF STREET ASSESSMENT			

SYMBOL	PRESET	COMMON	REMARKS
F1.	0.0	157	Heat of ablation for the initial heatshield material if MATLN1 is 6, BTV/lb.
F2	0.0	176	Heat of ablation for the heatshield material after ZTURN 1f MATIN2 is 6, BTU/lb.
G	32.174	27	Conversion factor, lb/slug.
GAMFO	0.0	105	Initial flight path angle (Note that this quantity must be input as a negative number to obtain mean-ingful results), degrees.
GAMMA	1.4	28	Ratio of specific heats for air.
H (in col- umn 1)		******	Heading information, can be in any format which can be keypunched.
H(1-40)	0.0	CPCCUR	Altitudes for the corridor tables and rader measurement errors, BCV, TCV, etc. and SV, SD, SB, etc. The initial value in this table should be equal to ZO and the NCPth value should be equal to ZST. The other (NCP-2) values must be monotonic but do not have to correspond in any way with the printout altitudes.
HH (40,40)	-	BLKØ	Upper right triangular input of the initial elements of the Davidon H matrix if FAC is zero. Note that this input is modified by the Davidon process and that the input values are not saved. This means that the modified matrix is carried over to the next sequential solution-finding operation or to the next case if FAC is zero. Only the first IN rows and columns are used.
HREF1	0.0	156	Combustion ablation constant for initial heatshield material if MATIN1 is 6.
HREF2	0.0	175	Combustion ablation constant for heatshield material after ZTURN if MATIN2 is 6.
HSTABL (1-25)	PRESET	TBLS12	Normalized enthalpy, independent variable for electron density table ENTABL.

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SYMBOL	PRESET	COMMON	KEMA RICS
FTAB(1-75)	0.0	3233	Altitude table, independent variable for either drag coefficients, CDTAB, or angles of attack, ALFTAB, depending on MAXCD and INALFR.
iatmøs	0	nøccur	Input atmosphere option code and indicator for the number of entries in the altitude table, TRATMZ. If IATMØS is zero, the 1962 standard atmosphere will be used.
ICØM(1)	0	IXCØM	Option controlling the use of the full trajectory calculation (0) or the use of the classic check case subroutine (1).
ICQM(S)	0	IXCØM	Not currently used.
ICØM(3)	0	IXCØM	Input code used in REDUCE to distinguish between the cases where an input quantity is being optimized (0) and an output quantity is being optimized (1). For example, if weight is to be minimized, ICØM (3) should be input equal to zero; but if probability of discrimination is to be minimized, then ICØM(3) should be input equal to one.
ICØM(4)	1	IXCØM	One more than the order of the first variable in the polynomial in MISC. If the polynomial is to be quadratic in the first variable, then input $ICOM(4)$ equal to 3.
IOM(5)	1	IXCÓM	One more than the order of the second variable in the polynomial in MISC.
ICØM(6)	1	TXCØM	One more than the order of the third variable in the polynomial in MISC. Note that there are ICOM(4)* ICOM(5)* ICOM(6) coefficients (ACOE) required for the polynomial.
ICØM(7)	136	IXCØM	$\phi_{\rm CCUR}$ subscript identifying the first variable in the polynomial in MISC.
ICMM(8)	135	IXCÓM	$\emptyset CCUR$ subscript identifying the second variable in the polynomial in MISC.
IOM(9)	134	IXCOM	ØCCUR subscript identifying the third variable in the polynomial in MISC.
ICOM(10)	0	тхсфм	Input option control for normal definition of ballistic coefficient, W/C_DA , (0); or for the definition of an apparent ballitic coefficient including thrust, $W/(C_DA - T/\varphi)$, (1), or for the reciprocal of the apparent ballistic coefficient, (2).
		1196	These are experimental options and should only be used cautiously.

SYMBOL	PRESET	COMMON	REMARKS
ІСФМ (11-200)	0	IXCØM	Not currently used.
IDBL	14	CICCUR	Input code controlling the units of the wake radar cross section for output; if 3, RCS in decibels; if 4, RCS in square meters. Note that the corridors and standard deviation inputs must be in compatible units however, the inputs SIGNL1, 2, and 3 are always input in square meters.
IDC(1-20)	0	idnøs	ØCCUR subscripts identifying the terms desired in the penalty equation. The first entry, IDC(1), must be the quantity to be optimized if LRED is greater than zero. See Table 4.
TDC (21-50)	0	idnøs	Not currently used.
IDNØ(1-20)	0	idnøs	ØCCUR subscripts identifying the independent or design variables for the search. See Table 3.
IDNØ (21-50)	0	idnøs	Not currently used.
IEX	2	IØPT	The exponent of the terms in the penalty equation.
IGDH(1-20)	0	IGDHL	OCCUR subscripts defining the first terms in the general differences in subroutine MISC. A zero in this table will terminate the general difference calculations.
IGDL(1-20)	0	IGDHL	ØCCUR subscripts defining the second terms in the general differences in subroutine MISC. A zero in this table will terminate the general difference calculations.
IKCMQ	0	NØCCUR	Option code for pitch damping derivative, C_{m_Q} , use calculated derivative if 0 and input value if greater than 0. This code applies both before and after ZTURN.
IN		IØPT	The number of design variables to be used. This should be 1 for IPRØC = 1 and 2 for IPRØC = 5. It is equal to the number of entries in the IDRØ and ØVECT tables. There must be at least one entry for all problems where MØDE = 3 and IREF = 2.

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SYMBOL	PRESET	COMMON	FEMARKS
INALPH	0	nøccur	Option code for an input angle of attack for use with a particle trajectory (LOPT = 1). If INALPH is 0 and LOPT = 1, zero angle of attack will be used. If the input angle of attack is desired, INALPH must equal the number of entries in the ALPTAB table. Note that if INALPH is greater than zero and LOPT= 0 or 2, then the ALPTAB inputs override and the rotational calculations are not performed. Note that MAXCD and INALPH must not both be greater than zero.
IND	0	CWAKE	Printout option control for wake flowfield calculations: 0, no printout; 1, printout at every altitude point where wake calculations are performed.
INDS	0	DRCSEC	Printout option control for wake radar cross section calculations: O, no printout; l, printout at every altitude point where wake radar cross section calculations are performed.
IØP(1-90)	See below	CICCUR	Option control matrix for various operations for the nine performance functions. The definitions are summarized in Table 2.
I&B(T-51)	1	CICCUR	Controls for trajectory performance functions.
1ØP (22-63)	0	CICCUR	Controls for wake performance functions. Some of these must be set to 1 if wake calculations are desired for the reference reentry vehicle. At least one must be turned on for each of the six performance parameters to be evaluated.
IMP (64-75)	***	CICCUR	These quantities are used internally. No input has any influence.
IØP (76)	_	CICCUR	Plotter option control which must be 1 at Avco and 0 at Aerospace (See subroutine AVPIN for details).
IØP(77-82)	0	CICCUR	Controls for wake plots of the wake parameter (not the difference) versus altitude.
IØP(83-90)	0	CICCUR	Not used.
IPMP	-	IØPT .	Input option for future use in FEV. Not currently used.
IPRØC	1	IØFT	Optimizer selection code:
			1 One-variable Fiboracci search
			2 (One decoy evaluation)
	First Section		3 Davidon Variable Metric Method for Minimization

II-22

SYMBOL	PRESET	COMMON	REMARKS
			4 Rosenbrock Rotating Coordinate Method
			5 Two-variable Fibonacci search
IRAND	0	IØ₽T	Davidon input for number of random starting points to be automatically used. This input should be left at O since the random number subroutine is not implimented and because the use of this option would conflict with the optimization technique if LRED is greater than zero.
IŘEF	1	CICCUR	Trajectory processing option code; 1, calculate R/V trajectory or other miscellaneous calculations; 2, calculate and compare decoy trajectories or use classic check case functions; 3, input set of R/V data for comparison with decoy data.
ISEN1	0	SENSE	Davidon printout control.
ISEN2	0	SENSE	Davidson printout control for future use.
ISP	1.0	282	Specific impulse of the thruster
ITAPE	0	nøccur	Option for a tape output for velocity, ballistic coefficient, and altitude for use as input to other Avco programs; O, no tape output; 1, tape output.
ITHRST	0	nøccur	Number of entries in the thrust table, THTHO. Maximum value is 25.
IWAKE	2	CWAKE	Number of entries in the wake-altitude table, WKALF. Maximum value is 10.
IWPRNT	0	CWAKE	Printout option in WAKE subroutine; 0, no printout; 1, printout at each altitude.
К	0	MINSK	Not currently used.
LAMDA1	0.0	137	Bluntness ratio (RN/RB) for initial configuration.
LAMDA2	0.0	143	Bluntness ratio for configuration after ZPURN.
LA1	0.0	138	Axial length for initial configuration, in.
LA2	0.0	144	Axial length for configuration after ZTURN, in.
IMIT	30	iøpt	Counter limit for the various optimizers: A. IPRØC = 1.

a. LIMIT greater than zero. The number of times
the function will be calculated is LIMIT + 1.
b. LIMIT equal to zero. This input has no effect
and the accuracy requirement, ERR, will control
II-23 the number of calculations of the function.

- B. IPR ϕ C = 2. This input is not used.
- C. IPRØJ = 3. LIMIT is the maximum number of iterations for the Davidon method. An iteration is the total process of selecting a direction, bracketing the minimum in that direction, and locating that minimum.
- D. IPR ϕ C = 4. Maximum number of successful steps allowed along any one coordinate.
- E. IPRØC = 5
 - a. IJMIT greater than zero. The number of times the function will be calculated is $(LIMIT)^2 +1$.
 - b. LIMIT equal to zero. This input has no effect and the accuracy requirement, ERR, will control the number of calculations of the function for each variable.

L∲PT	1	nøccur	Trajectory option cole: O, rotational trajectory; 1, particle trajectory; 2, simplified angle of angle attack trajectory; 3, input flight conditions for drag calculation; 4, input wind tunnel conditions for drag calculations.
LPL Ø T	1	CICCUR	Number of entries in the table ZPLØT of R/V input data. This number must agree with the number of altitude points produced in the decoy calculations by ZPR1, ZBAR, ZPR2, ZST, and TST. Maximum value is 160.
LÆD	0	ØWL	Maximum number of times that the factor WRF can be applied in the optimization process.
MATIN 1	1	nøccur	Code for initial heatshield material: 1 Teflon 2 In. 3 OTWR 4 Phenolic Nylon 5 Carbon Phenolic 6 Use input material properties
MATLNS	1	nøccur	Code for heatshield material after ZTURN. See MATINI list above for definitions.

SYMBOL	PRESET	COMMON	REMARKS
MAXCD	0	nøccur	Code for drag options in the trajectory calculation; O, calculate the drag and do not use the drag input in CDTAB, greater than zero, not calculate the drag but use the drag input in CDTAB plus perhaps the drag in WCDTAB depending on MAXWCD. If MAXCD is greater than zero, it must be equal to the number of entries in CDTAB. Maximum value is 75. Note that MAXCD and INALPH must not both be greater than zero.
AXVAL.	0	NØCCUR	Number of entries in the TRAJT table if LAPT is 3 or the number of entries in the WTZ table if LAPT is 4. Maximum value is 75.
MAXWCD	0	nøccur	Number of entries in the table for added drag coefficient values, WCDTAB. If zero, no drag will be added. Maximum value is 75.
memø	0.0	129	Memo number which can be used to identify the job, example: 1032.4.
MHEAT	0	nøccur	Heating and mass loss code: O, Aerodynamic heating only (if MCPT = 1) 1, Heating and mass loss (if MCPT = 1)
мǿDЕ	3	CICCUR	Fundamental option code which allows direct access to certain subroutines to simulate existing Avco programs.
			l Single trajectory calculations, or drag calculations without trajectories (simulates Avco program 2269)
			2 Comparisons of R/V and decoy performance and influence coefficients.
			3 Maximum capability with optimization searches.
мØРТ	0	nøccur	Heating and mass loss code: O No heating or mass loss 1 Test MHEAT
MW	28.9	117	Molecular weight of air, gram/mole.
MXTAB1	1	NØCCUR	Number of entries in mass properties table, TABZ1. Maximum value is 50.
MXTAB2	1	NØCCUR	Number of entries in mass properties table, TABZ2. Maximum value is 50.

SAMPOL	PRESET	COMMON	REMARKS
NALT	0	NALTEG	Code for alternate step size logic in Davidon method (subroutine READY): O Select a new direction after an undershoot (normal) l Double the step size and continue in the same direction after an undershoot (alternate).
ncømdv (1-50)	133	CICCUR	OCCUR subscripts identifying the design variables to be perturbed during the influence coefficient calculations of MODE = 2 option. The same subscripts may be entered more than once if desired.
ncøns	1	IØPT	Number of entries in the constraint table, IDC. This is also the number of terms in the penalty equation. Maximum value of 20.
NCP	1	CICCUR	Number of entries in each of the altitude, H, corridor and standard deviation tables. Maximum value of 40.
ndecøy	1	CICCUR	Code controlling the perturbations for influence coefficients of the MØDE = 2 options. NDECØY must be equal to 1 for all other options.
			1 R/V or one basic decoy 2 One perturbation of each design variable 3 Two perturbation of each design variable
NDVCH	1	CICCUR	Number of entries in the design variable table, NCOMDV. Maximum value of 50.
новом	1	NØCCUR	Geometry input code indicating which parameters are being input.
			 Nose radius, base radius, and cone angle Base radius, cone angle, and bluntness ratio Nose radius, base radius, and length
in T			Note that this code applies to both the initial configuration and to the configuration after ZTURN. This input must be compatible with the design variables listed in IDNØ if MØDE is 3 or those listed in NCØMDV if MØDE is 2, IREF is 2, and NDECØY is 2 or 3.
NGL1.	0.0	164	Laminar transpiration factor of gas for the initial heatshield material if MATIN1 is 6.

SYMBOL	PRESET	COMMON	REMARKS
NGL2	0.0	183	Laminar transpiration factor of gas for the heat-
NGT1	0.0	165	Discrete material after ZTURN if MATIN2 is 6.
Total Marco			Turbulent transpiration factor of the gas for the initial heatshield material if MATIN1 is 6.
NGT2	0.0	184	Turbulent transpiration factor of the gas for the heatshield material after ZTURN if MATIN2 is 6.
nøseøp	0	nøccur	Shape-change option for noseblunting and decreasing base radius. O no shape change 1 shape change (if MOPT and MHEAT are 1)
NPA	1	CICCUR	Number of entries in the altitude counter table, NPV, for plots of performance variables versus design variables at fixed altitudes for MØDE = 2 option. Maximum value of 160.
NPLØT (1-5)	****		Plotter codes. These are overridden during the gradient calculations in subroutine FCN to reduce the number of plots produced.
NPLØT(1)	0	NØCCUR	Code for drag coefficient plots:
			O No plots 1 Cp. vs. Z
			1 $^{C_{D_{total}}}$ vs . Z $^{C_{D_{p}}}$ + $^{C_{D_{g}}}$ vs . M
	•	-	C _{DF} ve. Z
NPLØT(2)	0	NØCCUR	Code for the code
		APOODI	Code for trajectory plots: O no plots
			1 BETA vs. Z
			V vs. t
			M vs. t
			Vg ve. t
	0.00	1 - 1	Q _{dym} vs. t

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SYMBOL	PRESET	COMMON	REMARKS
NPLØT(3)	0	NØCCUR	Code for programs and beating allets
	v	Nyocun	Code for pressure and heating plots: O no plots Wotal/Winitial vs. Z Ps vs. t
			H _S /RT _o vs. t
•			q _{stag} vs. t
			q ₈ vs. t
			for A≠ 0: qsonic vs. t qv vs. t
NP1 Ø T(3)	0	NØCCUR	Code for envelope angle of attack versus time plots: 0, no plots; 1, plot.
npiøt(5)	•	nøccur	Code for a generalized plot which allows any quantity in the ØCCUR array to be stored at each printout altitude and plotted versus time. Note that the use of this option requires some
			understanding of the arrangement and units of the data during the execution of the trajectory calculations. An input of 0 indicates no plot and an input of the appropriate OCCUR subscript indicates that a plot should be made.
ndrint	1	nøccur	Printout control for detailed trajectory data. O No detailed trajectory printout except for solutions and final decoys. 1 Detailed trajectory printout for every trajectory.
MPV(1-160)	1	CICCUR	I Detailed trajectory printout for every trajectory Index of altitudes for plots of performance variables versus design variables in MØDE = 2 option. This index is a list of numbers identifying the altitudes at which plots are desired. The altitudes are numbered according to the order in which they are printed out. The initial altitude (Z ₀) is 1, the next printout altitude (Z ₀ - ZPR) is 2, etc.
nsl)	0.0	162	Laminar transpiration factor of solid for initial heatshield material if MATINI is 6.
NSL2	0.0	181 .	Laminar transpiration factor of solid for heatshield material after ZTURN if MATIN2 is 6.
KSTVL	100	DRCSEC	Maximum number of rough sizing steps allowed in the wake length calculations.
ACTION STATE	147	1 1111	11+28

SYMBOL	PRESET	COMMON	REMA PKB
nsl.	0,0	163	Turbulent transpiration factor of solid for initial heatshield material if MATINI is 6.
NST?	0.0	182	Turbulent transpiration factor of solid for heat- shield material after ZTURN if MATINZ is 6
NTHRUST	0	NØCCU R	Thrusting option code:
٠. ا			O no thrust 1 thrust as a function of delta altitude 2 thrust as a function of delta time
ØCCUR (1-4000)	1980	1	Input symbol allowing input directly to the OCCUR array for research or debugging purposes.
ØVECT (1-20)	5.0	øwi.	Starting values of the design variables for IPFOC equal to 2,3, or 4. This defines the first configuration for the search processes. Note that the Fibonacci searches do not use this input. Units must be compatible with the normal input units.
PFD	0.03	CPCCUR	Probability of false dismissal of the reentry vehicle as a decoy. Also called & or Pp.
PHIO	0.0	1115	Initial value of roll Euler angle, degrees. (see figure 4.)
PHI1(1-10)	6.,6.	CWAKE	Look angle for radar of the first frequency, degrees.
PHI2(1-10)	6.,6.	CWAKE	Look angle for radar of the second frequency, degrees.
PHI3(1-10)	6.,6.	CWAKE	Look angle for redar of the third frequency, degrees.
PRAND	0.0	føpt	Random step size control for Davidon technique. This input is not used as long as IRAND is C.
PSIZET	0.0	223	First thrust offset angle (in X-Y plane, positive for right hand rotation), degrees (see Figure 5)
PSIO	0.0	114	Initial value of yaw Euler angle, degrees. (see Figure 4).
PO	0.0	109	Initial angular rate in roll, rad/sec.
କ 0	0.0	110	Initial angular rate in pitch, rad/sec.
R	53.5	57	Gas constant for air, ftlb./lbm-OR.
RBJ.	0.0	136	Initial base radius for the initial configuration, inches.

	©AWBOT!	PRESET	COMMON	REMARKS
	RB2	0.0	142	Initial base radius for the configuration after ZTURN, inches.
	FE	2.09022927	63	Radius of the earth, feet.
	FHØ21	0.0	158	Char density for the initial heatshield material if MATIN1 is 6, $1b/ft^3$.
	HHØ23	0.0	177	Char density for the heatshield material after ZTURN if MATIN2 is 6, lb/ft3.
	RHØSL	0.08042	NIMPUT	Sea level density for exponential atmosphere approximations in the wake calculations, lb/ft3.
	RHØW	115.0	NIMPUT	Heatshield density for wake calculations, lb/ft3.
	RN1	0.0	135	Initial nose radius for the initial configuration, inches.
	ENS	0.0	141	Initial nose radius for the configuration immediately after ZTURN, inches.
	RSTABL(1-9)	PRESET	TBLS12	Normalized density, independent variable for electron density table, ENTABL.
	RTO	8.47525	NIMPUT	Reference enthalpy for wake calculations, ft2/sec2.
1 1 1 1 1 1 1 1 1 1	SB(1-40)	0.0	CPCCUR	Standard deviation of radar errors for ballistic coefficient, lb/ft ² . This table of NCP values corresponds to the altitude table H.
	SD(1-40)	0.0	CPCCUR	Standard deviation of radar errors for deceleration, g's. This table of NCP values corresponds to the altitude Table H.
N 5	SIG	3.5	116	Collision cross section for air, angstroms.
	SIGNL1	1.0E-6	CWAKE	Noise level for wake length definition at first frequency, m2.
	BIGNIA	1.0E-6	CWAKE	Noise level for wake length definition at second frequency, m2.
Jan 3 Section	STGNL3	1.0E-6	CWAKE	Noise level for wake length definition at third frequency, m2.
-	SMRO			Initial angular rate in roll, rad/sec.
	300.1 (1-25)	1.0	MULIP	Multipliers of special penalty terms in subroutine SCREEN formearch and debugging purposes.

SYMBOL	PRESET	COMMON	REMARKS
SRS(1)	40.0	CPCCUR	Number of smoothed radar measurements of velocity.
SRS(2)	40.0	CPCCUR	Number of smoothed radar measurements of deceleration
SRS(3)	40.0	CPCCUR	Number of smoothed radar measurements of ballistic coefficient.
SRS(4)	40.0	CPCCUR	Number of smoothed radar measurements of wake length at 1st frequency.
SRS(5)	40.0	CPCCUR	Number of smoothed radar measurements of wake length at 2nd frequency.
srs(6)	40.0	CPCCUR	Number of smoothed radar measurements of wake length at 3rd frequency.
SRS(7)	40.0	CPCCUR	Number of smoothed radar measurements of wake RCS at 1st frequency.
SRS(8)	40.0	CPCCUR	Number of smoothed rader measurements of wake RCS at 2nd frequency.
SRS(9)	40.0	CPCCUR	Number of smoothed radar measurements of wake RCS at 3rd frequency.
sv(1-40)	0.0	CPCCUR	Standard deviation of radar errors for velocity, ft/sec. This table of NGP values corresponds to the altitude table H.
SWL1(1-40)	0.0	(PCCUR	Standard deviation of radar errors for wake length at the first frequency, m. (See H, NCP).
SWI2(1-40)	0.0	CPCCUR	Standard deviation of radar errors for wake length at the second frequency, m. (See H, NCP)
SWL3(1-40)	0.0	CPCCUR	Standard deviation of radar errors for wake length at the third frequency, m. (See H, NCP).
SWR1(1-40)	0.0	CPCCUR	Standard deviation of radar errors for wake RCS at the first frequency, units depend on IDBL. (See H, NCP).
swr2 (1-40)	0.0	OPCOUR	Standard deviation of radar errors for wake RCS at the second frequency, units depend on IDBL. (See H, NCP).
swR3(1-40)	0.0	CPCCUR	Standard deviation of radar errors for wake RCS at the third frequency, units depend on IDBL. (See H, NCP)

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SYMBOL	PRESET	in A graph and a binder and a properties and a propertie	SHIP CONTROL OF THE C					
TABIX1 (1-50)	1.0,0.0	3033	Table of roll moments of inertia, for initial shape, slug - ft2.					
TABIX2 (1-50)	1.0,0.0	3083	Table of roll moments of inertia for the configuration after ZTURN, slug - ft2.					
TABI1(1-50)	1.0,0.0	2933	Table of pitch-yaw moments of inertia for initial shape, slug - ft ² .					
TĀBI2(1-50)	1.0,0.0	2983	Table of pitch-yaw moments of inertia for the configuration after ZTURN, slug - ft2.					
TABL	1500.0	NIMPUT	Ablation temperature (OK) for the heatshield in the wake calculations. Note the difference in units from TW1, TWST, TINIT, etc.					
TABRHØ (1-50)	0.0	3771	Tabular input freestream density, lb/ft3.					
TABSND (1-50)	0.0	3821	Tabular input freestream speed of sound, ft/sec.					
TABZ1 (1-50)	0.0	3133	Altitudes for mass properties tables for initial shape, ft.					
TABZ2 (1-50)	0.0	3183	Altitudes for mass properties tables for the configuration after ZTURN, ft.					
TAUL	1.0	CWAKE	Pulse length for radar of first frequency, Assec.					
TAU2	0.4	CWAKE	Pulse length for radar of second frequency, M sec.					
TAU3	0.4	CWAKE	Pulse length for radar of third frequency, μ sec.					
TBATMZ (1-50)	0.0	3721	Table of atmosphere altitudes for use with TABRHØ and TABSND. This table must be input with the lowest (smallest) altitude first.					
TCB(1-40)	0.0	CPCCUR	Upper corridor for differences in ballistic coefficient in lb/ft ² . Note that the NCP values in this table correspond to the altitudes, H. The sign convention for the corridors is a source of					
			possible confusion. The difference itself is defined as the R/V parameter minus the decoy parameter. Thus a positive difference (upper side of corridor) implies that the R/V parameter is larger than the decoy parameter. However, if the R/V and decoy parameters were plotted, the decoy parameters would then be below the R/V, while on the difference plots the decoy data would be above the axis. The upper corridor is defined as the R/V parameter minus the minimum allowable value for the decoy. II-32					

SYMBOL	PRESET	COMMON	REMARKS
TCD(1-40)	0.0	CPCCUR	Upper corridor for differences in deceleration, g's. (See TCB).
TCRIT	0.0	77	Angle of attack cycle time test parameter, sec. Fecommended value is 1.0E-5.
TCV(1-40)	0.0	CPCCUR	Upper corridor for differences in velocity, ft/sec. (See TCB).
TCWL1(1-40)	0.0	CPCCUR	Upper corridor for difference in wake length at the first frequency, meters. (See TCB).
TCWL2(1-40)	0.0	CPCCUR	Upper corridor for differences in wake length at the second frequency, meters. (See TCB).
TCWL3(1-40)	0.0	CPCCUR	Upper corridor for differences in wake length at the third frequency, meters. (See TCB).
TCWR1(1-40)	0.0	CPCCUR	Upper corridor for differences in wake RCS at the first frequency. Units depend on IDBL (See TCB).
TCWR2(1-40)	0.0	CPCCUR	Upper corridor for differences in wake RCS at the second frequency, Units depend on IDBL. (See TCB).
TCWR3(1-40)	0.0	CPCCUR	Upper corridor for differences in wake RCS at the third frequency. Units depend on IDBL. (See TCB).
tecøn	2.0	78	Angle of attack cycle time test parameter, sec. Recommended value is 1.0E-5.
THO	0.0	207	Multiplier for the thrust table which can be considered as a reference thrust level in pounds to be multiplied times the non-dimensional values in the thrust table, THTHO, or alternatively it can be considered as a percentage throttling control to be multiplied times the thrust in pounds in the
		1	thrust table.
THDELT (1-25)	0.0	3618	Change in time from the thrust onset time. Townsec. This table is used only if NTHKUST is 2. (See THDELZ)
THDELZ (1-25)	0.0	3593	Change in altitude from the thrust onset altitude, ZØN, ft. This table is used only if NTHRUST is 1. A thrust table running from a ZØN at 3000000.0 ft. to 50000.0 ft. would have THDEIZ(1) equal to 0.0 for the high altitude thrust and THDEIZ (INHRST) equal to 250000.0 for the low altitude thrust.

SYMBOL	PRESET	COMMON	REMARKS					
THEALO	0.0	113	Initial pitch Euler angle, degrees. (See Figure 4).					
THETAL	0.0	134						
		_	Cone half angle for initial configuration, degrees.					
THETA2	0.0	140	Cone half angle for the configuration after ZTURN, degrees.					
THEZET	0.0	224	Second thrust offset angle (in modified Z-X plane, positive for right hand rotation), degrees. (See Figure 5).					
THTHO (1-25)	0.0	3568	Thrust table corresponding to THDELZ or THDELT depending on NTHRUST. The units of the table can be non-dimensional or pounds, opposite to the units chosen for THO.					
THITBL (1-11)	PRESET	TBLS12	Cone half angle, independent variable for wake Mach number table, ETABL, degrees.					
TINET	500.0	132	Internal temperature of the vehicle for ablation calculations, OR.					
TØFF	0.0	209	Time for thrust termination, sec.					
TØN	0.0	208	Time for thrust initiation, sec.					
Tiør (i- 160)	0.0	CPCCUR	Table of times for the input R/V trajectory data. This table is not currently used in the calculations but it can be useful for bookkeeping purposes and it may be required for future modifications of the program.					
TRAJEN (1+75)	0.0	1644	Nose radius table for drag calculations ($I\phi$ PT = 3), inches. This table must be input for this option in addition to RN1.					
TRAJT (1-75)	0.0	1344	Time table for LOPT = 3 drag calculations, sec. This table affects only the integrated heating.					
TRAJV (1-75)	0.0	1494	Velocity table for LAPT = 3 drag calculations, ft/sec.					
TRAJW (1-75)	0.0	1569	Weight table for IMPT = 3 drag calculations, lb. This table must be input for this option in addition to W1.					
TRAJZ (1-75)	0.0	1419	Altitude table for LAPT = 3 drag calculations, ft.					
TRUALE (1-75)	0.0	1719	Angle of attack table for LAPT = 3 drag calculations, degrees.					
TRZTR	0.0	243	Input transition altitude and option code, ft. If this input is greater than zero then the input altitude overrides the calculated transition altitude.					
			** 44					

SYMBOL	PRESET	COMMON	REMARKS				
TST	100.0	123	Trajectory stopping time, seconds				
TWST	580.0	148	Effective wall temperature used in free molecule				
TWl	1200.0	149	Wall temperature at onset of continuum flow for				
TW2	1200.0	168	ablation and drag calculations, OR. (See TABL). Wall temperature for atlation and drag calculations, at ZTURN (cr at onset of continuum flow if lower), OR. (See TABL).				
TXCGD1 (1-50)	0.0	2833	Table of center of gravity/diameter for initial configuration.				
TXCGD2 (1-50)	0.0	2883	Table of ce ter of gravity/diameter for configuration after ZTURN.				
TO	0.0	102	Initial trajectory time, seconds.				
UP(1-20)		MIN	Upper limits for independent (or design) variables in Fibonacci searches.				
UPBNDZ	0.0	247	Upper altitude boundary on use of tabular imput atmosphere, ft.				
VPLØT (1-160)	0.0	CPCCUR	Table of velocities for input R/V trajectory, ft/sec.				
VO	0.0	106	Initial velocity, ft/sec.				
VOGPLE (1-160)	0.0	CPCGUR	Table of deceleration $(\hat{\mathbf{v}}/\mathbf{g})$ for input R/V trajectory \mathbf{g} 's. Note that the sign convention is actually for acceleration. Increasing velocity is positive, decreasing velocity is negative.				
WCDTAB (1-75)	0.0	3458	Total drag coefficient table for use in overriding the calculated drag.				
WHTAB (1-75)	0.0	3308	Altitude table, ft. Indeperdent variable for added drag table WCDTAB.				
WKALIT (1-10)	5.25,0.	CWAKE	Altitude table, ft. Independent variable for scale height, BETAZ, and look anvies, PHII, PHIZ, and PHI3.				
WL1P (1-160)	0.0	CPCCUR	Table of wake length at first frequency for input R/V trajectory, meters.				
WI2P (1-160)	0.0	CPCCUR	Table of wake length at second frequency for input R/V trajectory, meters.				

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SAMBOL	PRESET	COMMON	REMARKS
WL3P (1-160)	0.0	CPCCUR	Table of wake length at third frequency for input R/V trajectory, meters.
WRF	0.9	ØWL	Factor for reducing the critical constraint or input design variable after each successful solution-finding-process in order to achieve an optimum.
WRIP (1-160)	0.0	CPCCUR	Table of wake RCS at first frequency for input R/V trajectory. Units must be consistant with IDBL input for the decoy calculations.
WB2P (1-160	0.0	. CPCCUR	Table of wake RCS at second frequency for input R/V trajectory. Units depend on decoy IDBL.
WR3P (1-160)	0.0	CPCCUR	Table of wake RCS at second frequency for input R/V trajectory. Units depend on decoy IDBL.
WSTALIT	1.8E5	CWAKE	Maximum altitude for wake calculations, ft.
WIMINE (1+75)	0.0	1119	Mach number table for LOPT = 4 drag calculations.
VIPION (1-75)	0.0	1269	Total pressure table for $IPP = 4$ drag calculations, $1b/ft^2$.
WIRLINF (1-75)	0.0	1194	Reynold's number per inch table for LOPT = 4 drag calculations, 1.0/in.
VTZ	0.0	1044	Altitude table for LOPT = 4 inputs, ft.
(1-75) W1	0.0	133	Initial total weight for the initial configuration, lb.
wa	0.0	139	Total weight of the configuration immediately after ZTURN, lb.
xcom(1)	3.0	IXCØM	Multiplier on step size after a successful step in Rosenbrock procedure.
xcom(5)	0.5	IXCOM	Multiplier (magnitude) on step size after an unsuccessful step in Rosenbrock procedure.
xcqn(3)	0.5	IXCØM	Multiplier times the total successful steps during a stage to obtain the initial step size for the next stage.

SYMBOL	PRESET	COMMON	REMARKS
XCØM(4)	0.01	IXC Ø M	One of the Rosenbrock stopping requirements. The function magnitudes of the last two stages must be within XCOM(4) times the third from last magnitude in order to stop.
хофм(5)	0.5	IXCÓM	One of the Rosenbrock stopping requirements. The ratio of difference between the last and next to last function to the difference between the third from last and second from last function must be less than XCOM(5) in order to stop.
хофм(6)	1.OE-4	ткофм	A step will be called a success in the Rosenbrock process if the function is less than or equal to (1. + XCØM(6)) times the previous value of the function. Note that this definition allows the process to become unstable on constant or very flat functions if XCØM(6) is positive.
XCØM(7)	1.0	IXC/M	The value for the R/V bare body radar cross section for comparison with the decoy cross section calculated from the polynomial with coefficients ACOE.
xcdm(8)	1.0	IXCÓM	Multiplier on the result of the polynomial with coefficients ACME.
(9-200)	0.0	IXOM	Not currently used.
XDTABL (1-11)	PRESET	TBIS12	Normalized air density. Independent variable for electron density table DTABL.
XIØW	4.0	238	The value of the interaction parameter defining the lower boundary of the fairing region between strong interaction and continuum flow regimes.
XRO	0.0	107	Initial range, ft.
XUP	6.0	237	The value of the interaction parameter defi ing the upper boundary of the fairing region between strong interaction and continuum flow regimes.
XIIØW	0.2	240	The value of the rarefaction parameter defining the lower boundary of the fairing region between the free molecule and strong interaction flow regimes.
XXVP	0.4	239	The value of the rarefaction parameter defining the upper boundary of the fairing region between free molecule and strong interaction flow regimes.

SAMBOL	PRESET	COMMON	FEMARKS
XSBQD.	0.0	DRCSEC	Two-body overdense length in wake radar cross section calculations. This input should be left at 0.0 in this model.
х3в	0.0	DRCSEC	Station where linear production terms first dominate the non-linear production terms in the wake radar cross section calculations. This input should be left at 0.0 in this model.
YDTABL (1-11)	PRESET	TBLS12	Normalized enthalpy table. Independent variable for electron density table, DTABL.
ZBAR	-1.0E5	120	Altitude at which printout altitude changes, ft. Note that there must not be more than 160 printout altitudes.

ZDTABL (1-11)	Preset	TBIS12	Ablation to boundary layer air mass flow ratio table. Independent variable for electron density table, DTABL.
ZETA	0.9	93	Accommodation Coefficient
ZNUS	2.0211	DRCSEC	Sea level collision frequency for wake calculations, CPS.
Zøff	0.0	206	Altitude of thrust termination if NTHRUST is 1.
zøn	0.0	205	Altitude of thrust initiation if NTHRUST is 2.
ZPIØ1 (1-160)	0.0	CPCCUR	Altitude table for the input R/V trajectory data. These altitudes must correspond to the decoy printout altitudes defined by ZPR1, ZRAR, ZPRZ, ZO, and ZST.
ZPR1.	1.0E4	118	Initial printout altitude increment, ft. Note that there must not be more than 160 printout altitudes.
ZPR	0.0	119	Printout altitude increment after ZBAB, ft. Note that there must not be more than 160 printout altitudes
ZSII	0.0	121	Trajectory stopping altitude, ft. Note that the program stops all processing if the Mach number goes below Mach 5 before the vehicle reaches the altitude 2ST. (also see H.)

SYMBOL	PRESET	COMMON	REMARKS
ZTURN	-1.0	145	Altitude at which a discontinuous change in the vehicle's configuration and/or weight and/or material is to be made, ft. The new configuration after ZTURN must be completely defined by the input. Changes of heatshield material at ZTURN may not be compatible with the wake calculations. The tests for the ZTURN operation are only made at printout events, thus small changes in ZTURN do not produce continuous results. A negative ZTURN indicates that no discontinuous shape change 'to take place. The screening subroutine (SCREEN): ires that ZTURN not be less than -10.0.
20	0.0	108	Initial altitude, feet. It is intended that the trajectories by initiated at 3000000.0 feet or above. Trajectories starting below this altitude may have numerical difficulties.

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4.0 Description of the Input Sheets

The 20 different input sheets which have been prepared for this program are included in Appendix 1. Certain infrequently used input symbols (Table 5) do not appear on these sheets. A typical memo will contain more than one copy of some inputs and no copies of others. The selection of the input sheets depends on the options being used. The symbols on a given sheet tend to be in groups corresponding to a particular option.

The first input sheet of Appendix 1 is associated with the selection of the search technique, definition of the design variables, and definition of the penalty function. Control data for the search techniques are also included. This sheet contains the primary inputs required to operate the classic check cases.

The second input sheet contains additional control data for the Davidon search technique on the top half and input data for the functions in Subroutine MISC on the botton half.

The third input sheet contains the corridor tables, standard deviation tables, and control codes for the processing of the trajectory performance data and the first wake length performance data.

The fourth input sheet contains similar input provisions for the remaining wake performance data.

The fifth input sheet is associated with the inputs required for the wake calculations. Both the inputs for the wake flow field and the wake radar response are included on this sheet. When this input sheet is used for the reference reentry vehicle, be sure that some IOP codes for wake calculations are one and that the IDBL code for the units is set properly. (See sample input discussion).

The sixth input sheet contains the trajectory initial conditions, stopping parameters, and printout controls.

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The seventh input sheet provides for the definition of the vehicle and the analysis options desired.

The eighth input sheet contains the trajectory plotting controls, the tape output control, and the trajectory printout control, along with some physical constants used in the trajectory calculations.

The ninth input sheet provides for the input of the reentry vehicle performance data for comparison with decoys in some later case. These inputs are associated with the option where MDE is equal to 2 or 3 and IREF is equal to 3. The use of this option involves the risk that the reentry vehicle data and the decoy data are being produced from different models and that the apparent differences in performance may really be differences in prediction techniques. Note that the altitudes in the EPLOT table must correspond to the decoy printout altitudes in the following cases. No interpolations are performed on this input data. Only the performance quantities to be compared with the decoy data are required to be input along with the altitude table.

The tenth input sheet is associated with the influence coefficient calculations of the MøDE = 2 option. This input sheet typically defines the third case of a memo where the first case is a reference reentry vehicle.

IREF = 1, the second case is a basic decoy, IREF = 2 and NDECOT 1, and be third case provides for perturbations of the specified d-sign variables the basic decoy to obtain the partial derivatives of the performance with respect to the design variables.

The eleventh input sheet allows for the calculated drag to be superseded by an input drag table, and/or allows for an added increment of drag to be added to either the calculated or input drag.

The twelfth input sheet provides for thrust as a function of either time after initiation or the absolute value of the altitude change after initiation. The thrust is provided as a multiplier, THO, times an input table, THTHO.

The thirteenth input sheet allows an angle of attack history to be input for use with particle trajectories so that the angle of attack effects on the drag can be included approximately.

The fourteenth input sheet is for use in providing new heatshield material ablation properties.

The fifteenth input sheet allows an input atmosphere table to override the 1962 Standard Atmosphere between the specified altitudes. Exponential interpolation is used for the density table. Note the restrictions on the order of the inputs in the tables.

The sixteenth input sheet allows the preset accuracy controls for the predictor-corrector integration subroutine to be modified. The smoothness and continuity of the penalty function as well as some aspects of the running time depend on these parameters.

The seventeenth and eighteenth input sheets provide access to the drag calculations for free-stream conditions defined in terms of wind tunnel parameters. This optica is not used in conjunction with trojectory or optimization calculations.

The nineteenth and twentieth input sheets provide similar access to the drag calculations for freestream conditions defined in terms of flight parameters. This option is not used in conjunction with trajectory or optimization calculations. Note that this option should not be confused with the IREF = 3 "trajectory input options" on the ninth input sheet.

5.0 Description of Sample Problem Inputs

The sample problems consist of eight cases illustrating a number of different options and capabilities of the program. The first three cases demonstrate the primary operations of evaluating a reference reentry vehicle's performance, optimizing a decoy configuration, and evaluating a single decoy's performance. The last five cases demonstrate the use of classic check cases to provide inexpensive tests of the correctness of the program. The input sheets for these eight check cases are included in Appendix 2.

The first three input sheets define the first case which is the evaluation of a reentry vehicle's trajectory and wake characteristics. The first input sheet provides for identification data and for the definition of the initial and final trajectory conditions. The printout code is set to zero to delete the detailed trajectory output.

The second input sheet defines the weight and geometry of the reentry vehicle along with the heatshield material and analysis options.

The third input sheet provides the inputs necessary to control the wake calculations for the reentry vehicle. Note that the IOP codes numbered 34 and 37 are set equal to one to indicate that the wake length and wake RCS at the first frequency are to be calculated. This indirect means is required in order to get the wake calculations executed for the reentry vehicle. This completes the required inputs for the reentry vehicle. A "transfer card" having a "1" in the first column is inserted at the end of each case to indicate that the program should stop reading input cards and begin to execute the calculations.

The second case, consisting of input sheets 4 through 7, provides an example of a decoy optimization problem. This problem includes results from all the ADTECH IV tasks. The problem is to determine the lightest weight decoy (and its corresponding base radius and length) which is within three specified corridors and is compatible with four geometric constraints. The probability of discrimination based on specified radar measurement errors, number of samples, and probability of false dismissal is to be calculated and printed out. For this example, the minimum weight is to be determined within 20 percent.

The fourth input sheet identifies the beginning of Case 2.0 and specifies that the Rosenbrock Rotating Coordinate Optimizer is to be used. radius and length are identified as the design variables with starting values of 2.5 and 20.0 inches respectively. The first entry in the constraint table identifies the quantity to be minimized. The lower and upper bounds are set so that there will be no contribution to the penalty equation. Since it is weight, which is an input quantity, the code ICØM(3) is left at zero. The next three constraints are the corridor functions. These have multipliers to reduce the numerical values of the corridor function to reasonable levels. Geometric constraints on the base radius, length, cone angle, and bluntness ratio are specified in the next four entries in the constraint table. The last two entries provide for the "difference in the means" and the probability of discrimination to be printed out; however, they will not contribute to the value of the penalty function because their multipliers (AMULAT) are set to zero. This illustrates the use of one of the more subtle output controls. Note that this sheet controls the

optimizers and the penalty function equation but does not control the actual calculations in the trajectory, observables, and effectiveness subroutines. The options and required input data for these calculations must be input to produce the quantities implied on this sheet. There is no automatic connection between the penalty function equation and the actual calculations in the function evaluator.

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The fifth sheet provides for the trajectory program to work with base radius and length (NGEØM = 3) and includes the input weight and nose radius for the decoy. This weight number will be changed in the program during the optimization process. The decoys have the same initial and final conditions and heatshield material as the reentry vehicle, so no other inputs are required. Note that the printout intervals must be the same for the reentry vehicle and decoy, thus they are not input again for the decoy. The wake calculations are to be under the same groundrules, so the wake input data are not repeated.

The sixth and seventh input sheets define the number of entries in the tables being used, the corridors and radar errors, the number of radar samples, the probability of false dismissal and the option codes which provide for the appropriate effectiveness integrals and corridor integrals to be calculated. The number of smoothed radar samples of the wake SRS (4) and SRS (7), are actually equivalent to a value of 0.391 since the effectiveness function is only calculated over a 90000 feet interval (160K to 70K)while the altitude difference in the effectiveness equations is 230000 feet (300K to 70K). This completes the inputs required to define a reference reentry vehicle and a minimum weight decoy. The remaining input sheets illustrate other features of the program.

The eighth input sheet provides for the evaluation and comparison of a single decoy with the reference reentry vehicle. This case is set up to evaluate and plot the data corresponding to the 20.48 pound decoy determined in case 2.0.

The ninth input sheet (Case 4.0) illustrates the inputs for a classic check case using the Rosenbrock Method on a quadratic function of two variables.

The tenth input sheet (Case 5.0) illustrates the inputs for the same problem using the Davidon method. This problem is initiated with FAC equal to 1.0 so the input sheet for the HH matrix is not required.

The eleventh input sheet (Case 6.0) shows a two-variable Fibonacci search of the same function between the limits of ± 10.0 for each variable.

The twelfth input sheet (Case 7.0) provides a constrained check case where the objective is to determine the smallest value of the second design variable which is compatible with the value of the function being between 0.0 and 1.1.

The thirteenth input sheet (Case 8.0) provides a check case for the one-variable Fibonacci optimizer. The second design variable is set equal to 1.0 and the first design variable is varied to locate the minimum of the function.

A listing of the input cards is shown at the end of Appendix 2 in order to illustrate the actual input formats. Note the transfer cards which are at the end of each case. The final "END OF JOB" card and the slash-asterisk card indicate that there are no more cases in this job.

Within a case, if the same input symbol is used more than once, the last input will be used. If an attempt is made to imput a symbol which is not contained in the list of input symbols for this program, the run will be terminated immediately.

6.0 Description of the Output

The primary printed output from the sample problems is reproduced in Appendix 3. The output from cases 1 and 3 through 8 are reproduced in their entirety and the output from case 2 has been edited to show the beginning of the search, the optimum, and the end of the search. The plots produced by these sample problems are reproduced in Appendix 4. These outputs correspond exactly to the inputs described in Section 5.0 and Appendix 2.

The sample problems were executed at the Avco Computer Center using an IBM 360/65 computer and a SC-4020 plotter. The first twelve pages are produced by the system to show the control cards and the memory map for the program. Note that these runs utilize the Avco plotter package which contains a large number of subroutines. When other plotter packages are used, such as the Aerospace PIM package, the list of subroutines marked with asterisks will change considerably.

6.1 Trajectory Printout

The preset input data is shown on the next 3 pages along with the input data for Case 1.0. The heading card identifying the case is on the next page. This is followed by the main output from the first case. This consists of the case, date, memo, and program numbers (where the case number has been incremented by 0.001), a title identifying this output as that of a reference reentry vehicle, a description of the vehicle design parameters, and a summary table of the trajectory and wake calculations. The code, IP, at the end of the design variables is an output of the trajectory calculations. If IP is 6, the calculations have failed to run to completion. The subroutine AIM4RK uses this code to indicate the manner in which the integration process concluded. All values except 6 are considered

normal. The results of the wake calculations are presented for the radar cross section (WAKE R1) and the wake length (WAKE L1) at the first frequency. The wake calculations were started at 160000 feet altitude, thus the values printed out above that altitude are artificial.

The inputs for case 2.0 are shown on the next page. This is followed by the title information on the next page. The title "Case 2.001" identifies the initial decoy in the search for a minimum weight decoy within the stated constraints. The initial decoy weighs 40.0 pounds, has a 0.10 inch nose radius, a 2.5 inch base radius, and a 20. inch length, as was requested on the input sheet. The trajectory and wake calculations are shown below the design variable information in the same format as the reentry vehicle. The second page of case 2.001 shows a number of values of diagnostic information. The corridor integral for velocity is 1.5 x 10 ft2/sec and the decoy left the velocity corridor at 113269. feet altitude. The effectiveness integral for velocity is 6.7x105. The printout showing the two wake corridor integrals to be 0.0 indicates that this decoy is within both wake corridors. The printout starting with "MISC"shows the results of some of the calculations in Subroutine MISC. Since the discontinuous shape change option, ZTURN, is not being used, those para ters which were designed for comparing the vehicle before and after shape change are not of interest. The average density of the vehicle, Wl/Vl, might be of interest if internal packaging problems are anticipated.

The table starting with "IZ" is of particular interest since it summarizes the constraints and provides a means for identifying those constraints which are active and those which are not active. The code numbers, lower bounds, and upper bounds are taken directly from the inputs IDC, CALOW, and CTP. The values listed under OCCUR(IZ) are the actual values of the quantities being constrained. Each of the terms in the penalty equation are listed under "PENALTY". In this case, the decoy meets all the requirements except for the velocity corridor. The penalty term for being out of the velocity corridor is 2.36x109. The probability of discrimination is 0.3219 as shown at the bottom of the ØCCUR(IZ) list. The line beginning "*FEV*" the total of the penalty terms, called F, and the values of the active design variables, called X. This is the information which allows the search to proceed. The function evaluator has been given the two design variables and it has determined the value of the function to be 2.36x109. The remaining output is diagnostic data from subroutine RØSBRK which indicates that the first trial of the first design variable of the zeroth stage has not yet lead to a successful step and the value of the function is U. Here, the 2.36 x 109 is too large for the programmed format and asterisks are substituted. The first values labeled P(I) are the current design variables, the values labeled DP(I) are the changes to these variables to obtain the design variables for the next trial which are printed out under the label P(I). The quantity E(N) is the current step size in the rotated coordinate system. This output indicates that the next decoy will have a base radius of 2.6 inches with a 20 inch length.

This decoy is evaluated in Case 2.002. It is found that it has a penalty function value of 1.98 x 109 which is an improvement over the previous 2.36x109. The next decoy will have a base radius of 2.9 inches, which has a penalty of 9.16x108 as shown in Case 2.003. Case 2.004 shows

continued improvement with increasing base radius; however, in case 2.005 the base radius has been made too large. For case 2.006 the base radius is set back to the best value so far (3.8 inches used in Case 2.004) and the length is increased to 21 inches. This combination of base radius and length produces a decoy which meets every one of the constraints. The value of the function is zero.

The detailed printout for this decoy (Case 2.007) has been edited from Appendix 3 along with Cases 2.008 through 2.025. After case 2.007 the factor WRF = 0.8 is applied to the weight and another search is conducted to determine if there is some combination of base radius and length which will allow a 32.0 pound decoy (0.8 x 40.0) to meet all the constraints. The search started with the base radius of 3.8 inches and the length of 21 inches; however, this decoy was out of the velocity corridor. In case 2.013, a 32 pound decoy with a base radius of 3.6 inches and a length of 22 inches was found to be acceptable. Case 2.014 provided a detailed printout for this decoy.

The weight was reduce 1 to 25.6 pounds (0.8 x 32) and the search was continued until a solution was found at a base radius of 3.2 inches and a length of 23 inches in case 2.025. The detailed output for this decoy was produced in case 2.026 which is included in Appendix 3.

The details of the trajectory, drag, configuration, pressure, heating, and mass loss data are provided for each printout altitude. The definitions of these quantities are provided in Table 6. The summary data are provided at the end of the detailed printout. It will be shown later that this decoy is the minimum weight decoy (within 20%) which meets all the constraints.

The weight is reduced to 16.38 pounds and another search is conducted for an acceptable decoy at this weight starting with the base radius of 3.2 inches and a length of 23 inches in case 2.027. A total of 30 combinations of base radius and length were evaluated at the 16.38 pound weight without finding a solution. Cases 2.028 through 2.055 are not included in Appendix 3. The RØSBRK stopping criteria were met after the 30th trial (case 2.056) which indicates that a minimum of the function has been found and that there is no acceptable solution at this weight. If there are questions regarding the possibility of multimodel functions, it would be advisable to execute another problem to determine if other starting points might lead to a solution at this weight. The summary table at the end of case 2.056 indicates that the only active penalty term is the wake length. If the requirements on this performance function (corridor) were relaxed sufficiently then this decoy would become a solution. However, for the stated problem, there is no solution at 16.38 pounds weight. Therefore the lightest acceptable decoy (within 20%) is the 20.48 pound decoy of case 2.026. This decoy is therefore the optimum.

If it were desired to obtain the optimum more accurately, another problem could be executed starting with the 20.48 pound solution and utilizing a reduction factor, WRF, of perhaps 0.90 or 0.95.

It is interesting to note that four solution-finding problems

(40, 32, 25.6, and 20.48 lbs) were conducted in 26 cases while it took

30 cases to prove that there was no solution at 16.38 pounds. Comparisons
of this type have discouraged the use of more conventional root-finding
methods in Subroutine REDUCE.

In general, no manual data reduction is required since the detailed trajectory output is provided for the optimum decoy (case 2.026 in this problem). Because of the expected length of case 2, the plotter was not utilized. The next case illustrates how a single given decoy might be evaluated and how plots are produced. The decoy utilized in case 3.0 is the same as the one in case 2.026. Note that in general one would not have foreknowledge of the results of case 2.0 when determining the inputs for case 3.0; however, case 2.0 had been run previously in checkout and the solution was known.

Case 3.0 provides a summary output for a single decoy. In addition to the trajectory and wake output, the tables for the differences and corridors are also printed out for each corridor parameter. The plots produced by this case are shown in Appendix 4.

6.2 Plotter Output

The first frame in Appendix 4 is an identification frame produced only at Avco. The total drag coefficient, C_D , versus altitude is shown in Frame 2 while the pressure and base drag coefficient, skin friction coefficient and induced drag coefficient are shown in Frames 3, 4, and 5, respectively. The ballistic coefficient is shown as a function of altitude in Frame 6 while the velocity, Mach number, deceleration, altitude, and dynamic pressure are shown as functions of time in Frames 7-11 respectively. The total decoy weight as a fraction of the initial weight is shown versus altitude in Frame 12. Frames 13-21 contain time histories of data related to the aerodynamic heating. Frame 13 is the stagnation pressure in atmospheres, while Frame 14 is the normalized stagnation enthalpy. Frames 15-18 are the heating rates at the stagnation

6.3 Classic Check Case Printout

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Cases 4 through 8 illustrate inexpensive check cases used to verify the operation of the four search methods and the technique of reducing a parameter to obtain an optimum. Selected portions of the output from these cases have been manually plotted in Figures 6-10.

Case 4.0 illustrates the operation of the Rosenbrock Method in an unconstrained optimization mode. A total of 35 trials organized in 4 stages (coordinate systems) are required to obtain the optimum and meet the stopping criteria. Eleven selected trials are shown in Figure 6 to illustrate the coordinate rotations and general pattern of the trials. The locations of the 35 trials are labeled "X" in the printout.

Case 5.0 contains the Davidon solution for the same problem. The locations of the trials are shown in Figure 7. At each of the points labeled in the figure, three evaluations of the function are performed II-55

in order to evaluate the gradient at each point. The coordinates are labeled "X" in the printout and the gradient is labeled "G". A total of 16 trials are performed. This illustrates the efficiency of the Davidon method on quadratic functions.

Case 6.0 contains the two-variable Fibonacci solution for the same function where the search is conducted between values of the independent variables of ±10.0. Twelve points are used for each variable for a total of 144 evaluations of the function plus 1 evaluation for final printout. Six of the locations are shown for each variable in Figure 8. The points labeled 1-5 show the technique of fixing the value of X_2 while finding the best value of X_1 . Next, the value of X_2 is changed and the points 13-17 are evaluated. This process is continued until the interval of uncertainty around the optimum has been reduced to the amount implied by the use of 12 points. The final (145th) printout is the best of the previous 144 trials. Note that the last of the group of 12 trials (or the last of the 144) is not necessarily the optimum.

Case 7.0 contains a rather long check case somewhat analogous to the problem solved in Case 2.0. The problem here is to find the minimum value of X2 and the corresponding value of X1 which is within the constraint of the function being not greater than 1.1. The Rosenbrock Method is used to vary both X1 and X2 until an acceptable value of the function is obtained. The constraint on X is tightened and the search is repeated. This process is illustrated in Figure 9 where the constraints are labeled "Ci". In the printout, the current value of the constraint is the first value under the label "UPPER BOUND". The trial counter, NTRIA, starts over for each new value of the constraint and the coordinate system returns to a system parallel to the initial one. The best values of X1 and X2

so far are used to restert the search. After it is shown that there is no acceptable solution with a constraint of 0.4519, the optimum can be identified as the last successful solution which occured when $X_1=1.721$, and $X_2=0.502$. The value of the function at this point was 1.099. The theoretical optimum is at (1.700, 0.500) with a function value of 1.100. This is considered to be good agreement.

Case 8.0 illustrates the one-variable Pibonacci search using 12 points. The locations of the first 5 points are shown in Figure 10. Note that X_2 has been set to 1.0 and that the search is for the best value of X_1 between the limits ± 10.0 .

The last case is followed by the "END OF JOB" card and two pages of systems output. The running time of these 8 cases plus the time required to load in the program was 19.25 minutes on the IRM 360/65 using overlay techniques. The corresponding time for the IRM 360/75 without overlay was 11.15 minutes using the binary program on magnetic tape. Slightly shorter time will be expected when using the binary program from the disk. The major part of the running time is used by Case 2.0 which may be bypassed by removing the transfer card immediately before the card punched "case 3.0". This provides a shorter check series requiring 5.2 minutes on the IRM 360/65 with overlay. Each trajectory is averaging about 13 seconds running time on the 360/65. Changes in the printout intervals or the number of performance parameters will change this number. Roughly 3 minutes are required to load the program into the core.

TABLE 1
SUMMARY OF SCREENING LIMITS

Item	Conditions	Lower Limit	Parameter	Upper Limit	Units
13g apless		0.0	RW 1.	10000.	inches
2	If ZTURN >0.0	0.0	RN2	10000.	inches
***] ***]		0.0	P_{N_1}/R_{B_1}	0.6	May right state
14	If ZTURN > 0.0	0.0	R_{N_2}/R_{B_2}	0.6	VER DED MER
5		3.0	LAl	168.0	inches
6	If ZTURN > 0.0	3.0	LAS	168.0	inches
T		4.0	THETA 1	40.0	degrees
8	It ZTURN > 0.0	4.0	THETA2	40.0	degrees
9		-10.0	ZTURN	ZO	feet
10	If ZTURN >0.0	0.0	W2	Wl	pounds
11	If NTHRUST ≠ 0	0.0	ISP	T0000 •	seconds
12	If NTHRUST - 1	z ø ff	zøn	ZO	feet
13	If NTHRUST = 2	TO	Tyn	TØFF	seconds
14	if nthrust = 1	est	zøff	zøn	feet
15	If NTHRUST = 2	Tøn	TØFF	TST	seconds

TABLE 2 MATELY OF DEPTON COSES, ISP

PLOTS VERSUS NETUDE	In I	To open and the same and the sa		To as	Enter the second section of the second secon	(//	0	Manageresis und 11 Homestan III	
		8	The second secon	11, 01, 0	O IC	In low-adved, inc. in.	. misseppes () 1000s. Table () 000 0000 () 00	C S C S C S C S C S C S C S C S C S C S	TO THE PROPERTY OF THE PROPERT
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	VELOCITY I DP (1)	DECENSION	BALLISTIC	W/KE LEI'G" FOR 1ST FRESO.	7445157577 FOR 2nd FRED.	WAKE LETSTE FOR 3M FREG.	WAKE RCS	WAKE RCS FOR 2W SPER.	FOR PREC.

IOF(76) SHOULD SE 1 AT AVCO AND O AT ARROSPACE I DEP(64-15; 83-90) ARE MET INFUTE



Table 3 Design Variables and Design Variable Constraints

ØCCUR Code No.	Ttem	Related Option
133	W.L.	
134	THETAL	NGEØM
135	EAN T	NGEØM
136	RBI	4830-644
137	LAMDAL.	NGEØM
1.38	1.A J.	NGEØM
139	MES.	ZTURN
71.140	THETTAS	NGEØM, ZTURN
141	RNS	NGEØM, ZITURN
3.442	RES	ZTURN
1.43	LAMDAS	NGEØM, ZTURN
144	LAS	NGEØM, ZTURN
145	ZTURN (Do not use)	-(3040009)
205	ZØN	NTHRUST
2 0 6	ZØFF	NTHRUST
207	THO	NTHRUST
208	TØN	NTHRUST
209	TØFF	NTHRUST
222	ISP	NTHRUST
3233-3307	HTAB(1-75)	MAXCD
3308-3382	WHTAB(1-75)	MAXWCD
3383-3457	CDTAB(175)	MAXCD
3458-3532	WCDTAB(1-75)	MAXWCD
3568-3592	THTHO(1-25)	NTHRUST, ITHRST
3593-3617	THDELZ(1-25)	NTKRUST, ITHRST
3618-3642	THDELT(1-25)	NTHRUST, ITHRST
3963	CCØN	IØP(22-63)

Table 4 Storage Locations for Special Functions

Ø CCUR	Related	
Code No.		
And And A Street TA P 1 B	Option	T T C C T T T T T T T T T T T T T T T T
2031		
3901	ZTURN	W2-W1F, 1bs
3900	ZTTUFAY	THETE-THETEF, deg.
3903	ZTURN	TOTAL CL. 1984 T. KN. J
3904		RNZ-RN1F, in.
3905	ZITURN	RB2-RB1F, in.
	ZIPU F3N	LAMDA2 - LAMD1 F
3906	ZTUEN	IA2-LAIF, in.
3907	ZTURW	W2/W1F
3908	ZTURN	
3909		THETAS/THET1F
3910	ZTURN	RN2/RNIF
	ZTURN	RB2/RBLF
3911	ZTURN	LAMDAE/LAMD1F
3912	ZTURN	LAZ/LAIF
3515	699	W1/V1. 1b/ft2
3914		W1/ VI + 10/ I U ₂
3915	en yek er v. d' my la.	Act Act and the first the state of the state
	IØP(7)	Volocity Corridor Integral for
3916	IØP(8)	Deceleration Corridor Integral, ft
3917	$I \not p P (9)$	polition corrade integral, it
	when And	Ballistic Coefficient Corridor
		Integral, 1b
2/22 9		It
3918	IØP (7)	Velocity Corridor Breakthrough
	1.7	Altitude, ft.
3919	IØP (8)	
	TAL (O)	Deceleration Corridor Breakthrough
3920		Altitude, ft.
3 (94) S	IØP (9)	Ballistic Coefficient Corridor Break-
	***	through Altitude, ft.
3921	IGDH, IGDL	General Difference, 1
3982	IGDH, IGDL	
3920+I		General Difference, 2
3940	IGDH, IGDL	General Difference, I
	IGDH, IGDL	General Difference, 20
3941	IØP(34), IDBL	Wake Length Corridor Integral 1,
		Meter 2-ft., or db-ft.
3942	IØF(35), IDBL	Wake Length Corridor Integral 2,
	The year of the second	Motor of the American St.
3943	IØP(36), IDBL	Meter -ft, or db-ft.
And the Paris of t	TAL (20), IDBD	Wake Length Corridor Integral 3,
2014	mendan Lauren N	Meter deft, or db-ft.
3944	IØP(37)	Wake RCS Corridor Integral 1, Meter-ft.
3945	IØP(38)	Wake RCS Corridor Integral 2, Meter-ft.
3946	IØP(39)	Webs 900 Comfdon Total 2 2 2 2
3947	IØP(34)	Wake RCS Corridor Integral 3, Meter-ft.
	101 (34)	Wake Length 1, Corridor Breakthrough
2010	4 4	Altitude, ft.
3948	IØP(35)	Wake Length 2, Corridor Breakthrough
		Altitude, ft.
3949	IØP(36)	
	Tht (20)	Wake Length 3, Corridor Breakthrough
2050	malan dan wa	Altitude, ft.
3950	IØP(37)	Wake RCS 1 Corridor Breakthrough
		Altitude, ft.
3951	IØP(38)	
	\ W /	Wake RCS 2 Corridor Breakthrough
3952	Tdn(20)	Altitude, ft.
J. 7 7	IØP(39)	Wake RCS 3 Corridor Breakthrough
2000	malm 43 A	Altitude, ft.
3953	IØP(4)	Velocity Effectiveness Integral, ft.
		TIVE TIME TO A SOURCE THE SERVICE THE SERV

TABLE 4 (CONTINUED)

OCCUR Code No.	Related Option	Ttem .
3954 3955 3956 3957 3958	IØP(5) IØP(6) IØP(28) IØP(29) IØP(30)	Deceleration Effectiveness Integral, ft. Ballistic Coefficient Effectiveness Integral, ft. Wake Length 1 Effectiveness Integral, ft. Wake Length 2 Effectiveness Integral, ft.
3959 3960 3961 3962 3963	IØF(31) IØF(32) IØF(33) IØF(4-6,28-33) CCØN	Wake Length 3 Effectiveness Integral, ft. Wake RCS 1 Effectiveness Integral, ft. W ke RCS 2 Effectiveness Integral, ft. Wake RCS 3 Effectiveness Integral, ft. Probability of Discrimination Wake Seeding Design Variable
3504	xcøm(γ-8), acøe	Free Space RCS Difference
3965	IØP(4-6, 28-33)	Difference in the Means, o, (Subroutine EFFECT)

TABLE 5 INPUT SYMBOLS NOT INCLUDED

ON THE INPUT SHEETS

SYMBOL	REFER TO	REMARKS
В	· ZPRS	For debugging only
DIABL	Preset Deck	Wake Tables
EMCTRL	Preset Deck	Wake Tables
ENTABL	Preset Deck	Wake Tables
ERNTBL	Preset Deck	Wake Tables
ETABL	Preset Deck	Wake Tables
HSTABL	Preset Deck	Wake Tables
IPMT		
K	risi.	For future use in FEV
ØCCUR	no.	Future optimizer control
RSTABL		For debugging only
THTTBL	Preset Deck	Wake Tables
	Preset Deck	Wake Tables
XDTABL	Preset Deck	Wake Tables
YDTABL	Preset Deck	Wake Tables
ZDTABL	Preset Deck	Wake Tables

The state of the s

TABLE 6

DEFINITION OF DETAILED OUTPUT QUANTITIES

Translational Quantities

TIME Flight time, seconds

Z Altitude, feet

V Free-stream velocity, ft/sec.

GAMF Flight path angle, degrees

XR Downrange component of range, feet

Bellistic coefficient, normally W/CDA, 1b/ft2,

but see the input ICOM(10).

ZTR Altitude of the beginning of transition from

laminar to turbulent flow, feet.

QD Free-stream dynamic pressure, 1b/ft².

MINF Free-stream Mach number

VDOTOG Deceleration (actually acceleration), g's.

BETAP Partial Perivative of BETA with respect

to altitude, normally lb/ft3.

TH Total thrust value, 1b.

TXT Axial component of the thrust vector, lb.

YR Cross-range component of range, feet.

PSIAIP Azimuth engle, degrees.

D/W Aerodynamic contribution to the deceleration,

drag/weight, g's.

TABLE 5 (cont'd)

DRAG QUANTITIES

Total drag coefficient based on base area, (AREF) and free-stream dynamic pressure, QD. In the fairing region between continuum and non-continuum flow regimes, it is not equal to the sum of the laminar flow terms.

CDP Forebody pressure drag including angle of attack effects, if any.

CDFINF(BL,__,WB)

The skin friction drag coefficient corrected for bluntness and blowing and indicating either laminar or turbulent flow regime.

CDB Base drag coefficient.

CDPO Pressure drag coefficient for zero angle of attack.

CDFINF(BL,__,NB) The skin friction drag coefficient corrected for bluntness but not blowing and indicating

either laminar or turbulent flow regime.

CDI Induced drag coefficient

XBAR Viscous interaction parameter

REYINFLA Free-stream Reynold's number based on

axial length.

XBAR1 Hypersonic rarefaction parameter

CDI/P Induced pressure drag coefficient

CDI/SF Pressure induced skin friction drag

coefficient

CDI/TC Transverse curvature induced drag coefficient

TABLE 6 (cont'd)

Configuration Quantities

EAN Nose radius, inches.

THETA Cone half angle, degrees.

LA Axial length of vehicle, inches.

LAMBDA Bluntness ratio, RN/RB.

AREF Reference area for the drag coefficient,

square feet. Note that the instantaneous base radius in inches is equal to 12 \sqrt{AREF}

Instantaneous weight of the vehicle.

DELW Total change in weight from initial weight.

WABL The weight change due to ablative mass loss.

WIHRST The weight change due to thrust mass loss.

Heating and Mass Loss Quantities

QDOT (STAG) Stagnation point aerodynamic heating.

BIU/FT sec.

QDOT (SONIC) Sonic point aerodynamic heating, BTI/FT2 sec.

HSRTO Normalized stagnation enthalpy

PSP0 Stagnation pressure, atmospheres.

PEPSB Pressure distribution $P_{\rm E}/P_{\rm S}$ along the body.

QDOT Aerodynamic heating distribution along the

body, BTU/PT2 sec.

MDOT Mass loss rate distribution along the body,

LB/FT sec.

QINT Integrated heating distribution along the

body, Bru/Fr2.

QINT (STAC) Integrated stagnation point heating, BTU/FT2.

QINT (SONIC) Integrated sonic point heating, BTU/FT2.

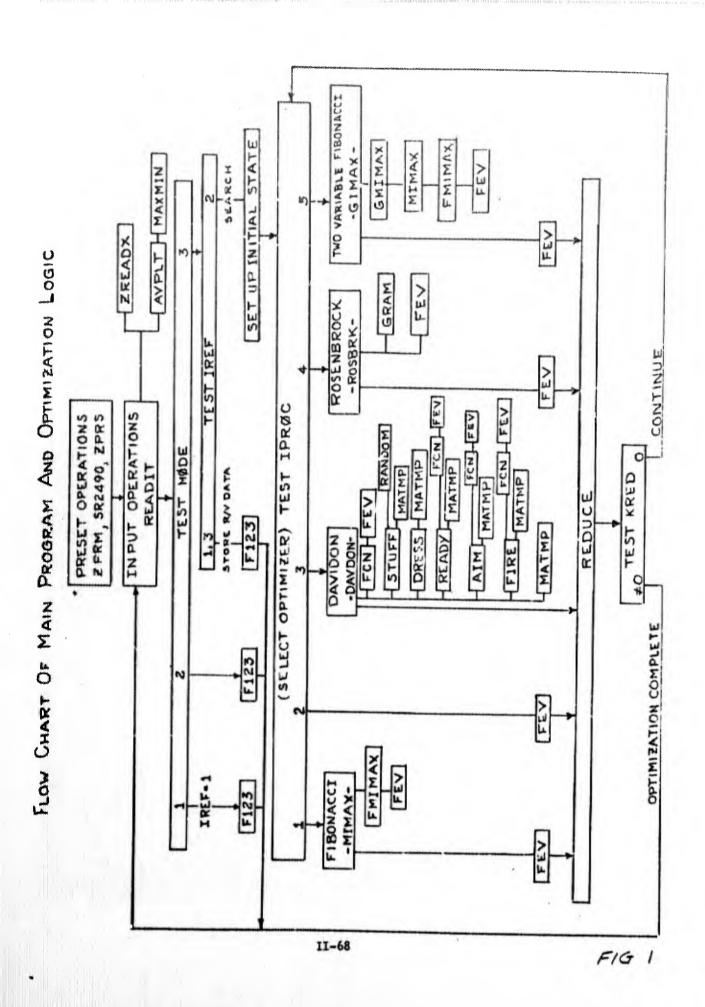
TABLE 6 (cont'a)

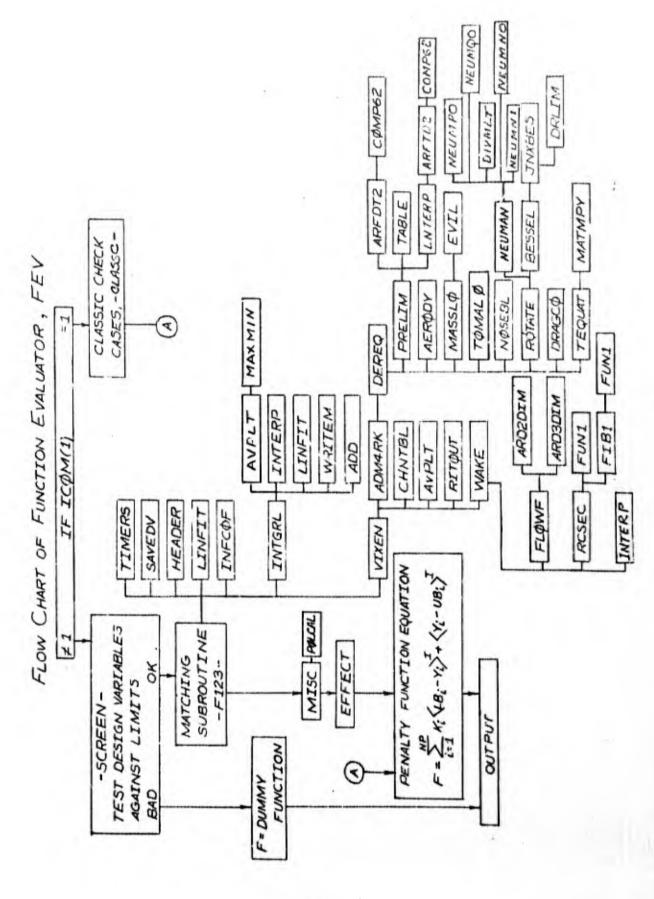
Rotational Quantities

PSI, THEALF, PHI Euler angles, degrees. ALPRIM Angle of attack. PR, Q, SMR, p, q, r angular rates, rad/sec. ALPENV Envelope angle of attack, degrees. C. 10 Stability derivative of pitching moment coefficient with q. CM Pitching moment coefficient. (31) Yawing moment coefficient. MP Frequency of oscillation, 1/sec.

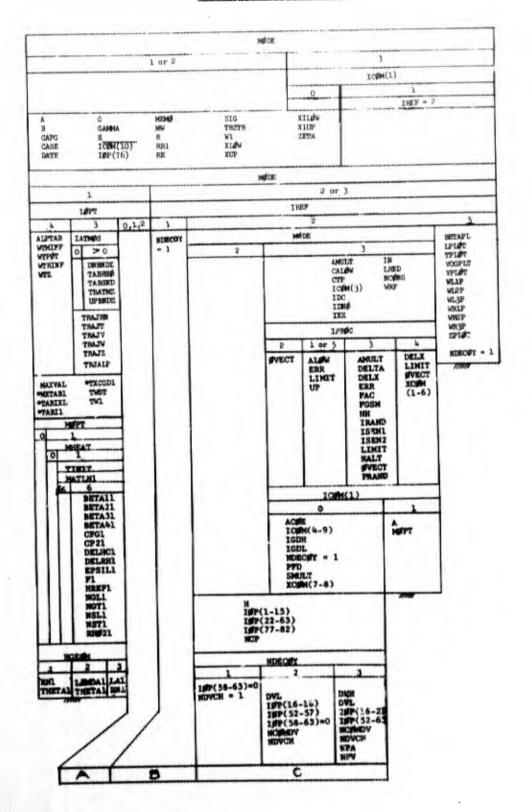
Completion Codes

TTERM	-1,	Function is equal to zero.
	ο,	Function is not equal to zero.
	1,	Function 1s undefined.
KRED	-1,	Number of applications of WRF has reached LRED limit
	ο,	Optimization will continue with another search
	1,	Process will stop with either a non-zero function or an undefined function (see ITERM above)
TP	1-5	Normal completion in trajectory integration
	6	Abnormal situation in trajectory calculations.





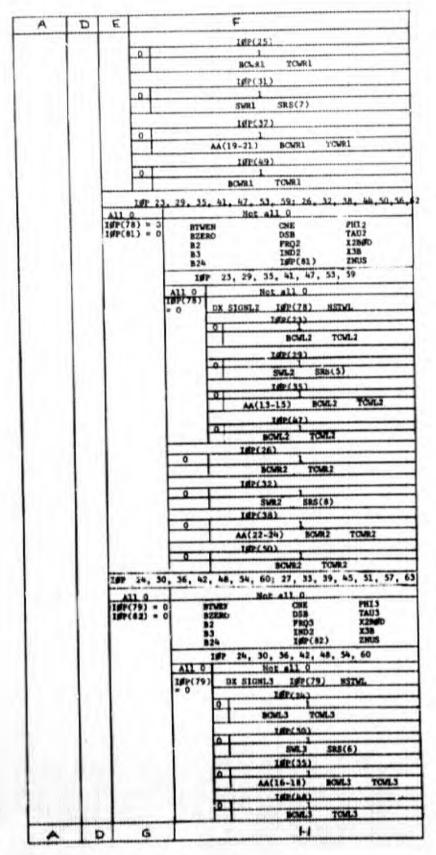
INPUT INTERRELATIONSHIPS

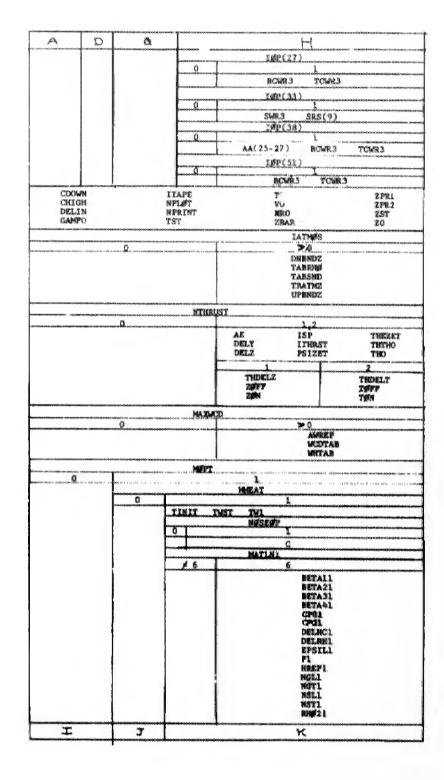


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	1#P (77)** 0 1#P (80)**	87WEN 82ER# 82 83 824	##	CNE DSB FRQ1 IND2 IFF(80		PHT1 TAUL X289D X38 ZNUS
			34. 40. 46. Not	52, 58; ell 0	25, 31, 3	7, 43,49,55, 61
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			0	SB (9)	SR5(3)	
			0 10	P(6)	TCB	
			9 16	BCD P(3)	TCD	******
				A(4-6) P(14)	BCD	TOD
				Sn P(B)	Sec. 2)	
			0 19	BCD P(5)	TCD	
			14	BCV P(2)	TCV	***
				A(1-3)	BCV	TCV
			12	SV (P(7)	5RS(1)	
			0 16	P(4)		
			0	SP(1) BCV	TCV	

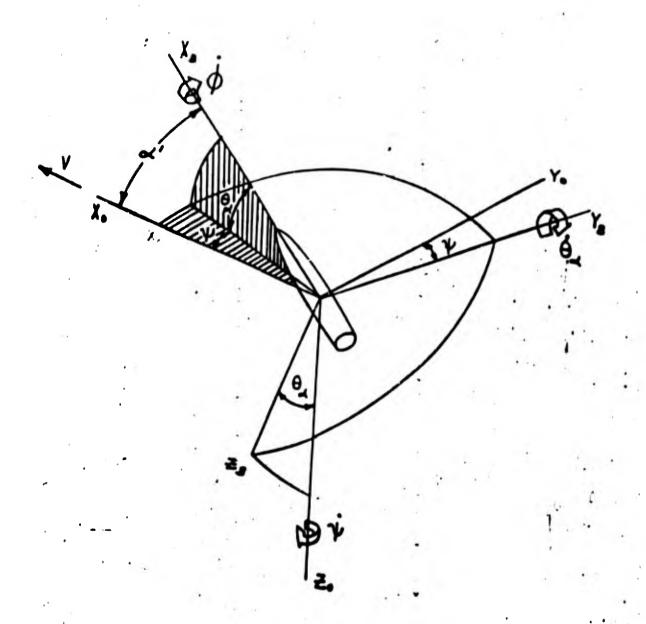
11-71

FIGURE 1-:



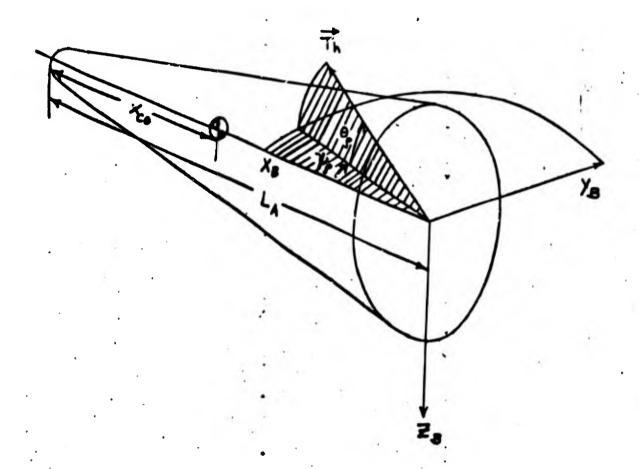


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FIG4



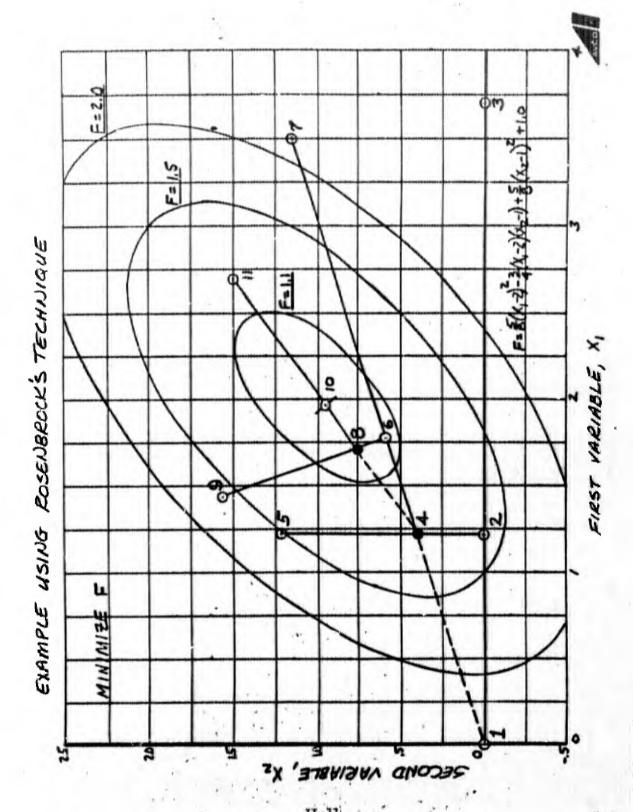
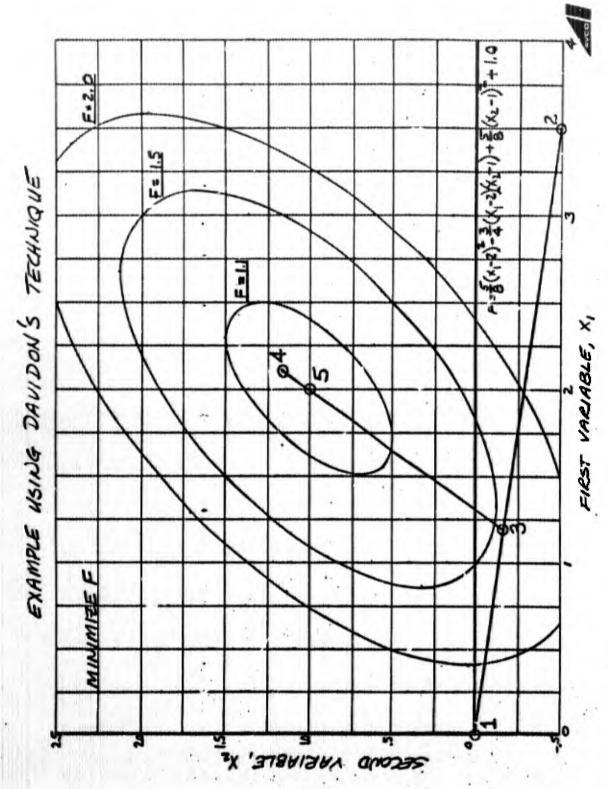


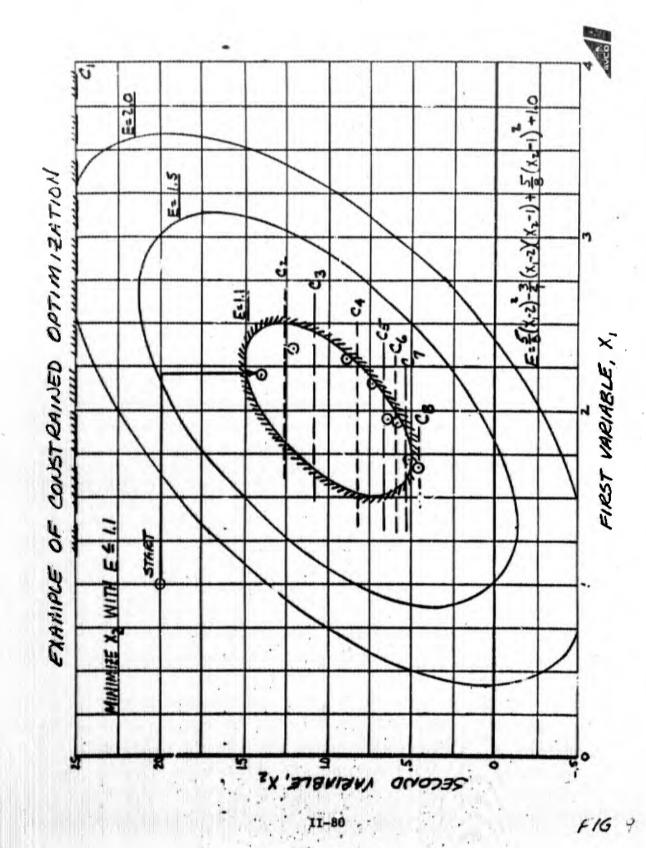
FIG 6

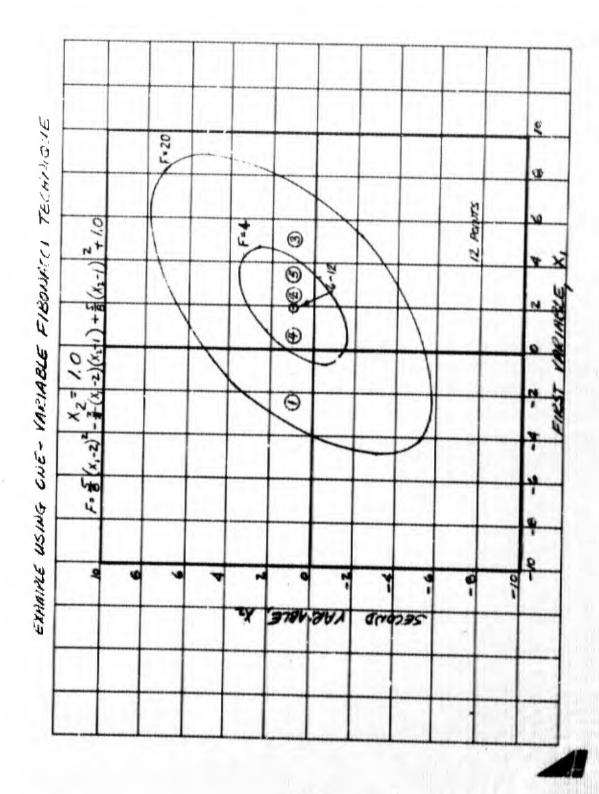


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FIG 10

APPENDIX I

MASTER INPUT SHEETS

REQUEST PO	# IMPUT	2542	KARINE INS.	MECTION HIS			TION SHEET	# 4
CASE		DATE					-	PE 3
Н				-				
"MODE	וא) "וכפו	N(i)	(o) ¢I	REF	(i) Mg	IPT	_(0)	
*IPRØC	_ 6) KC	\$M(1)	(3.0)	ERR	(44)	TCO	1(3)	10
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			(6.5)					
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1	(0-0)						
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			(1.0)		(1.0)		
3	If PAC is O.	O, the follo	owing table	will be use	d for the	initial	matrix:
HH(1)	BB(41)	BH(8).)	BEI (151)	BH (161)	HH (501)	HR (5)	m (s
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DIGITAL COMPLITER INP	TENER IN INC. THE SECOND	HALL CELLING MACE	COMPENSA	141 Pd 191 ()		1663
REQUEST FORM	2542		PAGE	# H	De prof. s	

INPUT MATERIAL PROPERTIES

If the material number designation = 5, input the list of material properties corresponding to the configuration. Both sets of input properties may be used at the same time.

II MATLNI = 6 input:		If MATLN2 = 6, input:
BETAIL	((t/sec / R)	BETALL
BETA21	(°R (t/sec)	BETAZZ
BETA31	_(dimensionless)	BETAIL
BETA41	_ (°R)	BETA42
CP21	_ (Btu/lbm/ ^o R)	CP22
CPGI	(Btu/lbm/ ^O R)	CPG2
DELHCI	(Bud/lbm)	DETHC5
DELRHI	(1bm/ft ³)	DELRHZ
EPSILI	(diracnaionless)	EPSIL2
FI	(dimensionless)	F2
HREFI	(dimensionless)	HREF2
NGL1	(dimensionless)	NGL2
NGTI	(dimensionless)	NGT2
RSLI	(dimensionless)	NSL2
NST1	_ (dimensionless)	NSTZ
RHØ21	(lbm/ft ³)	RHØ22

with LOPT = 4, wind t	s except LØPT = 4 (input wind unnel conditions override the in (0) If > 0 use input atmosph	put aimoophere.
UPBNDZ *NOTE*** Table mu	this range. Maximum	of 50 values allowed.
TBATMZ (ft.)	TABRΗΦ (lbm/ft ³)	TABSND (ft/sec)

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Y	CHIGH(2)			CDØWN	(2)
kimme	CHICH(3)	**************************************	Militarios concentration con constituints and constituints and constituents are constituents and constituents and constituents and constituent	CDOWN	
N _E	CHIGH(4)			CDØWN	
W	CHIGH(8)			CDØWN	(5)
ER W	CHIGH(6)			CDØWN	
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¥	CHICH(8)			CDØWN	8)
<u>a</u>	CHIGH(9)			CDOWN	
o	CHIGH(10)			CDØWN	10)
Q	CHIGH(11)			CDØWN	11)
R	CHIGH(12)			CD ØW N	12)
P	Снивн(13)				13)
Y	СНІСН(14)			CDØWN	14)
W.	CHIGH(15)			CDØWNI	15}
W	CHIGH(16)				16)

All the CHIGH's are preset to 1.0D-4 and all the CDOWN's to 1.0D-3. If the absolute value of the quantity being integrated is less than or equal to 1, then UPBND(N) = CHIGH(N) and DNBND(N) = CDOWN(N). If the absolute value, TEM(N) is greater than 1, then UPBND(N) = CHIGH(N) x TEM(N) and DNBND(N) = CDOWN(N) x TEM(N).

UPBND: The upper bound on absolute difference that is allowed between the extrapolated and interpolated values. If this bound is exceeded by the difference, the delta of integration is reduced and the integration retried.

DNBMD: The lower bound on absolute difference that is allowed between the extrapolated and interpolated values. If this bound <u>acceds</u> the difference, the delta of integration is increased and integration is carried on.

NOTE* W = WINITIAL - A WABLATION WTH " WINITIAL - AWTHRUST

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GITAL CO			PHOBLEM NO	2542	p q i	ORAMNEH:	
	EST FOR		TITLE:			IEXT. TEST. TAM	
O NO	CTION NO.	WORK O	MDER NO.	TE240 USE ONLY	REQUESTED BY:	EXT. EST. TIME	PAGE OF PAG
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atn.osp	heres,	thrust	, and tap	e options may	not be used.	Input mate	rial option
may be	used.	Case	s using the	his option may	be stacked v	vith those of	other options.
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Mass loss option is used only to obtain m and W for use in skin friction drag coefficient and base drag coefficient, respectively.

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TRAJT need not be actual time values unless correct integrating heating is needed; may be used merely as a counter.

APPENDIX 2

INPUT SHEETS FOR SAMPLE PROSLEMS

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REQUEST FORM	206			PAGE I	OF 1 PAGES
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	2241
W1 500.0 PN1 1.0 RB1 15.0 THETA1 8.0 M2PT 1 MHEAT 1 NØSEØP	2242
MATLN1 3 TW1 4850.0 TABL 2700.0 TOBL 3 TOPA34) 1 (37) 1	1 2241
WSTALT 160000.0 WKALT 300000.0 0.0 BETAZ 22.0 22.0	2241
PHI1 7.0 7.0	2241
The second secon	2241
CASE 2.0	
H DECOY ABITMISATION EXAMPLE	2241-2
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IDNO 136 138 DELX 0-1 1.0 ØVECT 2.5 20.0	and the second section of the second
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[0P(31) 1 (37) 1 SRS(7) 1.0	2241-2
BCV -30.0 -40.0 -110.0 -170.0 -200.0	2241-2
	2241-2
SV 30.0 40.0 110.0 170.0 200.0	
	2241-2
2011 TVV+V 400+0 300+0 100 t 120	
3884 4840 10a0 10a0 10 0 10 0	2241-2,
**************************************	2241-2
TCMR1 30.0 30.0 38.0 46.0 50.0	
CASE 3.0	2241-2
H DECOY EVALUATION WITH PLOTS	2244
1980c 2 Mplat	2241-3
IPROC 2 NPLOT 1 1 1 10P(1) 1 (22) 1 (25) 1 (77) 1 (80) 1	2241-3
OVECT 3.20 2 0 11	2241-3
DVECT 3.20 23.0 VI 20.48	2241-3
1	
CASE 4.0	
H RESENBROCK UNCONSTRAINED COTTURED	2241-4
	2291 44
	2241-4
0.25 2.625 IPROC A ICOMAS 0.625 -0.75 0.625 -1.75	2241-4
0.25 2.625 IPROC & ICOM(3) 0 LIMIT 200 LRED 0 IEX 1 IN 2	2241-4 . 2241-4
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APPENDIX 3

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DATA (1,3	311 1.29	3.33	7.00 11.	.2 16.2	22.0	29.0 3	37.7 48.6	6 63.0	
DATA. 11.	4)1 1.43	3.58	7.84 13	.4 21.0	33.4	49.2 3	80.0 100	0.	
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DATA (1.	611 1.70	9.5	10.2 21	.8 51.0	0.001 0	100.0	100.0	100.0	PRESE 72*DATA
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-	13	100	100.0	100.0	100.0	100.0	100.001	0.00	
8	XDTA BL	0000	100.00	.03	.1	1.56			PRESE 84*DATA
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047.50		:		:					PRESE STADATA
DATA (1).	5,111 0.0 0	0.70E1	1 0.38E	3 0.32E10		0.25E12 0.	0.32E13 0.	-17614	PRESE BB*DATA
DATAS (1)	4,111 0.0	0.50E1	0.38E	3 0.32610	10 0.25512	0	.32£13 0.	-17E14	-
DATA 11.	3,131 0.0	0.32E2	0.12E	4 0.11E10	10 0.78E11		0.10613 0.	.65E13	PRESE 92*DATA
DATAS 0.3	1614 0.8751	+0	0416E15 0	-16E15	0 0.05611		0 14515 0	90613	PRESE 934DATA
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110	1,111 0.0	0.90E2	0.30E	4 0.20E9	9 0.05E11		0.14512 0.2	.20E13	
A. (10.	5,211 0.0	-	3.5E7	2.0€11	1.3E12	2.0E12	4.5F12	1.4F13	
A	13 3.0513 1,2)1 0.0	1.065	3.567	2.0611	1.3512	2.0E12	4.5512	1.4513	PRESE 99*DATA
140 2.4	÷.	4.0E	m'						PRESEIGIFDATA
	13,211 0.0	2.25	***	T.IEII	2.3E11	4.5E11	2.0E12	5.0E12	PRESELOZ#DATA
DATA (11.2	1,211 0.0	2.0E4	7.0E6	4.0E10	1.6F11	2.0F11	6.0F11	1.4612	PRESEIO4 *DATA
DATA (1.1	1,211 0.0	2.064	7. 0E6	4.0619	1.6E11	2.0E11	6.0E11	1.4512	PRESEIO6*DATA
BAT 44 2.0	115 3.051	4.061	2	1			1		107*DAT
	1307 0-0	2	1.568	1130-1	2-0E11	1,3613	1.6513	2.1613	PRESE108*DATA
**	.311 0.0	2.465	1.5EB	7.6E11	5.5612	1.3613	1.6513	2,1513	PRESELLOMBIA
DATA: 11,3	1,311 0.0	1.265	7.167	3.7611	2.4612	5.8E12	6.2F12	7.7E12	PRESETTIONATA
DATA (1,2	311 0.0	4.554	2. TET	1.4E11	7.1611	1.7612	1.7E12	2.0E12	PRESEI14*DATA
DATA: [1.1	311 0.0	4.5E4	2.TE7	1.0E11	7.1511	1.7512	1.7512	2.0F12	PRESELLS*BATA
# 2 # 2	12 3.5E12	7.9ES	2.358	2.0512	1.8613	6.0F13	1.4514	1.3514	PRESETT-DATA
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3E12 8.0E12 3.0E13 3E12 8.0E12 3.0E13 3E11 2.5E12 1.1E13 3E11 2.5E12 1.1E13 3E11 2.5E12 1.1E13 0.5 0.0 1.0 1.0 1.0 1.0 0.0 0.0 0.0 0.0 0.0	3E8 2.0E12 1.8E13 6.0E13 5E8 1.3E12 8.0E12 3.0E13 0E7 4.3E11 2.5E12 1.1E13 0E7 4.3E11 2.5E12 1.1E13 0E7 4.3E11 2.5E12 1.1E13 0.0 0.0 0.0 1.0 0.10 0.0 0.0 1.0 0.10 0.0 0.0 1.0 0.10 0.0 0.0 0.0 1.0 0.04 86.0 (169) 1.0	7.9E5 2.3E8 2.0E12 1.8E13 6.0E13 5.2E5 1.5E8 1.3E12 8.0E12 3.0E13 1.7E5 5.0E7 4.3E11 2.5E12 1.1E13 2.0E13 1.7E5 5.0E7 4.3E11 2.5E12 1.1E13 2.0E13 2.0E14 0.0 0.0 0.5 2.0E14 0.0 1.0 0.0 1.0 246E14 0.0 1.0 0.0 1.0 20.0 1.0 1.0E3 0.0 0.0 1.0 20.0 1.0 0.0 1.0 0.0 1.0 20.0 1.0 0.0 1.0 0.0 1.0 20.0 1.0 1.0 0.0 0.5 20.0 (164) 0.04 86.0 (169) 1.0 20.0 (164) 0.04 86.0 (169) 1.0 20.0 (164) 0.04 86.0 (169) 1.0 20.0 (164) 0.04 86.0 (169) 1.0 20.0 (164) 0.05 8.0 8.0 1.0 20.0 (164) 0.04 86.0 (169) 1.0 20.0 (164) 0.04 86.0 (164) 1.0 20.0 (164) 0.04 86.0 (164) 1.0 20.0 (164) 0.04 86.0 (164) 1.0 20.0 (164) 0.04 86.0 (164) 1.0 20.0 (164) 0.04 86.0 (164) 1.0 20.0 (164) 0.06 86.0 (164) 1.0 20.0 (164) 0.06 86.0 (164) 1.0 20.0	111 0.0 7.9E5 2.3E8 2.0E12 1.8E13 6.0E13 11. 0.0 5.2E5 1.5E8 1.3E12 8.0E12 3.0E13 11. 0.0 1.7E5 5.0E7 4.3E11 2.5E12 1.1E13 1000.0 1.0 1.0 1.0 1.0 0.0 0.0 1.0 0.0 1.0 1000.0 1.0 1.0 1.0 1.0 0.0 0.0 0.0 0.0 0	1.3514 PRESELZO*DATA	5.2E13 PRESE12*DATA	1,3613 PRESE124-DATA	1.3E13 PRESEIZSADATA	PRESEIZBEDATA	PRESEISO#DATA	PRESEISZ#DATA	PRESEL34+DATA	PRESEL36*DATA	PRESENTATION OF THE PRESEN	PRESE140*DATA	PRESEL41#DATA	PRESEL42#DATA	PRESELAGOATA	PRESEI 46 #DATA	d		PRESEI 52 + DATA	0.03	PRESENCE TA	PRESE158*OATA			2241 #DATA
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	-	0.0	LEAVE CORREGOR AT	AT 0.0				
INTEGRAL	OF I WAKE !	OF I WAKE LI/SIGNA ITEZ I.	1.07286440 05					
-	HAKE AT THIEGRAL	0.0	LEAVE CORRIDOR	Af 0.0				
INTEGRAL	OF CHARE	INTEGRAL - OF - 1 - MAKE - RIF STGMA 10-2 - 8.09744018 05	99744918 95	ì				3
136	0.0	14E/141F	RNEYRMIF 0.0	0.0	1427141	14274 XII	0.295	24.55M
1	W2-W1F	TH2-TH1F	F RM2-RMIF	8-0 482-RB1F	0.0 LW2-LW1F	0.0 LA2-LA1F		
	COVER-DEMMO	4.0000000000	4-20000000 01	0.0				
1	90	0.0	3.53631609 07	2.34024680 09				
136 1.5	1.50000000 00	4.000000000 00	2.500000000 00	0.0		1		
134	3000000	1.20000000 01	1	0.0				
1 7	00	0.0	2.62786720 00	0.0	0		*	
MEN.	* 2.36026680 09 X			10 000000				
MTRIA	N NSTAG N	N NSTAG NSUCC	D					
	2.5000000000	20.00000000	00000	1				
	1		-		Ī.			
	0.10000000	0.0						
	P(1) 1 - 1 TC 2	2,00000000 01	10 0					

MASIC DECOV CHAR	DECOY CHARACTERISTICS					1	2. 1(34)	
,	THETAI			TAI				
40.00	7.16	TURN	20N 20FF	29.00	0.0 TON :DFF	0.0 OFF 15P	P NGEON LP	
	9.0		İ	0.0			4 - 4 0	
	118	WELDETTY	DECELERATION	- NEW			1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	
300000	0.0			298.62				
2900000				352.11		1		-
220000.0	2.2	23.028.01	0.32	104-63	4400	The second secon		
260000.0	5.07			683,12				
250000-0	#	11111		198.14		-		1
240000-0	1.59			919.28				
ж.		23090.33	1	1050+06				
220000.0	10.11		0.24	1330.10				
200000.0	12.62		0.25	1634.74				
•	13.13			2430.47			L.L.L.	-
180000-0	15.12			3128.11				
178600.0				3527.26	12.1	-		-
160000.0	17.62	23161.00		3941.87				
0.00001	20.11		0.05	4795.30				
	31.34			E104-67			1 1	4
12000.0	22.60			5545-12				
	A		The state of the s	5819.41			1. 一丁 日子 日子 一 一 上	100
100000	25.08		•	5971.17				
****	***			5974-14		and the second s	The second secon	-
20000.0	27.57	22 943.94	-3.71	5557.81				
***************************************	2444	***************************************	-			CT TANK	MAKELLE	j
300000	0.0	1. 000000000	1.090909090 00		0.0	0.0	0.0	
200000		00000000	1.00000000	1.00000000 00	000	0.0	0.00	
270000.0					0.0	0.0	000	
260000.0	5.07	1.00000000	1.00000000	1.00000000 00	0.0	0.0	0.0	
250000.0	6.33	1. 000000000	1.00000000	1	0.0	0.0	0.0	
240000.0	1.59	1.00000000	1.0000	_	0.0	0.0	0.0	
5300000	-	1. 00000000	1*0000000	1	8.6	0.6	0.0	
220000.0	11.01	000000000	1.00000000	1.00000000	0.0	0.0	0.0	
0400000	6411	00 000000000000000000000000000000000000	000000000	00 00000001	5 0	000	000	
000000	11.11	00000000	1-0000000				0 0	
180000.0	15.12	1-00000000	1.00000000	1-000000000	0-0	0.0	0.00	
120000		1.00000000	1.00000000	- 1		0.0	0	
160000-0	17.62	-1.33492660	0.0	0*0	62.95	0.0	0.0	
150000.0	18.87	-1-34725970	0.0	0.0	49.50	0.0	0.0	
140000.0	20.11	-1.40800770		0.0	68.65	0.0	0.0	
1,0000.0	51,36	-1.51075940	1	0.0	900	0.0	000	
120000.0	22.60	-1.65146370		0.0	113.78	0.0	200	
0 000001	25.08	-2.	9*0	0-0	135.74	0.0	0-0	
0.00001	96.33	-2 - 2 8021110		0.00	141.34	0.0	0.0	
0.000	27.67	-7 04400730						
	6.17	07140040		0.0	0.0	21.00	0.0	

11-141

\$.

INTEGRAL OF I WAKE LI/SIGNAJ++2 1.05214700 05			
WAKE AT INTEGRAL 0.0- LEAVE CORRIDOR AT 0.0			
INTEGNAL_OF-1-WAKE R1/516NA1002 0.17645938 05			
0.0 0.0 0.0 0.0 0.0		0.273	3 3
0.0 W2-W1F TH2-TH1F RN2-RNIF R82-RBIF LM2-LM1F	142-1416		
LOWER SOUND UPPER SOUND OCCURALIZED PENALTY 0.0 4.00000000 01 4.00000000 01 0.0 0.0 0.0 0.0 0.0			
1.5000000 00 4.0000000 00 2.60000000 00 0.0			
00 1.20000000-01 7.15822310 00 5.00000000-01 3.84615380-02 0 0.0 2.57026190 00 0			
1.98522090 09 X = 2.60000000 00 2.0000	a di salamani in di s	Charles I Market	
N MSTAG MSUCC			
2.600000000000 20.0000000000			
1856AK 0F - 0.0			
F(1) 1 - 1 TO 2 2. 00000000 01			

BASIC DECOY CHAI						The second of the second secon	-		
•	CHARACTERISTICS								
40.00	THE TA	TURN TURN	2:90 2:90	LANDA1	20.00 THO	0.02	THE TAZ	RN2 0.0	0.0
		00 000	0.0	0.0	0.0	0.0	9.0		
ALTITUDE	**************************************	1	TTY DECELERA	LERATION	BETA				
290000-0	1.57	23013.35	35	0.33	241.31				
280000.0	2.54		.53	0.35	301.29				
260000.0	10.5	23052.30	10		521 70				-
250000,0	P. 33		00	0.31	621.85				
240000-0	7.59		.20	0.30	117.73				
23000000	CO. 8	1 1 1	*	0.28	852.19				
210000 0	11.22	22110 04		0.26	852.22				
2000002	12.62		119	0.25	1617.05				
1900000	13.67	TITLE	94	0.24	2105.37				
180000.0	15.12		91.	0.22	2377.64				
1,000000	16.61			6	2631.07				
150000.0	11.62	23153.57	151	0.13	2894.98				
140000.0	20-11		92	-0-07	14.25. 27				
13000000	21.36		13	-0.25	3669.88				
120000.0	22.60		30	-6.54	3883.98				
11000000	53.85	23105.18		-1.02	+052.26				
100000.0	25.09	23050.80	000	-1.78	4148.30				
80000.0	27.59	22789.62	.62	-5-69	3687.76		Total Section	-	
1000000	28.86	22477.04	40	-9.80	1444.07				1 :
ALTITUDE	TIME	WARE			WAKE RS	WKE LI	N. W.		WAKE 13
200000.0	0.0	1 0000000000	000000000000000000000000000000000000000	00 00000	-	0.0		0.0	
280000.0	2.54	1.00000000		00 00000		0.0		0.0	0.0
270000-0	3.84	1. 00000000	1	00 00000		0.0		0.0	0.0
260000.0	5.07	1.00000000		6	1.000000000000	0.0		0.0	0.0
0.000062	6649	00000000		00 00000		0.0		0.0	0.0
230000	n. nr.	1. 000000000	1 000	00 0000				0.0	0.0
220000.0	10.11	1-00000000	000000001 00	00 0000	1.0000000000000000000000000000000000000	9 6		200	0.0
210000.0	11.37	1.0000000		00 0000					0.0
200000-0	12.62	1.00000000		00 0000		000		2.5	200
190000.0	13.67	1.00000000		00 0000		0.0		200	000
180000.0	15.12	1.0000000		00 00000		0.0		0.0	0.0
170000.0	15.31	1.0000000	00 1.0000000	00 0000	1	0.0		0.0	848
160000.0	17.62	-1.15904850			0.0	10.08		0.0	0.0
1500000	18.81	-1-18430200			0.0	62.44		0.0	0.0
130000	21.34	-1.25521940			0.0			0.0	0.0
1200000	22.60	-1 52 802 940	000		0.0			0.0	0.0
110000-0	20.65	1.72462170			0.0	133.46		0.0	0.0
1000001	25.09	-1.95248640			0.0	130.58		0.0	0.0
0.00000	26.34	_			0.0	140,31		0.0	0.0
800000	27.59	-6. 6469837	0.0			-			

INA 1442 8.22251520 05 THE THIS THIS 6.0 0.0 THE BORNO 01 0.00000000 01 9.5 706 3750 06 0.0 0.0 0.0 0.0 0.0 0.0 0.0
8.22251520 05 11
W2-WIF TH2-THIF RNZ-RNIF R82-R83F LH2/LH1F LA2-LA3F G-0 G-220 G-220 G-0 G-220 G-0 G-220 G-0 G-220 G-0 G-0 G-220 G-0 G-0 G-220 G-0 G-0 G-0 G-220 G-0 G-
W2-W1F
34ER 365ND UPPER BOARD 0ECURITE 4, 00000000 01 0.0 0.0 0.0 0.0 0.0 0.0 0.0
11 1.20000000 01 2.0000000 01 0.000 00 0.000 00 0.000 00 0.000 00 0.000 00
3965 0.0 0.0 2.35872320 00 0.0 2.35872320 00 0.0 2.35872320 00 0.0 2.35234340-01 0.0 2.95000000 00 2.950000000 01
WIRTH W NSTAG NSUCC
I = 1 TD 2 2.900000000 20.00000000
0.0 00 00 00 00 00 0.0
P(1) 1 = 1 TD 2 3-80000000 00 2-0000000 01

10000	BASIC DECOY CHARACTERISTICS								
10.04	THE 731	N TO	1000	LAMBAI 0.03	20.00	0.0	THE TA 2	0 8 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	RBZ 0.0
LAMONZ	7	00 00000	3		0.0	0.0	0.0		
ALTITUDE	1186	VELOCI	CITY	DECELERATION	- BETA				7
300000-0	0.0		23000.00	0.32	154-67		7		
280000.0	2.54		23026.04	0.31	171.56				
260000-0	5.07		23050.65	0.29	266.68				
250000.0	6.33	TITI	23062-23	0.28	312.91				
24000D-0	7.59		23073.02	0.26	399.72				
230000-0	8.85		23082.78	0.54	417.49		1		
220000-0	10-11	~ .	2.76	0.25	124-21				
2000000	12.62		74110.41	0.19	927.30				
0.000001	13.86		23117-13	0.14	1019.97				
180000.0	15.13		23121.45	10.0	1105.50				
170000-0	16.38	1	2.11	-0.02	1184.89				
160000-0	17.63		23119.55	-0-15	1262.87				
0.000041	18.8		-	96.0	1339.65				
140000.0	20.13		23090.30	-0.65	1414.95				1
130000-0	51.3		23055-35	1 87	1544.05				
110000-0	20.22	1	*	3.06	1592.26				1 1 1
100000-0	25.14		0.27	4.4	1623.75				
0000006	26.4		32462.04		1444.80			The second second second	
200000	29.03		21985.61	-14.24	1420.23	i			
*******	**	1		WAKE R2	WAKE R3	WAKE L	-	WAKE LZ	WAKE 13
300000	0.	-		000000			0.0	0.0	0.0
200000	. 54	1 000000000	90	1:0000000000000000000000000000000000000	1.00000000 00	0			0.0
000000		•		000000	1	0.0	0	0.0	0.0
250000.0	5.07	: -:		000000		0.0	0	0.0	0-0
250000	6-33	-		000000		0.0	0	0.0	0.0
240000.0	7.59	-				0.0	0	0.0	0.0
230000.0	-	1	1	1	1	0.0		0.0	0.0
220000-0	10.11	-		1.000000000		0.0		300	000
210000-0	11.31	7	200	000000	000000000000000000000000000000000000000			0 0	0.0
2000000	70-71	1-0000000						0.0	0.0
0.00000	15.11	-	12	000000		0.0		0.6	0.0
-	1	7		000000	- 1	0.0	-6	0.0	0.00
160000:0	17.63	.1.		0.0	0.0	158.	33	0.0	0.0
150000-0	18.88	7	-	0.0	0.0	124.26	55	0.0	0-0
140000-0	20-13			0.0	0.0	92.	90		000
13000000	75.13		-	0.0	0.0	40**0	*	0.0	
120000.0		-1.14873570	-	0.0	0.00	40.34	100	0.0	0.0
110000-0				0.00		110.3	31	0.0	0.0
100000		-1.61002690	50	0.00	0.0	0.0		0.0	0.0
0.000006	14403			200					1
	27.49	0 -6- 36101260		0.0	0.0	0.0	0	0.0	0-0

1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1							
	LEAVE CORRIDOR AT	0.0					
	PTO 05	RBZ/RB1F	L#27/L#1F	tar/taif	0.129	24/24	
0.0 -0.0 -0.0 -0.0	PN2-RNIF	R82-R81F	O-0 LEZ-LNIF	0.0		, , , , , , , , , , , , , , , , , , ,	1
133 0.0 4.00000000 01 4.00000000 01 2945 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.	000000 01 0.0 122620 06 1.04 0.0 0.0 000000 00 0.0	290300 07			7	1 = 2	
00 1,29000000 01 5,0000000000 0,0 0,0 90300 07 X = 3,800	2.000	10 0		L.			
NTRIA N NSTAG NSUCC 3 10429030, 4786984500 P(1) I = 1 TO 2 20,000000000000000000000000000000000	30						
8050RK 0F 0.0							
P(1) 1 = 1 T0 2 6.50000000 00 2,00000000 01							

3								
10.00	THETAL	0.10 6.	R81 LAMDA1	20,00	0.0 THETA2	11A2 0.0	2 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6	
LAWDAZ	7						MGE CUP	1
ALTITUDE	3WIL	VELOCITY	DECELERATION				1	1
300000-0	0.0	23000,00	0.31					
280000-0	2.54		0.24	2016			1	
270000.0	1846		0.21					
260000.0	2.07	23039.96	0.12	54.84				
250000.0	6.33		10.0	* * 1 1				1
240000-0	09*1	23046-84	90.0	102.27				
220000-0	10-12		0.00					
210000.0	*****		25		The second secon			
2000002	12.64	23026.19	74.0-	173.55				
19800000	13.89	23 001.29	-0.78	180.46				
180000-0	15.15	22961.03	-1-22	186.39				
170000-0		55900.00	**************************************	191.55			1	
160000-0	17.68	22809.19	-2.74	196.27				
. 40000	20.00			15.00				
120000	31.61	32147 10	100	16.402				
120000-0	22.83	21646.59	-15.13	105.34				
110000-0		20814-64	-23-38	188-50				
10000001	25.59	19533.05	-32.61	190.93				
0.00000	-	17663-02	12.00	192.67				
70000.0	30.99	11581,89	-50.14	191.99			1	1
ALTITUDE	114	WAKE RI		WAKE RA	WAKE !!	S. H.		
300000-0	0.0	1.00000000 00	2					
280000-0	7.54	1_000000000	1-00000000	1.0000000000000000000000000000000000000		0	0 0	-
270000.0	3.61	-	9		0.0	0.0	200	
260000.0	5.07				0.0	0.0	0.0	
250000.0		1. 000000000 00	1.000000000 00		0.0	0.0	0.0	
240000.0	7.60	1.00000000 00		1.000000000 00	0.0	0.0	0.0	
2300000	30.0	1.	1		0.0	040	0.0	
0.000022	10.12		2		0.0	0.0	0.0	
0.000000	13 61	00 000000001	1.0000000000000000000000000000000000000		0.0	0.0	0.0	
190000	13.80		2 9	00 0000000	0.0	0.0	0.0	
180000-0	15-15		1.00000000 00		0.0	0.00	0.0	
170000.0	1	1	4	- 1		0.0	0.0	
160000.0	_	3032782D			839.74	0.0	0-0	
150000-0	4		0-0	0.0	669.51	0.0	0.0	
14000000			0.0	0-0	445.71	0.0	0.0	
13000000		30327820	0.0	0.0	330.11	0.0	0.0	
120000*0	_		0.0	0.0	0.0	0.0	0.0	
00000	1	1	0.0	0.0	6640	6.6	0.0	
100000			0.0	0.0	0.0	0.0	0.0	
0.0000			0.0	0.0	0.0	0.0	0.0	
	28 86	TO THE PARTY OF TH					9 1	

DOR AT 0.0
0.0000

BASIC DECOT CHAI	CHARACTER! STICS												1
10.01	THETAI	1	0.10	en	#81 9×80	LAMBA1	- 21-600	2 11 6	T# 142	2 × ×	٠.	# d	
LMOAZ	E LES	3	TURN OG GO		NO. 2	100		ž ,	TOFF	8.0	SP RECEUM		
46711906		*		WELDCITY	T DECELE	15841100	- 118	ì					
300000-0		0:0	230	23000.00			error (
280000-0		2.54	23.	23 026-05		0.31	175.51						4
260000-0		9	220	2000		3						The same of the sa	
250000-0		1	230	23062		6.23	130 00						
240000.0	-	1.50	230	23073.4	الميس ا	0.26							
230000-0		1	23	23063.		0.24		1					
310000.0	11	10.11	162	23092.26		61.0							
2000000	12	3	231	23106.21		0	774-87	!			The second secon	-	
1000000		13.4	165	23 11 2 . 11	-	0-13				,			
0.000081	=	15.13	231	23116.85	100	0.0	-						
		:	231	23119.05		6	***						1
1000000		17.03	7.91	73116.91		-0-9	1369.63						
140000-0	. **	20.13	230	23092.32	1	9	1567.67						
130000		41.44	230	23061-69		*	4 44						
120000.0	13	22.62	230	23 000 42		-1.66	1701.20						
0.00001	報	3:	554	22-922-80		7.7	- 1760.02						-
0.00000	1		200	22.00.12		11	1799.05						
80000-0	K	9	221	22101.20		-13,13	1553.39						
1000000		23.82	- 214	21411.09		-19.89	1576.26						
ALTH-TUBE	THE REST.	**	**	RE RI		WAKE RE		MAKE	ودند. المد	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	***	818* 1404	
300000-0	(3) H	0.0	1.00000000		1.000000	8	1.00000000	0	ig.	d		10	
280000.0	~	7:5	1.000000000	00 00		00 00000	1.09000000 00		3 - c	5 6		0.0	-
27000000	1111	3-41	1. 00000000	_	**		7	0.0	ф	d		> 4 5 d	
260000.0	er)	5.07	1.000000000		-		1.00000000	0.0	Q.	9*0		0.0	
2000000			-1-0000000	1	-	9	1.00000000	0.0	4	± ±		\$\$*\$\$	
0.00000			1. 000000000	00 00	1.000000	00 00000	0.0	0.0	o i	o			
220000-0		10.11	1.00000000			00 0000	1.00000000	0.0	0.0	0.0	100	÷.	and the same of the same
230000.0			1.00000000	1	-	00	1	00	0.0) E		200	
2000000	ethy-	12-62	1.00000000		-	60		0	0*0	d		e co	
140000-0	\$	- 23 · EE	1.00000000	1	1	00	1.50000000 00	0.0	C)	E -0		0.0	
0.00000		15-13	1 000000000	00 00		00 00000	1.09000000 00	0.0	o i	o d		0*0	
40000	**		7 41464370	2	:	20000	- 000000000 00 C		8.	4		6.0	
150000-0		11.25	95.050.550				0,0	144.10	0 *	င) င ကို င		00	
140000-0	2		82259040				0.0	81.80		n • •		6.6	
130000.0		- 1	01475140	-			0.0	.04) c		3 c	
120000.0	77		19253430				0.0	86.10		5 6		* 4 5	
110000-0		J.	400014	1			0.0	106	*	0.0		0.0	
100000	K 3		65335650				0.0	122.45	45	0 0		6	
0.00008		77 . 48	10344 310	30			0.0	0 °	O	Ф * ф		6.0	
										()		* *	

11-149

	TELOGIA MATERIAL	0.00	THE COUNTY OF				The second secon	
FERRI O	F I VELOCE	INTEGRAL OF I VELOCIY/SIGNA 10+2 2.	2.77961430 04					
ME 11 1	NAKE LI INTEGAL .	0.0	LEAVE CORRIDOR AT	AT 0.0	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1			
ISEAN O	-	INTEGRAL OF 1 MAKE LIVSIGNAIMS 7. MINSRIPD ON	1058240 04					
WE HIL		1	LEAVE CORRIBOR AT	64 6.6 TA	1	ia i		
FERRI -3	F-L-MAKE A		19265690-05		· 1	*		
***	0.0	0.0	BNE/RHLF	0.0	Luzvenir 0.0	0.0	0.123	0.0
	0-0-01F	TH2-THIF	RN2-RNIF	RB2-RB1F	1H2-LH1F	LAZ-LAIF		
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	and the second		C090=	0.38290-01		COFINFIBL, LAM, NBJ=	- 1		=100	0.0
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rime- 5.07	:££	0.2600000 06 0.8736716 05	182	0.23049790 05	CANT.	0.25670 02	KR= 0.1082255 V00756- 9.28649 PSIALP= 0.0	250 06 649 90	8E TA: 8E TAP	0.23380 03
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	5	0.38540 00	C086-	0.38300-01	500	INFIBL, LAN, WBJ- INFIBL, LAN, WBJ- INFIBL	0.0		C08- C01-	0.0
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Ť					MASS LOSS					
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	11	0,10000	90 00	THETA	0.77070	CONFIGURATION 01 LA 0	0.23000 02	1 AMBDA - 0.3	0.31250-01	AREF	0-22340 00	
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-	-63	0.22060 00	CDP*	0.38380-01		LAM, WB) =	0.0	C080	0.0
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CDP- 0.3440D-01 CDFINF(BL, LAM, WB)= 0.0 CDP- 0.3840D-01 CDFINF(BL, LAM, WB)= 0.49450 05 X8AR- 0.4550 04 CFFFELL LAM, WB)= 0.49450 05 LAMBDA 0.1000 00 TMEIA- 0.71070 01 LA- 0.2700 02 LAMBDA 0.31250 01 W= 0.20480 02 DELW= 0.0 MASS LOSS CONTISTAGIS 0.0 WASS LOSS MASS LOSS MASS LOSS MASS LOSS HSRT0= 0.3182D 03 FFFSER TABLE PERGENTAGES OF UNMBLATED LEWTH TABLE 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.							QUANTITI	5					
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	500	0.19020 00			0.38420-01		BL. LAN. WBJ-		. CO 6	0,11600-02
			XB		0.43430 01	3	#	0.68940 05	XRAR	- 0.62750-01
			3	C01/P= 0.	0.14369-01			0.59850-02	21/102	C= 0.68160-02
1					8	CONFIGURATION				,
	**	0. 10000 00		THETA- 0	0.77070 01	14.	0.23000 02	LAMBDA = 0.31250-01	50-01 AREF.	0.22340 00
	1				1	MASS LOSS		Į Į		
SWOTTATE R	48E PERCENTAGES OF UN	* 0.19790	UNABLATED		QDUT (SONIC)*	0.0	HSH	HSRT0= 0.31840 03	*0 dS d	0.12260 00
	TANG. PT.	X/LA=0,2		8.71.4=0-	4-0-	X/LA-0.4	KALA-0-75	Kfi.A.O.S.	X/14=1.0	BH-H 34(7)
PEPSB= MDOT= P-TMT=	0.69909-01 6.44240 63 6.38560-01 6.27760 63 0 INT(STAG)	0.19330-01 0.49280 02 0.36350-02 0.30890-02	0-01 0-02 0-02 20 04	0.22650-01 0.37670 02 0.26970-02 0.23640 02		0.22650-01 0.30760-02 0.20970-02 0.19300-02-	0.22650-01 0.27510-02 0.18600-02 0.17270 02 INT(SONIC)=	0.22650-01 0.25110 02 0.15750-02 0.15760 02 0.0	0.22650-01 0.23830 02 0.14510-02 0.14950 02 TURBULENT ONL	0.22659-01 0.23839 02 0.0 0.14959 02
					TRANSLATIONAL	2 1	QUANTILLES			
TUNE - 13.88	24 - 0. 178 - 0.	0.1960609 06 0.8737900 05			3099670 05 0.20450 03 0.0	SAN TANK	0.21790 02	XR= 0.2970620 V9010G= 0.61130- PSIALP= 0.0	20 06 BETA** 30-02 BETAP	0.33350 00
					DRAG	19 QUANTITIES	63			
					20,000				C38=	0-13150-02
	å	0-14950 00		KBAR= 0	0.38430-01 0.38430-01 0.35840-01 LAMINAR 0.11108-01	COFINE BL. COFINE BL. RETINELA- COI COMPONENTS	BL. LAM. NB)=	0.71890-01	CB1776	
		1			2	CONFIGURATION				
1	##	0.10339 00		THETA 0	0.35420-02	1.A=	0.35420-02	LAMBDA - 0.32270-01	70-01 AREF=	- 0.22340 00
						MASS LUSS				
	QDD1(STAG)= 0.23430	= 0.23430		8	SOCT (SONIC)=	0.0 =	HSR	HSRT0= 0.31850 03	=0 d\$ d	0-17845 00
1045	TAME PERCE	•	0-2 X/			x/LA=0.6	X/LA=0.75	X/LA=0.9	X/1.4=1.0	CONF R=RB
	0.00000	1		07.454		271110 02	0.33190 02			
- 1000 - 1001	0.47130-01		0-02	0.333	-05	0.26500-02	0.23120-02	0.20550		0419130
	0.88439 03		0 02	0.15770		3.5196U UZ	0.55320 02	20 00505 05	0.47910 02	0.47910 02

TABULAR INPUT ANGLE OF ATTACK ALPHA =	0.0
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BEIBE 0-EDS60 03 BETAPE-0.1964D-01	0.36440 00		a	00 %	00-01	200		\$ 5 5	70-07	65 6
	0/4= 0/4=	4 5	arefi 0.22340 00	- 0.25730	0.2274	NT DNLY GABTELL		8614= 9,10500 86748=-0,24970= 0/#= 0,40400	CDB- 0-16650-02 CDI- 0-12130-01 XBARI 0-36220-01 CDI/IC- 0-22010-02	######################################
XR= 0.323982D 06 VDD10C=-0.2412D-01	V00T06=-0.2412D-01 PSIALP= 0.0	0.59800-01 0.13900-06 0.27280-02	LANBDA= 0.34450-01 #THRST= 0.0	0.31860 03	000	0.92250 02 0.8751D		жq= 0,350889D 05 vD0106=0,63150-01 PSIALP= 0,0	0.25110-01 0.45720-01 0.19660 06 0.19230-02	LAMBDA = 0.37180-01 #THRST= 0.0
ANTITIES -0_2022D_02	21540 02	01 COFINE(BL LAM, NB)= 01 REVINELA= 01 REVINELA= 02 CDI/SFA CONFISURATION			5550	03 0.11300 03 0.10110 03 0 INT(SONIC)= ANGLE OF ATTACK ALPHA = -0.0	NAL QUANTITIES	GANF# -0.20240 02 N MINE 0.21350 02 V YR= 0.0 P	01 CDFINF(BL, LAM, MB)= 01 COFINF(BL, LAM, MB)= 01 REYINFLA= R_CDI COMPONENTS 02 CDI/SF= CONFIGURATION	LA= 0.22380 02 LASE 0.22380-01 %
TRA 6.230993	0.2950D 03	CDP0= 0.38430-01 XBAR= 0.29710 01 LAMINAR CDI CD1/P= 0.94610-02	THETA= 0.77070 01 DELN= 0.11870-01	ONIC)	X/LA=0.4 0.22740-01 0.54620 02	4. O.13850 ULAR INPUF	TRANSLATIONAL	V= 0.23097620 05 00= 0.42420 03 TXT= 0.0 DRAG	CDP= 0.38440-01 CDR0= 0.38460-01 XBAR= 0.2469D 01 LAMINAR_CDI CDI/P= 0.79960-02	THETA= 0.7707D 01
00000001	0.8736160		0,11020.00 0,20470.02	4001(STAG) = 0.27300 04		A61= 0-7136	-	0-1700000 06 1- 0-8733990 05	0.87230-01	0,11900 00
IIMER 15-13 Zm	27R# 14#			QDOT(STAG)= 0.2730D		Q INT= 0.15960		71ME* 16.38 Z* Z1R	CO=	9) 32 11 0x 38

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### FIRST CONTRICT OF THE PART AND THE PART AND THE PART OF THE PA			0031(STAG1- 0.31500	161- 0.			DOOT (SON!	(SONIC)= 0.0		HSRT0=	0= 0.31865 03	* CdS d		0.36990 00	
17.63 2. 0.100000 0. 0.10000 0. 0.	1	X STATION	TANG. P	ACENTA	0	•	A-0.4	X/LA=	9.0	X/LA=0.75		X/LA=1.		COME R#RB	
17.03 Z	I	9001 -	0-69910		0-17460-01	0.65	170-01	0.5334	0 05	0.47670 02	0.43490	0.41240			
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THE 0.0 PSIALPE-0.0 DRAG QUANTITIES CDP 0.70230-01 CDP 0.38440-01 CDFINFIBL. LAM.WB)= 0.20480-01 CDB= 0.20440-01 CDF= 0.20440-01 CDF= 0.20440-01 CDF 0.20440-01 CDF 0.20440-01 CDF 0.20440-01 CDF 0.20440-01 CDF 0.20440-01 CDF 0.20440-01 CDF 0.20440-01 CDF 0.20440-01 CDF 0.20440-01 CDF 0.20440-02 CDF 0.20440-02 CDF 0.20440-02 DF 0.20440-02 DF 0.20440-02 DF 0.20440-02 DF 0.20440-02 DF 0.20440-02 DF 0.20440-02 DF 0.20440-02	1	1		0.160	88	0	3093940		1		3	90	ETAR		
CD= 0.70230-01 COP= 0.3846D-01 COFINFIBL LAM.WB)= 0.2048D-01 CDB= 0.2046D-01 COFINFIBL LAM.WB)= 0.2046D-01 COFINFIBL LAM.WB = 0.2046D-01 COFINFIBL LAM.WB = 0.2046D-01 COFINFIBL LAM.WB = 0.2046D-01 COFINFIBL LAM.WB = 0.2046D-02 CDI/TC= CDI/TC= CDI/F= 0.6559D-02 CDI/TC= CDI/TC= CDI/F= 0.6559D-02 CDI/TC=			T.	0.0			0.0	4		0.					i
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CONFIGURATION CONFIGURATION CONFIGURATIONAL CD-CONFIGURATION CONFIGURATIONAL CD-CONFIGURATIONAL CD-CONFIG	1					1		1	TINFLA			×	G.	0.36010-01	
AME PERCENTAGES OF THETA- 0.7707D 01 LA* 0.2281D 02 LAMBOA = 0.4050D-01 AREF= 0.2044D-02 OELW= 0.3506D-01 ARE = 0.3506D-01 ATHRITE 0.0 ATHRITE 0.0 BETA= 0.2044D-02 OELW= 0.3506D-01 ARE = 0.3506D-01 ATHRITE 0.0 ATHRITE 0.0 ATHRITES MASS LGSS	1				1			3	OMPONE 31/SF=	MTS	0.13250-02	Ü	D1/1C=		
NASS LOSS LAMBDA - 0.40500-01 AREF= 0.35060-01 471451= 0.0 NASS LOSS - 0.22610 02 LAMBDA - 0.40500-01 AREF= 0.0 NASS LOSS - 0.22610 02 LAMBDA - 0.40500-01 AREF= 0.0 NASS LOSS - 0.22600-01 AREL= 0.35060-01 471451= 0.0 NASS LOSS - 0.35060-01 AREL= 0.35060-01 471451= 0.0 NASS LOSS - 0.35060-01 AREL= 0.22770-01 0.2770-01 0.2777	+					1		CONF 1GL	MAT 108						
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ARE PERCENTAGES OF UMABLATED LENGTH ***********************************			100		1			- 1		454	0.31850]	2000000000	
2- 0-1500000 06 V- 0-2308510D 05 GAMF- 0-20280 02 XR- 0-4046590 06 BETA- 2 TABULAR INPUT ANGLE OF ATTACK ALPHA = 0.0 TABULAR INPUT ANGLE OF ATTACK ALPHA = 0.0 TRANSLATIONAL QUANTITIES 2- 0-1500000 06 V- 0-2308510D 05 GAMF- 0-20280 02 XR- 0-4046590 06 BETA- 276- 0-8728270 05 QD- 0-92080 03 MINF- 0-21510 G2 VNUTOG0-30520 00 BETA- 276- 0-8728270 05 QD- 0-92080 03 MINF- 0-0 PSIALP-0-0 DRAG QUANTITIES		PEPSB-	TANG 0.00	HCENTA POST	1000	-	2770-01 2770-01 8990 02	K/LA 0.227	70-01	X/LA=0.75 0.22770-01 0.57400 92 0.42720-02		X4.4-1	055	CONE D-RB 0.22770-01 0.49648 62 0.36680-02	
Z= 0-1500000 06 V= 0.2308510D 05 GAMF= -0.2028D 02 KR= 0.404659D 06 BETA= ZTR= 0.872827D 05 QD= 0.9208D 03 MINF= 0.2151D GZ VNUTOG=-0.3052N 00 BETAP= ZTR= 0.872827D 05 QD= 0.9208D 03 MINF= 0.2151D GZ VNUTOG=-0.3052N 00 BETAP= ZTR= 0.00000000000000000000000000000000000			O INT	\$1461=	0.15070		4		\$	INTESONICE	200	TURBULEN	S S		
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2= 0-1500000 06 V= 0.2308510D 05 GAMF= -0.2028D 02 XR= 0.404659D 06 BETA= ZTR= 0.87282TD 05 QD= 0.9208D 03 MINF= 0.2151D G2 VNGTOG=-0.3052N 00 BETAP= ZTR= 0.87282TD 05 QD= 0.9208D 03 MINF= 0.2151D G2 VNGTOG=-0.3052N 00 BETAP= ZTR= 0.87282TD 05 QD= TATE 0.0 DRAG QUANTITIES		1					TRANS	LATIONA		IIIES					
DRAG QUANTITIES		7 IME 14.	27.2 27.2	4			0.92080	88			5	900		0.14230 04	
								RAG	ANTITI	83					

			V BABA	0 0 0 0 0			-		
	1	7		0.5467D-02	CD1/SF=	MTS	0.89180-03	CD1/TC=	C* 0.10130-03
-				-	CONF ISURATION				
		0.14200 00	TMETAS 0.72	O STOTE OF	-	22730 02	4	66 360-01 APEFIL	0.22340 00
	**	0, 20430 02	DEI N. 0.50	0.50380-01	WABLE O	0.50360-01	THRST= 0.0		
				*	MASS LOSS				i
		0.42440 34	1000	MIC)=	0.0	HSR TO=	3= 0.3182D 03	#045 d	0. 00310 00
PERSO.	TANG. PT. C.49910-01 0.94910 03	PT. X/LA=0.2 210-01 0-15790-01 310 03 0-11490 03	X/LA-0. 0.22740 0.96020	+58	X/LA=0.6 0.27740-01 0.78010 02	X/LA=0.75 0.22740-01 0.69640 02	X/LA=0.9 0.22740-01 0.63480 02	7/LA-1.0 0.22740-01 0.60190 02 0.44870-02	COME R=RB 0.22740-01 0.60190 02 0.44870-02
POOT P	0.44720 04	0.51710 03	0.41340 03		0.33680 03	0.30090 03	Т	0.26040 03	0.26040 03
		M-1	ALAR IN	PUT ANGLE OF	ATTACK ALPHA	H PHA = 0.0	1		i
-							İ	j	1
			1	TRANSLAT IONAL	100	QUANTITIES			
TIME= 20.13	7. 2. FF.	0.1400000 06	V* 0.73064	3068120 05	GAMF.	0.21850 02	XR= 0.4315220 06 400106=-0.56860 00 PSIALP= 0.0	20 06 BETA*	0.1542D 04 0.11950-01 0491140 00
				DRAG	QUANTITIES	12			
	-63	0, 59250-01		0.38590-01	COFINFIBL.	1	0.13110-01	# COB	0-18770-02
			XBAR- 0-1	0.13760 01	REY INFLA-	A. LAM, NBJ-	0.67090 06	XBAR1=	
			CD1/P- 0-4	0-44240-02	1-		0.58330-03	C01/TC*	C= 0.6652D-03
				CONF	CONF. IGURAY ION				
	× 3	0.15650 00	THETA- 0.7	0.77070 01	-184	0.22640 02	LAMBDA = 0.48910-01	10-01 AREF.	0.22340 00
		. 1		2	MASS LOSS				
X STATIONS	GOOTESTAGE	GDD1151AG1- 0.4986D G4	ABLATED	NIC	0.0	HSRT0=	-	#045 d	0.12260 01
		X/14-6-2	*	1	*/FY-0-4	XALASON IS	10 00000	10.00700	10 08366 n
PEP56*	0.69900-01		0.21		0.22680-01	0.22680-01	0.22680-01	0.73770 02	
- 1000	0.11150 64				0.95750 02	0.64840-02	0.58650-02		
MD07=	0.57610 04	0.67560			0.44530 03	-	0.36299 02	TURBULENT ONE	0.3441D

Control Cont		Ėŧ	0.0	0.8720410 05	181	0.21740 04	WINF-	0.22180 02	KA= 0.4583710 VDD105=-0.97190 PSIALP= 0.0	900	0.16530	\$ 5 5
Color Colo						DRA	1	163				
No. Control		5	1	10-02255	100	140-04	5			CD8	1	88=
NASS LOSS NASS					47103	Ī	2/102		0.37720-64	193	1	20
### Calibo Co					Ti Li	103	# IGURAT					-
MASS LOSS MASS		#:	99	20390 02	11E 14	6.0	TV9F	0.92929-01	LAMBOA- 0.54248		0.22340	Q
#### CANADA CONTISTACE C.18910 04							MSS LOS					
THE THE FOLLOWING THE STAND OF THE PARTY OF THE STAND OF		900T(S)	TAG 1-	0.58710 0			0.0	HSR	0.31670	0 45 4	0.18960	
TABLES OF 0.19950 03 0.11940 03 0.11910 0? 0.10930 03 0.9980 02 0.90950 02 0.90950 02 0.90950 02 0.90950 02 0.90950 02 0.90950 02 0.90950 02 0.90950 02 0.90950 02 0.90950 03 0.91700 03 0.	SHAM	TANG.		X/14-0.			V.A-0.6		X/LA-0.		175	193 ±
22.43 Z = 0.1200000 06 V = 0.2299400 03 0.47700 03 03 03 03 03 03 03 03 03 03 03 03 0	-	200	1	0101		1	0 01811				0.90850	2
TRANSLATIONAL QUANTITIES 22.43 Z* 0.1200000 06 V* 0.22987400 05 54MF= -0.20340 02 K2= 0.46950 01 66 5FTA* 0.17510 04 6FTA* 0.17510 05 05 05 05 05 05 05 05 05 05 05 05 05	- INT -	0.727	35	0.15680	355	55	95160-0		73830-02 47140 03		0.44700	24.00
22.63 Z= 0.1200000 06 V= 0.2298740D 05 GAMF= -0.22934D 02 K2= 0.485205D 06 BFTA= 0.986950-02 THE 0.0 TITE 0.0 PRAC QUANTITIES CD= 0.52050-01 CDF= 0.39000-01 CDFINEIBL: LAM.WB)= 0.17770-02 CDB= 0.186150-02 KBAR= 0.8991D 00 KEYINELA= 0.17770-02 CDB= 0.186150-02 KBAR= 0.8991D 00 KEYINELA= 0.17770-02 CDB= 0.186150-02 KBAR= 0.8991D 00 KEYINELA= 0.17770-02 CDB= 0.186150-02 KBAR= 0.8991D 00 KEYINELA= 0.17770-02 CDB= 0.186150-02 CDI/F= 0.28790-02 CDI/F= 0.2239D 02 LAMBDA= 0.60540-01 AREF= 0.27520-0 W= 0.2036D 02 UELW= 0.1232D 09 4ABL= 0.1237D 00 THETA= 0.1232D 09 THETA= 0.1232D 09 THETA= 0.1232D 09 THETA= 0.1232D 09 THETA= 0.1232D 09 THETA= 0.1237D 09 THETA=		-	1	1	1	TRANSLAT	TONAL QUI	WITTES				
COP- 0.39000-01 CDFINFIBL, LAM, WB)= 0.77770-02 CDB= 0.18810-02 CDB- 0.39000-01 CDFINFIBL, LAM, WB)= 0.17020-01 CDT= 0.18810-02 XBAR- 0.89510 00 CDFINFIBL, LAM, WB)= 0.17020-01 CDT= 0.33960-02 LAMINAR CDI COMPONENTS 0.24240-03 CDI TC= 0.33960-02 CDI /P= 0.28790-02 CDI /SF= 0.22390 02 LAMBDA= 0.60540-01 AREF= 0.22340 00 THETA- 0.77070 01 LA= 0.22390 02 LAMBDA= 0.60540-01 AREF= 0.22340 00 WASS 1.055 QUOTISONIC)= 0.0 488- 0.12370 00 WIHIST= 0.0 HSRTO- 0.3152F 03 PSFU- 0.29770 01			0.12	90 00000 0	185		SAMF.	0.22550	KRR 0.4852057 VBOTOG=-0.16030 PSIALP= 0.0	90	0.17510	419.00
COP- 0.39000-01 CDFINFIBL, LAM, MB)= 0.77770-02 CDB= 0.18810-02 CBP- 0.39000-01 CDFINFIBL, LAM, MB)= 0.15920-01 CBF= 0.33960-02 CBF= 0.39500-02 CBF= 0.28790-02 CBF= 0.28790-02 CBF= 0.28790-02 CBF= 0.28790-03 CBFF= 0.28790-03 CBFF= 0.28790-03 CBFF= 0.28790-03 CBFF= 0.28790-03 CBFF= 0.28790-03 CBFF= 0.28790-03 CBFF= 0.28790-03 CBFF= 0.28790-03 CBFF= 0.28790-03 CBFF= 0.28790-03 CBFF= 0.28790-03 CBFF= 0.28790-03 CBFF= 0.28770-03 CB						DRAC		1165				-
THETA- 0.28790-02 CDI/SF* CONFIGURATION THETA- 0.77070 01 1A* 0.22390 02 LAMBDA- 0.60540-01 AREF- 0.22340 00 OCLW- 0.12320 09 4ABL- 0.12320 00 *THAST- 0.0 WASS-1.055 QUOTISONICS- 0.0 HSRTO- 0.3152F G3 PSFUL 0.29770 01		3		52050-01	C000	000	3		100000	998	200	
THETA- 0.7707D 01 1A= 0.2239D 02 LAMBDA= 0.60540-01 AREF- 0.2234D DELW- 0.1232D 09 4ABL- 0.1237D 00 474457= 0.0 MASS 1.08S QUOTISONICY- 0.0 HSRTO- 0.3152F 03 PSFU- 0.2972D 0					100	0.00	# IGURAT	10k	20-047470			
QDGT(SGNIC)+ 0.0 HSRTD+ 0.3152F G3 PSPG+	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	1:	1	20360 02	THETA	0.12320					0,22340	Q
GOOT(SONIC) - 0.0 HSRTD+ 0.3152F G; PSPU-												
		900118	TAG 1-	0.69260 0		SOUT (SONIC)	0.0	HSH.	0.3152F	PSPG		

PEPS B-	0.16400-01	0.14000-01		0.18810-01	0.22580-01	0.22580-01	0.22580-01	0.22580-01	0-22580-01	
0.141e	0.17130 00 0.90480 04 0 INT(STAG)	0-20930-01		0.15175-01	0.12720-01	0.10840-01 0.66430 03 INT (SONIC)=	0.95710-02	-	0.57410 03	
		TA	TABULAR INPUT	INPUT ANGLE	OF ATTACK	ALPHA - 0.0		, , , , , , , , , , , , , , , , , , ,		
				Tames	TRANSLATIONAL QUANTITIES	11168		+	1	
1115 23.4	278 - 0. 1100 278 - 0. 5709	0. 8709450 05 0. 8709450 05	18	0.54190	OA MINF	0.22840 02	X8= 0.5120240 VD0T06=-0.26160 PSIALP= 0.0	40 06 BETAE 60 01 BETAP=-	0.18300 04	
	60-		1	1		Ť				
			0403 0403	0.39400-01 0.31280 00 LAMINAR 0.23370-02	OD COFINE AL. OD REVINE AL. OR CONPONENTS	L. LAMENB)=	0.12100-01 0.12100-01 0.27840 07	CDB* CDI* ABARL= CDI/TC*	0.26680-02 0.26680-02 0.10130-01	
	4.	6.21760 00 0.20320 02	1ME 1A	0.77370	CONFIGURATION OU LAST 0	-2224D 02 -1624D 00	LAMBDA = 0.68010-01	10-01 AREF=	0-22340-00	1
					MASS 1.055					
- K STATION		ACE & OF UNA	UMABLATED	LENGTH SON	4IC)= 0.0	HSRT	HSRT0= 0.3128D 03	P 500*	0.47260 01	
900T-	7ANG. PT. 0.45980-01 0.18280 04	0.13840-01 0.26300 03	2777	0.17380-01 0.19880 03	X/LA=0.6 0.22560-01 0.18220 03	0.22530-01 0.16190 03	X/LA=0.9 0.22530-01 0.14730 03		CONE R-RB 0-22530-01 0-13940 03	
-111	0.11160 05 0.11160 05	0.14010 04		0.11240 04	95030 03	0.84750 03 INT(SONIC)=	0.17220 03	0. 73180 03 TURBULENT ONLY	0.73180 03	
		T46	TABULAR I NPUF	SW4	LE OF ATTACK ALPHA	LPHA = 0.0				
			io.	TRANSLA	LATIONAL QUANTITIES	ITTES				
TIME= 25.14	0.1000000	88	V= 0.	0.22770070 05	GAMF	0.5	0	9	0.18810 04	
			TXT=	0.0	₹.	0	VDSIOC=-0.4228D PSIALP= 0.0	0 01 85 148-	0.45730 01	
4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4	CD= 0.4	0.48230-01	-400	0.40030-01	9	. LAM. WB)=	1	# 00	20-0(161*)	
1			XBAR.	0.56439 00 LAMINAR CBI	9	S I I	0.44960 67	XBAKI=	0.21239-02 0.80030-02	
1		ی	CD1/P*	0-16060-02	C01/SF=		0.99850-04	=31/10	- 6-11310-03	

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25		1000	0000000	1	POST 15 ONICH	- 1	WSH .	á	# Sp.0=	0.75010-01
TABULAR INPUT ANGLE OF ATTACK ALPHA = 0.0 1 INTISTACI- 0.61170 05 1 INTISTACI- 0.61170 06 1 I		1	****	1		X/14-0.6	XA. A-0.75	X/LA-0.9	X/LA-3.0	COME R-R9
TABULAR TABURAR TABULAR TABU	-0001-	0.21390				21570 03	0.22510-01	0.22515-01	0.22510-01	0.22510-01
TABULAR INPUT ANGLE DF ATTACK ALPHA = 0.0 TURBULER	**************************************	24700				0-51670-01	0-19740-01	0.17320-01	0.15020-01	0.15020-01
TABULAR INPUT ANGLE OF ATTACK ALPHA = 0.0 TRANSLATIONAL QUANTITIES 27x = 0.0000000 05 v = 0.22551500 05 GAMF = 0.20000 02 XA 0.2555110 0.0 ZTR = 0.0000000 05 pc 0.13510 05 MINF = 0.22910 02 V0DTGG = 0.678110 0.0 ZTR = 0.0000000 05 pc 0.13510 05 MINF = 0.22910 02 V0DTGG = 0.678110 0.0 ZTR = 0.0000000 05 pc 0.1050 05 pc 0.40000 01 CPFHFTEL. LAM.HB1 = 0.20410 0.0 TABLE		O INTESTA	. 0.61170			\$		100000	N O	¥ 0.00 120 00
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- CG C4-7700-O1 CDP O-4096D-O1 CDFINFIBL: LAM.NB1 0-3041D-O2 CDPO 0-4096D-O1 CDFINFIBL: LAM.NB1 0.6826D-O2 CDPO 0-1541D-O2 CDPO 0-1541D-O3 CDPO- 0-1541D-O3 CDPO0-1541D-O3 CDPO0-1541D-O3 CDPO0-1541D-O3 CDPO0-1541D-O3 CDPO0-1541D-O3 CDPO0-1541D-O3 CDPO0-		24 ZTR =				CAMP-		XR* 0.565611 VDOTOG = 0.6781	# e	#114- 0,1\$940 94
CDPG- 0.4096D-01 CDFINFIBL: LAM,NB1= 0.3041B-02 CDPG- 0.4096D-01 CDFINFIBL: LAM,NB1= 0.2041B-02 RAMR. 0.4096D-01 CDFINFIBL: LAM,NB1= 0.6826D-02 RAMR. 0.4096D-01 CDFINFIBL: LAM,NB1= 0.6826D-02 RAMR. 0.4145D-02 CDPT COMPTINENTS CDPT COMPT COMPTINENTS CDPT COMPTINENTS CDPT COMPTINENTS CDPT COMPTINENTS CDPT COMPTINENTS CDPT COMPTINENTS CDPT COMPTINENTS CDPT COMPTINENTS CDPT COMPTINENTS CDPT COMPTINENTS CDPT COMPTINENTS CDPT COMPTINENTS CDPT COMPTINENTS CDPT COMPTINENTS CDPT COM	*	1		- :		IS QUARTITY	res		1	
LAMINAR CD1 COMPONENTS		1	0-47709-01		0-40960-0			0.68260-02	3 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	
### 0.20210 02 DELM= 0.27360 00 MARL= 0.27360 00 JIMEST= 0.0 ### 0.20210 02 DELM= 0.27360 00 MARL= 0.27360 00 JIMEST= 0.0 ### 0.20210 02 DELM= 0.27360 00 MARL= 0.27360 00 JIMEST= 0.0 #### 0.20210 02 DELM= 0.27360 00 JIMEST= 0.0 ##################################			Ī		LAMINAN O. 15410-02	ē	ENTS.	0.63510-04	11/10)	. 0.45219-02
## 0.20210 02 DELW 0.27360 00 WARL = 0.27360 00 JIMEST = 0.0 ##\$\$ LOS\$ QDOT(STAG) 0.11020 05		1		011111111111111111111111111111111111111	83	MF TGURATIC	*			
900T(STAG) - 0.1102D 05 GOOT(SOMIC) - 0.0 HSRTO= 0.3033D 03 PSP 100.5 ARC PERCENTAGES OF UNMOLATED LENGTH		:		1		***	0.21840 02		1	0.23340 00
9007(5746)* 0.11020 45						MASS LOSS				
TANG. PI. X/LA=0.2 X/LA=0.4 X/LA=0.6 X/LA=0.75 X/LA=0.9 X/LA=1.4		QDOT(STAG)	0.11020	44.60	OOT (SONIC)		HSR	0.30330 0	PSPG	0.11790 02
- 0.16590 09 0.36930-01 0.22990-01 0.20160-01 0.24150-01 0.22300-01 0.20910-01 0.16590 05 0.22580 04 0.17280 04 0.14970 04 0.13540 04 0.12340 04 0.11690 09 144757461- 0.74200 05 12890 05 14804.8 INPUT ANGLE OF ATTACK ALPHA = 0.0	- PEPS8*	TANG. PT.	X/14=0.	0.15		4/LA-0.6 -18930-01				CONE R-RE C.22550-01
TABULAR INPUT ANGLE OF ATTACK ALPHA & D.O.	- 100H		0.36930			20160-01				11.0
		9454411	4- 0-74-29B	4 44	1944	***************************************	I		1	

The second secon					-		and the same of th		
				DRAG	G QUANTITIES	5			
	5	0-54670-01	C0802	0.42290-01		COFINE (6L. TURB, MB) COFINE (6L. TURB, MB) REVINFLA-	0.69850-02 0.13600-01 0.11289 08	500 144 144	0.17780-02 6.34240-02 8 0.42930-02
				9	CONF IGURATION	-			
	1:	0.31520 00	THETA-	0.17070 01	***	0.21520 02	4 THRST - 0.0	D-01 AREF	0,22150 00
					HASS 1055				
	DOOTISTAGIA	000115TAG14 0-12490 05		QUOT L'SONIC.	ONICIA 0.85170 0	- HSR TO-	0. 0.26320 03	P 5 P Q=	0, 183.9 02
X STATIONS		ARE PERCENTAGES OF UNABLATED LENGTH	BLATED	ED LENGTH	4.0.4.0.4	X/14=0-75	X/LA=0.9	X/LAs 1.0	COME R#88
PEP58+	0.69890-01	0.16070-01	1	-	0-17480-01	0.20170-01		0.22560-01	0.22550-01
PODOL	A 33780 04	0.48140 03			0.08030 03		00 01770 00	13410	0.13730 00
#DOT -	0.41930 00 0.20330 05 0 INTESTACE	0.72180-01 0.29470 04 - 0.89250 05		0.78720-01 0	0.22920 04	0.13310 00 0.22490 04 INT(SONIC)-	33		V 0.20640
		146	TABULAR INPUT	ANGLE	OF ATTACK	ALPHA - 0.0			A company of the comp
				TRANSLAT	TRANSLATIONAL QUANTITIES	HH65-			
TIME 30 0		90 0000000	4	21482190 05	CAME	-	KR. 0.4191120	90	
	ZTR - 0.	0.0657230 05	- 1X	0.32120 05	VR.	0.0	PSIALP= 0.0	DINE	0-21000 02
		İ	1	DRAG	G GUANTITIES	83			4
		10 00100	-005	10-04144 0		COLINGIAL TURBANES	0.82920-02	-800	0.20030-02
			COPO.	99	1	COFINFIBL, TURB, NB)=	0.19390-01	XBAR1	0.46540-02
	1			8	CONFIGURATION				
		00 05446	11614	40 05055-0	-			10 00 WEEF	0.31810 00
	•	0.19720 02	DELM	0.76100 00	4481.	0. 76100 00	WINKSIE DED		
					MASS LOSS				
	900 T (S TAG) -	. 0.13570 05			0.10990	05 HSR1	MSRT0= 0.27540 03	≥DdSd	0,28010 02
* 5147104	C 405 PERCE	3		TED LENGTH	4.0.4.0.4	# // A=0.75	X/LA=0.9	X/LA=1.0	CONE RorRB
	0.69900-01	0-185-0-01	0	. 10	0-16290-01				0.22640-
*1000	0.43590 04	0.97570			0-11570 04		0.15050	0.15280 04	0.18920 00
#D01*	0.25400 05	1		0.33680 04	0.36980 04	0.19060 04	0.39730 04		0.38720
		1	70		1				

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2.01 2304-17 0.29 174-4 5.07 2304-17 0.29 2.174-4 10.11 2304-17 0.29 2.174-4 12.42 2304-17 0.14 143-13 12.42 2304-17 0.14 143-13 13.13 2304-17 0.14 143-13 14.53 2304-17 0.14 143-19 15.13 2304-17 0.14 143-19 15.14 2304-17 0.14 143-19 15.14 2304-17 0.14 143-19 15.14 2304-17 0.14 143-19 15.14 2304-17 0.14 143-19 15.14 2304-17 0.14 143-19 15.14 2304-17 0.14 143-19 15.14 2304-17 0.14 143-19 15.14 2304-17 0.14 143-19 15.14 2304-17 0.14 143-19 15.14 2304-17 0.14 143-19 15.14 2304-17 0.14 143-19 15.15 2304-17 0.14 143-19 15.16 2304-17 0.14 143-19 15.17 1.0000000 0.1 1.0000000 0.0 0.0 0.0 0.0			0.92	111.62				
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7.59 1.00000000 00 1.00000000 00 1.00000000 00 0.0 10.11 1.00000000 00 1.00000000 00 1.00000000	1 :	1.00000000			9.0	0.0	0.0	
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17-53 1-0805000 00 1-0805000 00 1-0806000 00 84-22 17-63 -1-08637240 01 1-09000000 00 1-09000000 00 84-22 18-80 -1-11879140 01 0-0		1.00000000	000000		0.0	0.0	0-0	
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18.08 -1.11879140 01 0.0		-1.08637240	0000000	0000000	84.22	0.0	0.0	
		-1.11679140	0.0	0.0	45.54	50	5 C	
		-1-19508400	0.0		10.00			

## ## ## ## ## ## ## ## ## ## ## ## ##	12-15-04-20 to 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.	120000.0		-1-46566610 01	0.0	0.0		104.06	0.0	0.0
######################################	-1-1515-12120 0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0	100000.0			0.0	0*0		131.88	0 0	000
		200000			0.0	20	-1	139.87	0.0	0.00
######################################	######################################	0.0000	21-69-17		0.0	0.0		0.0	0.0	0
#### CORRIGOR AT 0.0 ##################################	######################################	-	1		0.0	0.0		0.0	0.0	0-0
######################################	######################################	THE STREET	-	LEANE CORR.	17 800					
######################################	######################################	AL - 0F + - FELDET		10235630 04						and the second s
### ### ### ### ######################	### CONTROL OF THE CO	- INTEGRAL	- 1	LEAVE CORE						
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### ##################################	### NWZYKHIF NWZZKHIF LMZZLHIF LTZZLHIF UTZZLHIF	H-147E#AL-	0.0	LEAVE CORRI	1	0.0				100 miles
#2-#1F 1#2/L#1F #2/K#1F #2/K#1F #2/K#1F L#2/L#1F L#2/L#1F #1/2/L#1F #2/L#1F #2	#2-#15	H -04-1 -44-E-4	1/510HA 1002 8.	37677140 05		1	Ī			
### ### ### ##########################	### ### #### #########################	0.0	THZ 2.1811	KNZZKR 0.0		RBZZKBIF	LAZZERIF	TAZZIATE	19719	34 /28 0.0
#### #50400	0.00 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.	414-2N	143-ти1	1		A-G RB2-RB1F	3.9 LM2-LM1F	1A2-LA1F		
0.00 00 1.2000000 01 7.70709345 00 0.0 5.0000000-01 3.1250000-02 0.0 0.0 2.06022280 00 0.0 1.51502960-01 0.0 0.17ER# = -1	0000000 00 1.2000000 01 7.70709345 00 0.0 5.0000000-01 3.12500000-02 0.0 0.0 1.51502960-01 0.0 0.0 X = 3.20000000 00 2.30000000 0 17584 = -1	000000			1					
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77011	=======================================	210RN	3.20 0.0) 20H 20FF	7.E	0.0 TON	10FF 15	##£@#
ATTTUDE	114		DECEL FRATION	8ETA			
0.000042	12.1		0.11	89.30			
2900000	2.54		0.30	107.71			
260000-0	20.5	07 23048.61	0.27	160.27			
0-00005	\$		0.25	221.34			
240000.0			0.24	255.31			
230000-0			0.18	292,30			
210000-0	11.0.	27 23063.78	0.14	332.45			
0.00000	12.63		-0.03	385.68			
1-000041	13.01		-0.0-	440.14			
180000*0	19.13		11.0-	847.98			
0.00004			91.0	940.04			
0.0000		23087.14	57.0-	1041.65			
0.0000	20.14		-0.79	1232.01			
0.000061	21.30	42	-1.30	1319.85			
120000.3	22.65		-2.08	1397.44			
110000.0	53.90		+1.1	1460,23			
0.000001	25-17	22642.08	-5.33	1499.66		100	
0.0000	27.74		-15.28	1314.06			
10000-0	50.0	1	-55.09	1210-12			
300000.0	0.0	1.00000000 00	1.00000000 00	1.00000000 00	WAKE LI	MAKE LZ	WAKE L
0.00000	177	1 1 00000000	- 1	- 1	0.0	0.0	970
0.000085	2.54	1. 000000000			0.0	0.0	0*0
270000-0	10.6	00 000000000000000000000000000000000000	1*00000000 90	1.000000000 00	0.0	9.6	5 6
0.000032	1000	1.00000000	1-0000000000000000000000000000000000000		0.0	0.0	0.0
240000.0	7.5	1.000000000			0.0	0.0	0.0
830000°	9.0	1. 00000000	1	-1	010	9*8	940
220000-0	10.11	1.00000000			0.0	0.0	0.0
210000.0	11.37	1 * 06000000		.1	0.0	0.0	0.0
0.00000	12.03	1. 00000000	00000000		0.0	ф. ф.	0.0
0.000001	16.13	13 1-000000000 00	1-0000000000000000000000000000000000000	1 -000000000000000000000000000000000000	0.0	0.0	0.0
30000		1. 60000000	90000000	- 1	O.C.	9.0	0.40
160000.0	17.6	-1.0873550	00000000		84.02	0.0	0*0
0*000061	19.89	-1-11958920			65.05	0.0	0.0
40000-0	20.1	-1.19716990	0.0	0*0	54.65	0.0	0.0
0*000061	51.39	-1. 11250400		0.0	83.08	0.0	0.0
200002	22.65	10 07988050 01	0.0	0.0	103.89	0.0	0.0
00000	75.17	7		0.0	171.15	0.0	0.0
0.00000	26.4	-2,16230970	0*0	0.0	135.66	0.0	0.0
6.00008	27.74	7. 04.844.50					
		14 10 10 10 10 10 10 10 10 10 10 10 10 10	200		2.50	0.0	F-64

INTEGRAL OF I VELOCT	OF I VELOCTY/SIGNAINT 2.38429620 05	*36429620 05			THE PERSON NAMED IN COLUMN TWO IS NOT THE OWNER.	The second second	
MAKE LI INTEGAL .	0.0	LEAVE CORRIDOR AT	T 0.0				
INTEGRAL OF I WARE LIFSIGNATIONS		9. 7831 8340 04					
WAKE AT INTEGRAL .	0.0	LEAVE CORRIDOR AT	1 0.0		The second secon		
INTEGRAL OF & WAKE RI/SIGNAJONE 8.35969020 05	I/SIGNA Jee2 8.	35969020 05					
WZ/WIF	TH2/TH1F	PNZ/RNIF	RB2/RBIF	LMZ/LKIF	LAZZLAIF	WIVE	ZA / ZW
0.0	6.0 THE-THIF		0.0	3.3 LM2-LMIF	LA2-LAIF	590.0	0.0
LOWER BOUND	UPPER BOUND	OCCURITZ!	PERALTY				
0.0 0.0 1.5000000 00 1.5000000 01	0.0 0.0 0.0 4.0000000 00	1	0.00				
0.0 0.0 0.0 0.0 0.0 0.0	200000	2.25757710 00 0.0 2.02519810-01 0.0 3.2020000 00 2.30000000 01	10 000				
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BASIC DECOY CHARACTERISTICS	CTERISTICS									
16.36 LAMBA2	THETA1	ZTURN ZTURN	3.01 LAMDA1 3.01 0.03 2.00 2.0FF	11 LA1 13 22,52 140	0 to 0	THETA2 0.0 10FF	O S S S S S S S S S S S S S S S S S S S	REEDM 0.0	N. B. V	
	## THE	VELOCITY	PECELERA	## 9ETA						
200000-0	0.0	23000.00								
280000.0	2.54			120.81	ı.					
240000.0	2.0.5	23.097.54								
250050-0	6.33	23060.37	0.26	14.017						
240030.0	7.59					1.	i	1	1	-
230000.0	9*82	23079.35				+	1		1	j
220000.0	10.11	23086.75	0.16							
200000	67 61	23005							-	
180000.0	12.02	22002.44				*				
180000.0	18.13	22000.07	200							
170000.0	16.38	22086. 65		160						
160000.0	17.64	23080.47	-0-18	•						
150000.0	18.89	23070.23		1						
140000.0	20.14	23051.20	-0-62	1455.44						
130000-0	21-39	23016,38		1566-87	1				1 10 1	- 1
110000-0	22.64	22964.30	7'							
100000-0	25.14	22723.26	25							-
0.00000	24.42	22503.16		1806.63						
0.00008	27.71	22115.51	-12.67	16						
700000	20-02	21 395.81	-21.69	1						
ALTITUME	*11	WAKE P		2000						
300000*0	0.0		1.000000000	00 0000000001 00	0.0	4	0.0	O-O		1
2000000		0000000000	1.00000000	1.00000000	0.0		0.0	0.0		1
270000.0	3.81	1.0000000000000000000000000000000000000	000000001	0000000001	0.0		0.0	0.0		
260000.0	5.07	1		20000000	0.0		0.0	0.0		
250000.0	6.33		1,00000000	00000000	0.0		0.0	0.0		
240000-0	7.59		1-00000000	000000000000000000000000000000000000000			0.0	0.0		
230000.0	58.0		1.0000000	1-0000000	0.0		0.0	0.0		
220000+0	10.11		1.00000000	1.00000000	0.0		0.0			-
21000000	11.37	-	1.00000000	-	0.0		0.0	0		
200000*0	12.62			1,30000000	0.0		0.0	0-0		
1500000	13,88	1	-1	1.00000000	0.0		0.0	0.0		
18000000	15.13	_	-	1.00000000	0.0		0.0	0.0		
170000.0	1	1	-1	Ť	0.0		0.0	0.0		
16000043	17.64	1		-	0.0		0.0	0.0		
15000000	- 1	1	1.000000000 00	1.00000000	7.65		0.0	0.0		
0.000041			0.0	0.0	57.82		0.0	0+0		
0000000		131200310	0.0	0.0	85.1		0.0	0.0		
110000	00 25	10 18689476-1-	0.0	0.0	108.46			0*0		
100000*0	15		0.0	0.0	137.80		0.0	000		-
0.00006			0.0	0.0	140.19		0.0	000		
800000.3	11.11	.19718490	0.0	6.0	0.0			0.0		
1								40.4		

M 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	GAAL OF (WAKE RE/SIGHA 1**2 6.31997310 05 GAAL OF (WAKE RE/SIGHA 1**2 6.31997310 05 GAAL OF (WAKE RE/SIGHA 1**2 6.31997310 05 G.**0 G.*	### CORRIDOR AT 0.0 ##################################	
GANL OF & WAKE RI/SIGNA1942 B.3199731D 05 GANL OF & WAKE RI/SIGNA1942 B.3199731D 05 GANL OF & WALTH BASZARIF HASZARIF	### ##################################	### ##################################	
#1/41 W2/W1F TW2/FWEF RW2/FWEF RW2/FWEF LW2/FWEF LW2/FWEF LW2/FWEF LW2/FWEF W1/41	## ## ## ## ## ## ## ### #############	W2-W1F	
### ### ##############################	LOWER BOOMS UPPER BOOMS 1.4384.0000 01 0.0 1.50000000 00 4.0000000 01 1.4384.0000 01 0.0 1.50000000 00 4.0000000 00 3.00900100 00 0.0 1.50000000 00 4.0000000 00 3.00900100 00 0.0 1.50000000 01 1.20000000 01 2.25226500 01 0.0 1.50000000 01 1.20000000 01 3.25226500 01 0.0 1.50000000 01 1.20000000 01 3.25226500 01 0.0 1.50000000 01 1.20000000 01 3.25226500 01 0.0 1.50000000 01 1.20000000 01 3.25226500 01 1.50000000 01 1.74645090 04 X = 3.00900100 00 2.25226500 01 1.50000010083 22.5226503878	### ### ### ### #### #################	
1.5000000 00 4.0000000 01 1.53840000 01 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0	1.5000000 00 4.0000000 01 1.53840000 01 1.53840000 01 1.53840000 01 1.53840000 01 1.53840000 01 1.53840000 01 1.50000000 01 1.5000000 00 1.50000000 01 1.55224500 01 1.50000000 01 1.55224500 01 1.50000000 01 1.55224500 01 1.50000000 01 1.55224500 01 1.50000000 01 1.50224500 01 1.50000000 01 1.50224500 01 1.50224500 01 1.50000000 01 1.502245000 01 1.50224500 01 1.50224500 01 1.50224500 01 1.50224500 01 1.50	0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0	
0.0	0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0	0000000 00 4.0000000 00 3.00900100 00 0000000 00 1.20000000 01 7.38987190 00 0.0000000 01 7.38987190 00 0.0000000 01 3.3234249 02 0.00 0.0000000 01 2.09258820 00	
1.50000000 00 4.0000000 00 3.00900100 1.50000000 01 4.0000000 01 7.35256500 4.00000000 00 1.20000000 01 7.38987190 0.0 2.0 2.092588210 0.0 0.0 2.00258210 0.0 0.0 2.00258210 0.0 0.0 2.00258210 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.	1.50000000 00 4.0000000 00 3.00900100 1.50000000 01 4.0000000 01 2.2524500 4.00000000 01 1.20000000 01 7.38987190 0.0 0.0 0.0 2.092588210 0.0 0.0 2.092588210 0.0 0.0 2.092588210 1.59224460 1.59224460 1.59224460 1.59224650 1.59224660 1.59224660 1.59224650 1.59224660 1.59224650 1.59224660 1.59224650	0000000 00 4.00000000 00 3.00903100 00 00000000 01 2.5226500 01 00000000 01 7.38987190 00 00000000 01 3.32362190 00 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.	
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0.0 2.09258820 0.0 1.59226168 F = 1.74645090 04 X = 3.00900100 00 IA N NSTAC NSUCC U 17464.5099089967	0.0 2.09258820 0.0 1.59226168 1.59226168 1.59226168 1.59226168 1.59226503878	0.0 2.09258820 00	
1A N NSTAE NSUCC 17464,509908967	IA N NSTAE NSUCC 17464.5098089967 1 = 1 TD 2 -3.0090010083 22.5226503878	= 1.74445090 04 X = 3.00900100 00	
7 01 1 1	1	SUCC 1 - 17464-509089967	
FINAL PERFORMANCE FUNCTION ULAST 0.17460928270 05		0.30103145770 0L0.22522002080 32	

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250000.0	1.33		0.27	276.66			
240000-0	7.59		0.24	319.12			
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220000-0	10.11		0.18	415.53			
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140000.0	20-13	23068-12	15.0-	1541.97			
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18000000	15.13	1.00000000 00	1,000000000		0.0	0.0	0.0
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00000	24-41		0.0	0.0	139.87	0.0	5.0
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00000	27.69	-7.01416430 01	0.0	0.0	0.0	124.1	5*0

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280000.0	2.54	-32-00	0.30		
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240000	2.50	34.00	10.1	3 0	
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0.00000	H*61	-38.00	19.61	0 00-0	
2000000	11.37	-39.00	14.64		
1000001		00.04	20-26		
140000	13.88	-54.00	27.55		
170000.0	14.36	00.00	34.04	9	
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100000	10.67	108-00	16.08	158.00 0.0	
00000		-140.00	62.59		
0.0000	14.07	-180.00	+0-33		
20000	69-12	-190.00	34.60	9	
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2,00000.0	1.27		0.0	645.00 0.0	
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2400000	6.33	-400.00	0.0		
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0.0000	12-62	-400 00			
190000.0	13.88	-380.00		00.00	
18000000	15.13	-360.00		290.00.00	
170000.0	16.38	-340		80.00	
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150000.0	18.88		340.54	00.00	CONTRACTOR OF THE PROPERTY OF
140000.0	20-13	-276-00	238 17		
130000.0	21.38	-252.00	110.00	00.01	
120000.0	22.63	-228-00	104.04	00.00	
110000.0	23.89	-204-00	-122.34		
0.00000	25.14	180.00	121 00	0.000.056	
0.00006	26.41	-153.33	-139-87		
8000000	27.69	-124-67	00.0	00.00	
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				00000	
INTEGRAL OF (WAKE LI/SIGNA) ** 2		9.7338477D 34			
ALTITUDE	TIME LONGR	CORRIDE RV-DE	WAKE OIL HOOF		
300000.0		10.00	10	30.00 DEGRE VALUE LEAVE CORRIDO	ENTER CORREDR SLOPE
2,00000.0	1.27	-60 00	1		
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30.00 0.0	30.00 0.0	30.00 0.0	30.00 0.0	30.00 0.0	30.00 0.0	30.00 0.0	31.60 0.0	33-20 0-0	34-80 0.0	36.40 0.0	38.00 0.0	39.60 0.0	41.20 0.0	42.80 0.C	*****	46.00 0.0	47,33 0.0	0 00 00			BIF LHZ/LHIF	-RBFF LAZ-LMEF		HTY.			1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1			
0.0	0.0	0.0	0.0	0-0	0.0	0.0	0.0	0.0	0.0	13-19	13.52	13.44	12.66	-67.92	-45.35	+43.04	40.04-		1100		RB2/RB1F	WFF RB2-RB1F		PEMALTY		0.0		1	17.5	
-40.00	~60.00	-00-00	-40.00	-40.00	-40.00	00.04-	-41,60	-43.20	-44.80	-46.40	-48.00	04.64	-51.20	-52.80	-54.40	-56,00	-57.33	28.61	-00-00	8.37877140 05	RNZ/RN1F	F RN2-RNE	0.0	OCCUR(12)	0.0	0.0	3-20000000 00	- 1	3-12500000-02	1.51502960-01
3.81	6-33	7.59	8.85	10.11	11.37	12.62	13.88	15-13	16,38	-17.63	18.86	20.13	21.38	22.63	23.89	25-14	26.41	27.69	00*67		TH2/TH1F	TH2-TH1F	0.0	UPPER BOUND	0.0	0.0	** 000000000	1.20000000 01	5.0000000000000	
270000.0	250000-0	240000.0	230000.0	220000.0	216060-0	200000.0	190000.0	1.80000.0	170000.0	150000.0	150000.0	140000.0	130000.0	120000-0	110000.0	1,00000,0	0"00006	800000	10000.0	INTEGRAL OF I MAKE RI/SIGMA)++2	WZ/WIF	J. 28	0.0	LOWER BOUND	0.0	0.0	0000000	4 66666666 01		0.0
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DATACH ROSENBROCK WRCONSTRAINED OPTIMIZER EXAMPLE DATACH ROSENBROCK WRCONSTRAINED OPTIMIZER EXAMPLE DATACH ROSE 2-1-COMILL - 1 FREF - 2 MOPT - 5 A 0.625 - 0.75 0.625 - 1.75 - 224.4 ADATA DATACH ROSE 2-5-5-5-5-5-5-5-5-5-5-5-5-5-5-5-5-5-5-5	nd M ip
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HTREA M MSTAG MSUCC	
0.0	
1 0.1000000-01 0.0	
P(1) (= 1 TO 2 1,00000000-02 0.0	
- 1H ROSENK H + - ElM) 0-100000000000000000000000000000	
400 0.0 4.0 0.0 0.0 0.0 2.60756250 00 2.60754250 00 46544 50 00	
0.0100000000 0.0	
IN ROSBAK DP -	
++++++++++++++++++++++++++++++++++++++	
IN ROSONK N = 1 E(N) = 0.3000000000-01	
12 LOWER SOUND UPPER BOLND OCCURET2) PERALTY 400 0.0 2.556.00000 00 X = 4.00000000-02 0.0	
1 H MSTAG MSUCC	
P.1.) 1 - 1.70 2 4.0	
IN ROSERS W 1 E(N) 0. 9000 0000000-01	
12 LOWER-BOUND UPPER BOUND 0CCURIIZ) PERMITY 400 0.0 0.0 0.0 0.0 0.0 2.40806250 00 2.40806250 00	

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IN ROSBAK DP 4.00000000-01 0.0 IN ROSBAK N = 1 E(N) = 0.2700000000 00 IN ROSBAK N = 1 E(N) = 0.2700000000 00 IN ROSBAK N = 1 E(N) = 0.2700000000 00 IN ROSBAK N = 1 E(N) = 0.2700000000 00 PENALTY 4.000000000		
0.0 W) - 0.270000000 00 WPPER BULSO 00 X - 4.00000000-01 0.0 SUCC U		
0.0 UPPER BUIND DCCUR(12) 0.0 00 X = 4.00000000-01 0.0 SUCC		
LOWER BOUND UPPER BOUND OCCUR(12) LOWER BOUND UPPER BOUND 2.02500000 00 F = 2.02500000 00 X = 4.06000000-01 0.0 N NSTAG NSUCC U 1 TO 2		
LOWER BOUND UPPER BOLNO DCCURITZ) 6.0 F = 2.02500000 00 X = 4.00000000-01 0.0 N NSTAG NSUCC U 2.025000000		
FTAG NSUCC		
-		
0.1000000000000000000000000000000000000		44
0.81000000 00 0.0		
P(11 1 = 1 TO 2		
IN KOSONA H - 1 ETH1 - 0-8100000000 00	Sales Sales	
12 LOMER SOUND UPPER BOUND OCCUR(12) PENALTY 400 0.0 0.0 0.0 1.42256250 00 1.42256250 00 1.42256250 00		
##### ################################		
IN ROSBEK DP -		
3.64000000 00 0.0		
IN ROSBAK N . 1 E(N) . 0.2430000000 01		
12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY -400 0.0 - 0.0 0.0 4.53600000 00 4.53500000 00		
MFRIA N NSTAG YSUCC		
0.0		
IN MUSBER DF = 0.10000000-01		

1.21000000 00 1.00000000 01		N MSTAG MSUCC 1.34765000000 1.100000000000000000000000000000	LINER BOUND UPFER BOUND OCCUREIZE PENALTY A 1.34765000 00 X = 1.21000000 00 1.30000000-01	. O . (M	1.21000000 00 1.30000000-01	X 09 - 0.0000000-01	Succ.	CONCR BOUND UPPER BOUND 0000UR(12) PENALTY 0 0.0 1.39726250 00 1.39726250 00 1.39726250 00 0 1.39726250 00 X = 1.21000000 00 4.00000000-02	## #050## # 5 -F4M} 0" 3000000000000000000000000000000	PELL 1 = 1 TO 2 1-21000000 00 - 4.00000000-02	0 - 3000 0000-01	0.0	N NSTAG NSUCC 1.4140500000		10000000000000000000000000000000000000
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IN ROSBAR DP -	0×41007000 00	
1.21000000 00	1.21000000 00	
*	00 0000000	
440 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0	UP PER ROLAND 1.54285988 00 1.54285908 00 00 X = 1.21000008 00 1.21000008 00 1.21000008 00	
MTRIA H NSTAG NSUCC	1 - 54 2-05-000-00 1 - 24 2-05-000-00	
C MATERY PROM BRAN 0.9494650593 0.8136727469	-0.3138727469 0.9494.650593	
C HATRIX FROM ROSBAK		
3.140-01 9.490-01		
MTRIA N NS TAG NSUCC	1, 2595625000	
IN ROSENT DF	0. 1898 930n 00	
F111 1 - 1 TO 2 00	7°-010-010-01	
"	00 00 000000000000000000000000000000000	
400 0.0 0.00 0.00 0.00 0.00 0.00 0.00 0	UPPER BOUND OCCURITY) PENALTY 0. 1.06785610 00 1.06785610 00 00 X 1.06462640 00 5.89893010 01	E
. MTRIA N MSTAG WSUCC 13 1 1 1 2 1.7544263609	1.0e785&19.29 0.5898930119	
IN RUSBUK DF - 17232790 01	00 3848 3448 00	
3,50770540 00	1.15¢72@ 3^	1
IN BOCKBE H = 1 (14)	1. 181 50000030 CI	

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	3D 00				14LTY 10-00					
P(1) 1 = 1 TO 2 P(1) 1 = 1 TO	FR BGUN	15 2 1 1.0327603223 P441 1 10 2 1 1.0327603223	IN ROSSAK DP	IN ROSORK N = 2 E(N) = 0.6000000000 00	12 LOWER BOUND UPFER BOUND OCCUR(12) PENALTY 8FEVE F # 1.33475690 00 X = 1.533282D 00 1.3494651D 00	NTRIA N NSTAG NSUCC 1.3347568950 P(1) 1 = 1 TO 2 1.5333281634 1.3494650593	C MATRIX FROM GRAM 0.8029677975 -0.5960224124 0.5960224124 0.8029677975	6-MATRIX-FROM ROSDAN 8-030-01 -5-960-01 5-960-01 8-030-01	NTAIA N NSTAG NSUCC 1+0327603223 PII) I = 1 TC 2 1-0-7797860237	1 0.24289780 00 0.18029680 00

	7
800MD OCCUR!	
F = 1.00072000 00 X = 1.96454960 00 9.500	
MTRIA N NSTAG NSUCC	
76-0-	1
1 0.72869330 00 0.54089030 00	100
P(1) 1 = 1 TO 2 2,469324280 00	
1N-ROSBAK-N =- 1 -FINI- = 0.96750000000 -00	
12 LUNER BOUND UPPER BOUND DCCURITZ) PENALTY 403 0.0 0.0 0.0 1.19675280 00 1.19675280 00 00 00 00 00 00 00 00 00 00 00 00 0	
101	
2,5932428465 1,5009731427	
IN MDSBPK DP = 0.59602240-01 - 0.50296780-01	
1.90494730 00 1.04087960 00	
IN ADSBRK N = 2 E(N) = 0.1000000000 00	
12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 100 0.0 0.0 0.0 0.0 X = 1.90494730 00 1.0403796D 00	
NTRIA N NSTAG NSUCC 1,0095445911 19 2 2 0 0 1,0095445911 P(1) 1 = 1 TO 2 1,0403795832	
1 0.29801120-01	
P(1) 1 = 1 TO 2	
400 0.0 1.00368730 00 X = 1.99435079 00 1.00368730 00 4.00 0.0	
A N NSTAC NSUCC	
- H	

FDSBRK DP = 0+14900560=01 0+20074190=01	
1.94964900 00 9.80157000-01	
IN ROSBRK N = 2 E(N) = 0.2500000000-01	
LOWER SOUND UPPER BOUND DCCUR(12) PENALTY 0.0	
1 2	
1 TO 2 -9496490699	
- 1950NK 07 - 02 -0.10037100-01	
1 TD 2 *971999-00 00 9-50045710-01	
IN ROSONK N = 2 E(N) = -0.1250000000-01	
LOWER-BOUND UPPER BOUND	
3000	
1.9719998504 0.9500457050 K DP = 0.50185400-02	1
P+1+ 1 - 1-70 2 1.96082440 00 9.65101350-01	
N = 2 E(N) = 0.6250000000-02	
LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 0.0	
N NSTAG NSUCC 1 CONCERNO	
12.44302 - 0.965	
IN ROSORK DP = 20-01 0-15055650-01	

HTRIA N NSTAG NSUC P(11) 1 = 1 TO 2 C-MATRIX FROM GRAH O. 7904845805 C-MATRIX FROM ROSBRK T. 500-01 -6-12D-01 6-12D-01 7.900-01 NTRIA N NSTAG NSUC IN ROSBRK DF 1 FtN) IN ROSBRK N 1 EtN) IN ROSBRK DF 2 P(11) 1 = 1 TO 2	FEV* F = 1.00108130 00 X = 1.94964900 00 9.8015700D-01 NTRIA N NSTAG NSUCC U 1.0010812682	6.0	505 -0.6124819409 60.7904845805		***************************************	N NSTAG NSUCC 1 3 - 0 - 1 - 0006950197 10 2 1608244302 - 0.9651013522	00 0.42637890-01	00 1*05773920-00	£fw) = -0=1512500000 00	LOWER BOUND UPPER BOUND OCCURTIZED PENALTY 0 0.0 1.00264120 00 X = 2.08038520 00 1.05773920 00	N NSTAG NSUCC 1	01 0*46316950-91	000 9.18782410-01	E(N) = -0.75625000000-01	R BOUND UPPER BOLND OCCUR(12) PENALTY 1.00421510 00 1.00421510 00 x = 1.90104400 00 9.18782410-01	N NSTAG MSUCC
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00 9.88260830-01 E(N) = 0.3781250000-01	0.0 830 00 X	6 NSUCC U	0.96	0-01 0.69478420-01			LOWER BOUND UPPER SOUND 05.0 1.0026412D 00 1.0026412D 00 5.0 1.002	NSUCE 1.00264121.62	30 1,0577392458	-02 0-24702640-02	00 9.90731090-01	EINI = 0.3125000000-02	D UPPER BOUND OCCUR(12) PENALTY 5-2D 00 X = 1,9888006D 00 9.9073109D-01	NSUCC 1.0000542322	-02 0.74107930-02	00
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2.0027411914 1.0035024180	
IN ROSBRK DP = 1 0.41821-710-01 0.38313980-01	
2.04456290 00 1.04181640 00	
IN ROSBAK N = 1 E(N) = 0.56718750000-01	
12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 180 0.0 1.00093640 00 X = 2.04456290 00 1.04181640 00	
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1.9890199298 1.0184798890

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# F = 1.00000000 00 X - 1.99988020 00 9.99946130-01 PRINT FCH	-4.685240-05	
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(MRKANONINI DANG DEMONSTRATOR DESCRIPTION

400 0.0 B.9526671D 00 X = -2.3607427D 00 -2.3607427D 00	12 LOWER BOUND UPPER BOUND OCCURITZ) PENALTY 400 D.0 *** 0.04972680 00 X = 2.36074270 00 -2.36074270 00	12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 403 0.0 2.28236930 01 X = -5.27851460 00 -2.36074270 00	12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 400 0.0 0.0 5.7004807D 00 5.7004807D 00 9FEV* F = 5.7004807D 00 X = -5.57029180-01 -2.3607427D 00	12 LOWER BOUND UPPER BOUND OCCURIIZ) PENALTY 400 0.0 5.72338250 00 4FEV* F = 5.72338255 00 X = 5.57029180-01 -2.36074270 00	12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 400 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0	12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY +00 0.0 0.0 0.0 5.52627280 00 5.52627280 00 •FEV* F = 5.52627280 00 X = -1.32625990-01 -2.36074270 00	12 LDWER BOUND UPPER BOUND OCCUR(12) PENALTY +00 0.0 5.53172560 00 X = 1.32625990-01 -2.36074270 00	12 LOWER BOUND UPPER BOUND DCCUR(12) PENALT, 400 0.0 0.0 0.0 5.56521630 00 = -2,91777190-01 -2,36074270 00	12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY -400-0.0 -0.0 -0.0 -0.0 5.51790010 00 *FEV* F = 5.51790010 00 X = -2.65251990-02 -2.36074270 00	12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 400 0.0 **FEV* F = 5.51899070 00 X = 2.65251990-02 -2.36074270 00	12 LOWER BOUND UPPER BOUND OCCUR(112) PENALTY 408 0.0 0.0 0.5.52032759 00 x = -7.95755;70-02 -2.36074270 00	12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 400-2.0	12 LONER BOUND UPPER BOUND OCCUR(12) PENALTY 400 0.0 1.87043900.00 **FEV* F = 1.87043900.00 X = 2.36074270.00 2.36074270.00	12 LIWER BOUNG (PPER BOUNG OCCURALLY) PENALTY

2241-5

TWO-VARIABLE FIBONACCI EXAMPLE

0.00	LOWER BOUND UPPER BOUND OCCUR(12)	LOWER BOUND UPPER BOUND 0.0 F = 8.63858620 00 X = 5.278	IZ LOWER BOUND UPPER BOUND UCCUR(IZ) PENALTY 400 0.0 FEV* F = 1.13648070 01 X = 2.36074270 00 5.27851460 00		LOWER BOUND LOWER
LOWER BOUND UPPER BOUND OCCUR(12) 1.13648070 01 X = 2.36074270 00 5.27851460 1.04ER BOUND UPPER BOUND OCCUR(12) 1.04ER BOUND UPPER BOUND OCCUR(12) 1.04ER BOUND UPPER BOUND OCCUR(12) 1.04ER BOUND UPPER BOUND OCCUR(12) 1.04ER BOUND UPPER BOUND OCCUR(12)	LOWER BOUND UPPER BOUND UCCUR(12) F = 1.13648070 01 X = 2.36074270 00 5.27851460 LOWER BOUND UPPER BOUND OCCUR(12) 0.0 8.63858620 00 X = 5.27851460 00 5.27851460	LOWER BOUND UPPER BOUND OCCUR(12) 0.0 1.13648070 01 X = 2.36074270 00 5.27851460			0.0 OCCUR(12) 0.0 0.0 3.83192280 01 3.83 F = 3.83192280 01 X = -2.36074270 00 5.2785146
LOWER BOUND UPPER BOUND DCCUR(12) 1.00 F = 3.83192280 01 X = -2.36074270 00 5.27851460 1.00 F = 1.13648070 01 X = 2.36074270 00 5.27851460 1.00 F = 1.13648070 01 X = 2.36074270 00 5.27851460 1.00 F = 8.63858620 00 X = 5.27851460 05.27851460 1.00 F = 8.63858620 00 X = 5.27851460 05.27851460 1.00 F = 8.63858620 00 X = 5.27851460 05.27851460	LOWER BOUND UPPER BOUND 0CCUR(12) 1.00	LOWER BOUND UPPER BOUND OCCUR(12) F = 3.83192280 01 X = -2.36074270 00 5.27851460 LOWER BOUND UPPER BOUND OCCUR(12) F = 1.13648070 01 X = 2.36074270 00 5.27851460	0.0 DCCUR(12) 0.0 3.83192280 01 X = -2.36074270 00 5.2785146		LOWER BOUND UPPER BOUND 0CCUR!
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LOWER BOUND UPPER BOUND CCCURILIZATION	LOWER BOUND UPPER BOUND COUNTING	LOWER BOUND UPPER BOUND OCCURENTIAL	LOWER BOWND UPPER BOWND CCURRITY		LOWER BOUND UPPER BOUND 0CCUR(12) 0.0 4.93125000 00 4.9 F = 4.93125000 00 X = 5.57029180-01 2.360742

LOWER BOUND	*FEV* F = 8.32257050 00 X = 4.58885940 00 5.27851460 00	
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7D OI	NALTY 60 00	TD 00	NALTY OD OO	30 00	NALTY OD OO	NALTY 6D 00	2D 00	NALTY 20 00	NALTY 60 00	NALTY 90 00	4B 00	NALTY 2n 01	NALTY
12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 400 0.0 0.0 1.37298770 01 1.37298770 01 01 01 01 01 01 01 01 01 01 01 01 01	12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 400 0.0 0.0 0.0 0.0 0.0 1.05262560 00 1.05262560 00 9FEV* F = 1.05262560 00 X = 2.36074270 CO 1.24668440 00	12 LOWER BOUND UPPER BOUND OCCUR:12) PENALTY 400 0.0 0.0 7.14937570 00 00 0.24663440 00 00 00 00 00 00 00 00 00 00 00 00	12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 400 0.0 0.0 0.0 0.0 2.60635500 00 2.60635500 00 WFEV* F = 2.60635500 00 X = 5.57029180-01 1.2466844E 00	12 LOWER BOUND UPPER BOUND OCCURITZ) PENALTY 400 0.0 0.12457430 00 2.12457430 00 #FEV* F = 2.12457430 00 x = 3.47480110 00 1.24668440 00	12 LOWER BOUND UPPER BOUND OCCURITZ) PENALTY 400 0.0 0.0 0.16650100 00 0.16650100 00 0.16650100 00 0.16650100 00 0.16650100 00 0.16650100 00 00 0.16650100 00 00 0.16650100 00 00 0.16650100 00 0.16650100 00 00 0.16650100 00 00 0.16650100 00 00 00 00 00 00 00 00 00 00 00 00	12 LOWER BOUND UPPER BOUND OCCURITZ) PENALTY 400 0.0 0.0 1.27805460 00 1.27805460 00 00 00 00 00 00 00 00 00 00 00 00 0	12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY NO. 0.0 0.0 1.02404520 00 1.02464520 00 1.24668440 00	12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 400 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0	12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 400 0.0 0.0 0.0 1.02613560 00 1.02613560 00 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.	12 1.34ER BOUND UPPER BOUND OCCUR(12) PENALTY 400 0.0 1.03130690 00 X = 2.04244030 00 1.254668440 00	12 LOWER BOUND UPPER BOUND 0CCUR(12) PENALTY 400 0.0 0.0 0.0 1.02434140 00 1.02434140 00 x = 2.14854110 00 1.24668440 00	12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 400 0.0 0.0 0.0 1.53613520 01 1.53613520 01 1.53613520 01 1.53613520 01 1.53613520 00	IZ LOWER BOUND UPPER BOUND DCCUR(IZ) PENALTY

DCCUR(12) PENALTY 1.18873449 00 1.67108750 00 B9390 00 1.67108750 00 B9390 00 1.67108750 00 CCCUR(12) PENALTY 1.19383540 00 1.67108750 00 CCCUR(12) PENALTY 1.19383540 00 1.19383540 00 CCCUR(12) PENALTY 1.18271870 00 1.18271870 00 B4350 00 1.67108750 00 CCCUR(12) PENALTY CCCUR(12) PENAL	2) PENALTY 60 1.18873440 90 67108750 00 67108750 00 67108750 00 67108750 00 67108750 00 67108750 00 70 1.18271870 00 67108750 00 71 PENALTY 60 1.18271870 00 71 PENALTY 61 1.28245370 00 72 PENALTY 63 1.28245370 00 73 PENALTY 74 PENALTY 75 PENALTY 76 1.28245370 00 76 1.28245370 00 77 PENALTY 78 PENALTY 79 PENALTY 70 7.76378240 00 81432360-01 71 PENALTY 72 PENALTY 73 PENALTY 74 PENALTY 75 PENALTY 76 PENALTY 77 PENALTY 78 PENALTY 78 PENALTY 79 PENALTY 70 7.76378240 00 81432360-01 81432360-01
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R(12) PENALTY 440 00 1.19383540 00 1.67108750 00 1.67108750 00 1.67108750 00 1.67108750 00 1.67108750 00 1.67108750 00 1.67108750 00 1.67108750 00 1.67108750 00 1.67108750 00 1.67108750 00 1.67108750 00 8(12) PENALTY 9.81432360-01 R(12) PENALTY 40 00 7.76378240 00 9.81432360-01	* BOUND DCCUR(12) PENALTY * 2.25464190 00 1.67108750 00 * 80UND 0CCUR(12) PENALTY * 2.41379310 00 1.67108750 00 * BOUND DCCUR(12) PENALTY * BOUND DCCUR(12) PENALTY * BOUND DCCUR(12) PENALTY * BOUND DCCUR(12) PENALTY * BOUND DCCUR(12) PENALTY * BOUND DCCUR(12) PENALTY * BOUND DCCUR(12) PENALTY * BOUND DCCUR(12) PENALTY * BOUND DCCUR(12) PENALTY * BOUND DCCUR(12) PENALTY * BOUND DCCUR(12) PENALTY * BOUND DCCUR(12) PENALTY * BOUND DCCUR(12) PENALTY * SAC74270 00 9.81432360-01 * BOUND DCCUR(12) PENALTY * SAC74270 00 9.81432360-01 * SAC74270 00 9.814323
R(12) PENALTY 00 0 1.1802100 00 1.67108750 00 1.67108750 00 1.67108750 00 1.67108750 00 1.67108750 00 1.67108750 00 9.81432360-01 R(12) PENALTY 50 0 7.76378240 00 9.81432360-01	JR(12) JR(12)
R(12) PENALTY 1.67106750 00 1.67106750 00 R(12) PENALTY 70 01 1.28245370 01 9.81432360-01 R(12) PENALTY 58 00 7.76378240 00 9.81432360-01	JR(12) J. 67108750 00 J. 67108750 00 JR(12) JR(12) PENALTY PENALTY PENALTY PENALTY PO. 81432360-01 PR(12) PENALTY PENALTY PENALTY PENALTY PO. 0 7.76378240 00 PENALTY
R(12) PENALTY 9.8143236D-01 R(12) PENALTY 60 0 1.08657360 00 9.8143236D-01 R(12) PENALTY 40 00 7.76378240 00 9.8143236D-01	PENALTY 9.6112) PENALTY 9.61432360-01 PENALTY PENALTY PENALTY PENALTY 9.81432360-01 7.76378240 00 9.81432360-01 PENALTY PENALTY 9.81432360-01 9.81432360-01 9.81432360-01 9.81432360-01
4 4	R(1Z) 9-814 9-814 40 00 9-814 9-814
14 <u>0</u> *	R(1Z) 9-814 8(1Z) 00 00 9-814
	R(12) PENALT 0D 00 2.28147400 0 9.8143236D-01

3.1785145 00 1.6710875D 00

aFEV* F = 6.34925750 00 X =

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#EV* F = 1.34440280 00 X = 1.24668440 00 9.81432360-01 12	12 LOWER BOUND UPPER BOUND 0CCUR(12) PENALTY 400 0.0 0.0 1.00724430 00 1	LOWER BOUND UPPER BOUND DCCUR(12) PENALTY D.O. 1.01586320 00 X = 1.83023870 00 9.81432360-01	LOWER BOUND UPPER BOUND OCCUR(12) PENALTY F = 1.0001381D 00 X = 1.98938990 00 9.81432360-01	12 LOWER BOUND UPPER BOUND DCCUR(12) PENALTY 400 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0	LOWER BOUND DER ROUND DCCUR(12) PENALTY 0.0 1.23235490 01 X = -2.36074270 00 8.22781170-01	LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 0.0 1.14915760 00 X = 2.36074270 00 8.22281170-01	LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 0.0 0.0 0.0 0.0 8.17464150 00 F = 8.17464150 00 8.22281170-01	LOWER BOUND UPPER BOLND 0CCUR(12) PENALTY 0.0 F = 2.12876070 00 X = 5.57029180-01 8.22221170-01	LOWER BOUND UPPER BOUND OCCUR(IZ) PENALTY 9.0 F = 2.57571380 00 X = 3.4748C110 00 8.22281170-01	LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 0.0 1.27400900 00 1.27400900 00 F = 1.27400900 00 X = 1.24668440 00 8.22281170-	LOWER BOUND UPPER BOUND OCCUR(12) PENAL; Y	LOWER BOUND UPPER BOUND OCCUR(12) PENALTY

FEV F = 1.06324980 00 X = 1.67108750 00 9.81432360-01

12) PENALIY	12)	12) PENALTY	12) PENALTY	12) PENALTY	12) PENALTY	12) PENALTY	12) PENALTY	112) PENALTY	11.25 PENALTY	(12) PENALTY	PENALTY	(12) PENALTY
1 00 1,01512450 00	90 1.01839610 00	5 00 1.61269718 08 -	01 1.31761190 01	100 1.06244060 00	-00 7.50746590 00	9-0-2-40087269 09	00 2-26736710 00	10 00 1.09399650 00	0.00 1.33852790 00	B 08 1.00421890 00	1 1.01150100 90	B 00 1.01695380 00
8,22281170-01	8.22281170-01	8.22281170-01	1.08753320 03	1.08753320 00	1.08753320 00	1-08753320 00	1-08753320 00	1.08753320 00	1.08753320 00	1.08753320 00	8753320 00	1.08753320 00
12 LOWER BOUND UPPER BOUNE DCCUR(17) 400 0.0 1.01512450 00 X = 1.83023870 00 8.2	12 LOWER BOUND UPPER BOUND 1.01839610 00 ** 1.98938990 00 8.2	12 LOWER BOUND UPPER BOLND 0.0269718 00 4.25 4718 00 4.269718 00 X = 1.86328918 00 8.2	12 LOWER BOUND UPPER BOUND OCCUR(12) 400 0.0 1.31761190 01 X = -2,36074270 00 1.0	0 0 0 X = 2.360742 TO 00	12 LOWER BOUND UPPER BOUND OCCUR(12) 400 0.0 7.50746590 00 x = 5.27851460 00 1.0	12 LOWER BOUND UPPER BOUND 2.40887269 00 00 00 00 00 00 00 00 00 00 00 00 00	12 LOWER BOUND UPPER BOUND 2.26736710 00 eFEV* F = 2.26736710 00 x = 3.47480110 00 1.0	UPPER BOUND 1.0939965 50 00 X = 1.67108750 00	12 LOWER BOUND UPPER BOUND OCCUR(125) 400 0.0 4FEV* F = 1.33852790 00 x = 2.78514590 00 1.01	12 LOWER BOUND UPPER BOUND OCCUR(12) 400 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0	12 LOWER BOUND UPPER BOUND OCCUR(IZ) $400 - 0.0$ 0.0	12 LOWER BOUND UPPER BOUND 0CCUR(12) 4.00 0.0 1.01695380 00 x = 2.20159150 00 1.0

Å.I.	90	17 00	17♥ 00	11Y 00	7 Y	1 A A A A A A A A A A A A A A A A A A A	1TY 000	11¥ 00	1.K	000	± ₹ 00	10 10	1.4 00	>
LOWER BOUND	• F = 1.10391700 00 X = 2.36074270 00 9.283	12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 400 0.0 7.89721750 00 X = 5.27851460 00 9.28381960-01	12 LOWER BOUND UPPER BOSNO OCCUR(12) PENALTY +000 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.	12 LOWER BOUND UPPER BOIND DCCUR(IZ) PENALTY 400 0.0 2.44182130 00 X = 3.47480110 00 9.2838196D-01	12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 400 0.0 0.0 1.05315330 00 PFEV* F = 1.05315330 00 X = 1.67108750 00 9.28381960-01	12 LOWER BOUND UPPER BOUND DCCUR(IZ) PENALTY +00 0.0 1.3174203D 00 X = 1.2466844D 00 9.2838196D-01	12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 400 0.0 1.00231920 00 1.00231920 00 +FEV* F = 1.00231920 00 X = 1.93633950 00 9.28381960-01	12 LOWER BUIND UPPER BOUND 0CCUR(IZ) PENALTY 400 0.0 0.0 1.01403390 00 1	12 LOWER BOUND UPPER BOUND OCCUR(12) FENALTY 400 0.0 1.0120990D 00 X = 1.8302387D 00 9.2838196D-01	12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 400 0.0 1.0027062D 00 1.0027062D 00 1.0027062D 00 1.9893899D 00 9,2838196D-01	12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 400 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0	12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 400 0.0 0.0 0.0 0.0 0.0 1.29985690 01 1.29985690 01 4.29985690 01 4.29985690 00 0.03448280 00	12 LOWER BOUND UPPER BOUND 0CCUR(12) PENALTY 400 0.0 1.07274820 00 1.07274820 00 *FEV* F = 1.07274820 00 X = 2.36074270 00 1.03448280 00	LOWER BOWND UPPER BOUND OCCUR

FEV F = 1.00555570 00 X = 1.98938990 00 1.08753320 00

5,57029180-01 1,03448280 00	ND OCCUR(12) PENALTY 2,32200060 00 2,32200060 00 4,748011D 00 1,0344828D 00	ND 0CCUR(12) PENALTY 1.07686420 00 1.07686420 00 67108750 00 1.03446280 00	ND 0CCUR(12) PENALTY 1.36572158 90 1.36572150 00 2.78514590 00 1.03448280 00	2.09549070 00 1.039448280 00	ND DCCUR(12) PENALTY -1.0049225B 00 -1.0049225B 00 -9363395D 00 1.0344828D 00	ND 0CCUR(IZ) PENALTY - 1.02092918 00 1.02092910 00	NAID DCCURITZ) PENALTY 1.00077130 00 1.00077130 00 2.04244030 00 1.03448280 00	ND DCCUR(12) PENALTY 1.00108798-00-1.00108790-00 98938990-00-1.03448280-00	JUND DCCURFIZI PENALTY 1.98938990 00 9.8143236D-01	
F = 2.33941440 00 X = 5.5	LOWER BOUND UPPER BOUND 0.0CU	12 LOWER FOUND UPPER BOUND OCCUI 406 0.0 1.07686420 00 X = 1.67108750 00	LOWER SOUND UPPER BOWND 0CCU 0.0 1.36572150 00 x = 2,78514590 00	LOWER BOUND UPPER BOUND F = 1.00397260 00 X = 2.0	1.0MER BOUND UPPER BOUND 0CCU	LOWER BOUND UPPER BOUND 0CCUI	LOWER BOUND UPPER BOUND F = 1.00077130 00 X = 2.0	LOWER BOUND UPPER BOUND DCCUU 0.0 1.00108790 00 X = 1.98938990 00	LONER BOUND UPPER BOUND -0.0 -0.0 -0.0 -0.0 -0.0 -0.0 -0.0 -0.0 -0.0 -0.0 -0.0 -0.0 -0.0 -0.0 -0.0	1 ITERM - 0

FEV F = 7.63385520 00 X = 5.2735146, 00 1.03448280 00

104 204 3 10411 1 105 2 100 1 5 4 0 4 5 5 - 4 7 5 0 4 5 5 2 4 1 7 5 0 4 7 5 4 4 1 7 5 6 4 1 7 5	DATA CASE 7.0 DATA HOSENBROCK DESIGN VARIABLE OPTIMIZER, CONSTRAINED	2241-7 *DATA 2241-7 *DATA	*DATA
YLVO.		IN 2 2241 IN 2 2241 IDC 2 400 2241	
	DATA+1		*DATA

00 00000000	1000000 n1 2.50000000 00 2.00000000 00 0.0
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2.21000000 00 2.01000000 00	
IN ROSBAK N . Z E(N) * 0.1000000000-01	
12 LOWER BOUND UPPER BOUND OCCUR 12) PENALTY 2 1.80000000 01 2.50000000 00 1.50605000 00 1.64876600-01 1.10000000 00 1.50605000 00 1.64876600-01	
##### # #\$746 #3466 0.1648766025 #41-1-170-2 2.2100830630 2.0100000000	
ROSBAX DP = -0.50000000-02	
2.2100000 00 1.99500000 00	
IN ROSBRK N = 2 E(N) = -0.5000000000-02	
12 LOWER BOUND UPPER BOUND OCCURILLY PENALTY 2 -1.00000000 01 2.50000000 00 1.99500000 00 0.0 1.10000000 00 1.48961560 00 1.51800340-01 400 0.0 1.51800340-01 X = 2.21000000 00 1.99500000 00	
##### # WSTAG WSUCC 0.1518003352 ### 1 - 1 10 2 0.1518003352 ### 2.2100000000 1.9950.000000	
IN RUSBAK DP =	
2.21000000 00 1.98000000 00	
IN RUSBRK N = 2 F(N) :: -0.1500000000-01	
12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 2 -1.80888606 04 2.50880090 00 1.98000000 00 0.0 400 0.0 400 0.0 4.39474240-01 X = 2.21060009 00 1.98000000 00	
##### N WSTAG W30CC 0.1394742389 10 2 0 2 0.1394742389 #### 1 = 1 19 2 2.2100000000 1.980C000000	
IN ROSSINK OF = - O. TOCOGO OF	

12 LOWER BOUND UPPER BOUND OCCUR(12: PENALTY 400 0.0 0.0 1.06726760-01 1.10000000 00 1.42669060 00 1.06726760-01 PERALTY PERALTY F x 1.06726760-01 X = 2.21000000 00 1.93500000 00 PENALTY 1.94602500-02 2.5000000 00 1.3950000 00 0.0 1.10000000 00 1.06285560 00 0.0 X = 2.2100000 00 1.3950000 00 1.39500000 00 1.39500000 00 LOWER BOUND UPPER BOUND OCCURITZ)
00000000 01 1.25550000 00 1.39500000 00 1.94
0 1.10000000 00 1.06286560 00 0.0
1.194602500=02 x = 2.21000000 00 1.3950000 1.39500000 00 1.06286560 00 2,21000000 00 0. 1067267645 0.0406274414 -0.13500000000000000 -0.40500000000 00 0. 01 94e 02500 0.1350000 00 1.9350000000 1.80000000 00 1.8000000000 1.39500000 00 1.3950000000 2.50000000 00 1.10000000 00 2 -1.00000000 01 1.25550000 00 7 IN ROSBRK N = 2 E(N) = IN ROSBRK N = 2 EIN) = O ITERM = 2.2100000000 2.21000000 00 2.2100000000 2.21000000 00 12 LOWER BOUND 2 -1-00000000 01 400 0.0 2.2100000000 -1.0000000001 PULL 1 - 1 TO 2 ** F. 0.0 IN ROSBAK DP -IN ROSBRK OF 1 KRED = 90 11-209

-0.4500000000n-01

2 E(N) =

IN ROSBRK N =

20 000000000000000000000000000000000000		
IN ROSSAX N = 1 E(N) = 0.1000000000-01		
155000 00 1,3950000 1000000 00 1,0625060 1000000 00 1,062506	112) PENALTY 0 00 1.9460250B-32 0 00 0.0 1.39500000 00	
##### ################################		
### ### ### ##########################		
BOUND UPPER BOUND 0CCUR 100 00 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	112) PENALTY 0 00 1.94602500-02 0 00 0.0 1.39500000 00	
MTRIA 1 NSTAG WSUCE 0.01946 02500 0111 1 1 10 2 0.250000000 0.3950 0000000		
" - P+11 1 - 1 10 2 1.39500000 00		
12 LONER BOUND UPPER BOUND OCCURITZ) 2 -1.0000000-01 -1.25550000 00 1.39500000 00 400 0.0 1.06904060 00 39FEV* F = 1.94602509-02 X = 2.34000000 00 1.39	1121 PENALTY 00 1.94602500-02 00 0.0 1.39500000 00	
PILL I = 1 TO 2 0 0.0194602500 2.3400000000 1.3950000000		

- 1

DENALTY 0.02802500-02 112) PENALTY C 00 1.94602500-02 D 00 2.43696490-03 1.39500000 00 LOWER BOWN UPPER BOUND UCCUR(12) PENALTY
-1-00000000-01-1-25550000-00-1-40500000-00-2-23502500-02
0.0 1.0714906D 00 0.0
1-10000000 00 1.0714906D 00 0.0 1-8090250D-02 12 LOWER BOUND UPPER BOUND DCCUR(12) PER 400 0.0 1.25550000 00 1.39000000 00 1.8090256 400 0.0 1.10000000 00 1.06786250 00 0.0 9EEV F x 1.80902500-02 X = 2.34000000 00 1.39000000 00 12 LOWER BOUND UPPER BOUND OCCURITY)
-2 -1-00000000 01 1.25550000 00 1.3950000 00
400 0.0
1.10000000 00 1.14936560 00
000000 0.0
1.10000000 00 1.14936560 00
000000 0.0
1.10000000 00 1.14936560 00
0000000 00 1.39 0.0218972149 0. 0223502500 0-1000000000000000 -0.50000000000-02 0. 01 809 02 5 00 -0.150000000000-01 1.3 50000000 0-100000001-0 1.40500000 00 1.4050000000 1.39000000 00 1.390000000 1.37500000 00 12 LOWER BOUND UPPER BOLNO 2 -1.00000000 01 1.25550000 00 400 0.0 1.10000000 00 IN ROSBRK N = 2 E(N) = TH ROSBRK N = 2 FINS = IN ROSBRK N - 2 E(N) -2.6100000000 2.34000000 00 2.3400000 00 2.3400000000 2.340000000 2. 340000 00 IN ROSBRK DP -IN ROSBEK DP -DILL 1 - 1 TO IN ROSBAK OP 1

11

0.27000000000 00

IN ROSBRK N = 1 EIN) =

|--|

1 170 2 0 1 1 2 0 1 1 2 0 1 1 2 0 1 1 2 0 1 1 2 0 1 1 2 0 1 1 2 0 1 1 2 0 1 2 2 2 3 8 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	12 LINER BOUND UPFER B 400 0.0 1.0755000 600 0.0 1.1000000	1.07550000 00 1.1950000 00 1.42802500-02 1.10000000 00 1.04914060 00 0.0 00-02 x 2.3500000 00 1.1950000 00	
2.350000000 1.19500000 0 588K DP	2 1 1 0 2 0 1 1 1 1 1 1 1 1 1 1 1 1 1 1	0.0142802500	
2.380000000 00 1.19500000 00 1.00000000 00 1.19500000 00 1.00000000 01 1.19500000 00 1.1950000 1.10000000 01 1.19500000 00 1.1950000 1.10000000 01 1.19500000 00 1.19500000 1.10000000 01 1.19500000 00 1.19500000 00 1.10000000 01 1.19500000 00 1.19500000 0	2. 3500000000 RUSBAK DP -		
1	2.38000000 00	1.19500000 00	
LOWER BOUND UPPER BOUND OCCURRITZ) 1.00000000 00 1.05844060 00 0.0 1.100000000 00 1.05844060 00 0.0 2.380000000 02 X = 2.38000000 00 1.195000 2.380000000 01 1.195000000 00 1.19500000 00 2.470000000 01 1.0750000 00 1.19500000 00 1.40500000 00 1.40500000 00 1.40500000 00 1.19500000 00 1.19500000 00 1.19500000 00 1.19500000 00 1.19500000 00 1.19500000 00 1.19500000 00 1.19500000 00 1.19500000 00 1.19500000 00 1.1950000000 00 1.19500000	1 E(N) -	0.30000000000-01	
### ##################################	1.00000000 01 0.0 F x 1.4280250	0000 00 1.1950000 00 1.4	
## DP = 0.0 2.470000000 01 1.19500000 00 BRK N = 1 E(N) = 0.900000000 01 1.00000000 01 1.07550000 00 1.10000000 01 1.9500000 00 F 1.42802500-02 X = 2.47000000 00 1.170 2 0.0142802500 2.470000000 00 1.195000000 00 1.195000000 00 3.74000000 00 1.195000000 00 1.195000000 00	3 1 0 2 1 - 1 10 2 2 - 3 8000 00000	0.0142802500	
2.4 TOGGO O		0.0	Ī
LOWER BOUND UPPER BOUND DCCUR(12) 1.00002000 01 1.07550000 00 1.10000000 00 1.1950000 00 5 1.42802500-02 X = 2.47000000 00 1.195 1 4 5 6 7 5 6 7 6 7 6 7 6 7 6 7 6 7 6 7 6 7	00	.19500000 00	
1.00000000 01 1.07550000 00 1.19500000 00 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.	ROSBRK N = 1 E(N) =	D. 9000000000000000000000000000000000000	
2.470000000 1.19	10MER 80UND -1-00000000 01 0-0 F - 1-4280250	0.00 1.19500000 00 0.00 1.19500000 00 0.00 1.09309060 00 2.4 7000000 00 1.195	
2.740000 00 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	1 - 1 TQ 2 2.4700000000	0.0142802500	
2.74000000 00 1	IN RUSORY DP -	0.0	
ROSBRK W . 1 FEW) =	2.140000 00	.19500000 00	
The second secon	ROSBRK W . 1 FERS .	0.2 700 0000000 00	

11-21

+	
1.195000000	
IN ROSSHK DP - 0.10000000-01	
P(1) 1 = 1 10 2 2 4.2050000 00 1.20500000 00	
IN ROSBAR N = 2 - EIN = - 0.100000000-01	
12 1.00000000 01 1.07550000 00 1.20500000 00 1.67702500-02 400 0.0 1.67702500-02 x = 2.47000000 00 1.20500000 00	
NTRIA N IISTAG NSUCC	
1.24	
1 ************************************	
P(1) 1 = 1 TO 2 2 4.1900000 00 1.19000000 00	
-IN ROSBKK N = -2 -E(N) = -0.5000000000-02	
11 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 2 -1.00000000 01 1.07550000 00 1.19000000 00 1.31102500-02 400 0.0 1.10000000 00 1.09365000 00 0.0 *FEV* F = 1.31102500-02 x = 2.47000000 00 1.19000000 00	
NTRIA N NSTAG NSUCC	
P(1) 1 - 1 TC 2 2.4700000000 1.1900000000	
14 ADSBAK DP = -0.1 5000000-01	
P(1) 1 = 1 T0 2 2.47798088 00 - 1.17508000 00	
IN RUSBAK N = - 2 - EIN) = -0.15000000000-01	
12 LOWER SOUND UPPER BOUND OCCURITY PENALTY 2 -1.00000000 01 1.07550000 00 1.17500000 00 9.90025000-03 400 0.0 1.10000000 00 1.09551560 00 0.0 eFEV* F = 9.90025000-03 x = 2.47000000 00 1.17500000 00	
NTRIA N NSTAG NSUCC	

P+11 1 - 1 10 2 00	1.13000000 00
IN ROSBRK N = 2 EIN) =	-0.4500000000-01
12 100ER BOUND UPPER B 12 10000000 01 1.0755000 100 0.0 1.1000000	UPPER BOUND OCCURAIL2) PENALTY 1.07\$\$0000 00 1.13000000 00 2.9702\$0000-03 1.100000000 00 1.10280000 00 7.84000000-06 05-03 X = 2.47000000 00 1.13000000 00
NTAIR N NS TAG N5UGG	0.0029780900
IN ROSERK OF	-0.1350000 00
+111 1 - 1 TO 2 2.47000300 00	9.95000000-01
IN ROSBRK N = 2 E(N) =	-0.1350000000 00
800MD 000 01	UPPER BOUND DCCURILZ) PENALTY 1.07550000 00 1.13984060 00 1.58727540-03 40-03 X = 2.47000000 00 9.95000000-01
MINIS NSTAB NSUCC	0.0015872754
IN RUSBRK DP =	-0, 40500000 00
2.4700000 00	5.90000000-01 -0.40500000000 00
2 LOWER BDUND 2 -1,00000000-01-1,00000000-01-1,000000000	5. 9000 1.3870
MTRIE N NSTAR NSUCE PILL I TO 2	0.0827425225
C MATRIX FROM GRAM 0.5449833506 -0.8384436163	-0.8384436163 -0.5449883506
C MATRIX FROM ROSBRK	

SBRK DP = 0.0015872754 1	5-450-01 -8-380-01
	1 NSTAE NSUEE 0.0015872754
	2.470000000 0.995000000
IN ADSSECT N	RDSBAX DP = 0.354248-01
It	1 1 10 2 2.50542420 00 9.40501160-01
It	N = 1 E(N) = 0.6500000000-01
IN ROSERK N = 1 E(N) = -0.325-0000-01 IN ROSERK N = 1 E(N) = -0.325-0000000-01 IN ROSERK N = 1 E(N) = -0.325-00000000-01 IN ROSERK N = 1 E(N) = -0.325-0000000-01 IN ROSERK N = 1 E(N) = -0.925-000000-01 IN ROSERK N = 1 E(N) = -0.925-000000-01 IN ROSERK N = 1 E(N) = -0.925-0000000-01 IN ROSERK N = 1 E	LOWER BOUND UPPER BOUND OCCUR(II2) PENALTY -1.00000000 01 1.0000000 00 1.12425250 00 7.12761870-03
N ROSSMR P 1 1 1 1 1 1 1 1 1	N WSTAG WSUCC 0.0071276187
IN RDSBAK DF	2.5054242428 0.94
	ROSBRK DP *
IN ROSBRK N = 1 E(N) = -0.3250000000-01 IZ LUMER BUUND UPPER BURND OCCURILY -2.1-00000000	11-10-2 2-45228790 00 1-02224940 00
IZ LOWER BOUND UPPER BOUND OCCURRIZED PENALTY	RDSBKK N = 1 E(N) = -0.3250000000-01
N N N N N N N N N N	LOWER BOUND UPPER BOUND DCCUR(II2) PENALTY -1.00000000 01 1.02224948 -00 0.0 0.0 1.10000000 00 1.12061470 00 4.24967700-04 0.0 1.24967700-04 X = 2.45228700 00 1.02224940 00
IN ROSBRK DP =	- WTRIA N WSTAG WSUCC 0.0004249677 13 1 1 0.0004249677 - P411 13 1 0.0004249677 2.4522878786 1.0222494175
IN ROSBRK N = 1 E(N) = -0.9750000000-01 12 LOWER BOUND UPPER BOUND OCCUR(12) 2 -1.00000000 01 1.07520200 00 0.00 400 0.0	IN ROSBRK DP = -0.5313636D-01
IN ROSBRK N = 1 E(N) = -0.9750000000-01 12 LOWER BOUND UPPER BOUND OCCUR(12) 2 -1.00000000 01 1.07550000 0 1.10399770 00 0.121 400 0.0	2.39915150 00 1.10399770
12 LOWER BOUND UPPER BOUND OCCUR(12) 2 -1.00000000 01 1.00399770 00 0.0 400 0.0 1.100000000 00 1.07520280 00 0.0	N = 1 E(N) =
	12 LOWER BOUND UPPER BOUND OCCUR(12) 2 -1.00000000 01 1.007550000 00 1.10399770 00 0.0 400 0.0 1.0177700 04 1.1040077

## # # ## ## ## ## ## ## ## ## ## ## ##	2.3991515144 1.1039976701	
IN ADSSERT N = 2 E(H) = -0.1000000000 00 IN ADSSERT N = 2 E(H) = -0.1000000000 00 IN ADSSERT N = 2 E(H) = -0.1000000000 00 IN ADSSERT N = 2 E(H) = -0.1000000000 00 IN ADSSERT N = 2 E(H) = -0.100000000 00 IN ADSSERT N = 2 E(H) = -0.000000000 IN ADSSERT N = 2 E(H) = -0.000000000 IN ADSSERT N = 2 E(H) = -0.3000000000 IN ADSSERT N = 2 E(H) = -0.300000000	ROSBRK DP = 0.63844340-01	
IN ROSBRK N = 2 E(N) = -0.1000000000 00 12	8	
12 LOWER BOUND UPPER BOUND DCCURILE) 1400 0.0 0.0 1.07574830 00 1.07574830 00 1.07574830 00 1.07574830 00 1.07574830 00 2.07574830 00 2.07574830 00 2.07574830 00 2.07574830 00 2.07574830 00 2.075748200 00 2.07574820	= 2 E(N) = -0.1000000000	
WERLA	LOWER SOUND UPPER BOUND DCCUR(12) -1.00000200-01-1.07550000-00-1.07674830-00-1. 0.0 1.100000000 00 1.15246960 00 2.	
IN ROSBAK DP = 2.4103457 D0 9.95000000-01 IN ROSBAK N = 2 E(N) = 0.50000000-01 IZ LOWER BOUND UPPER BOUND 0.110460450 00 ENTARA N NSTAC NSUCC 0.241034570 00 8.9 NYALA N NSTAC NSUCC 0.0000463012 P411 1 = 1 T0 2 1 1 1 0.0000000 0 IN ROSBAK DP = 0.413251750-01 IN ROSBAK DP = 0.13251750-01 IN ROSBAK N = 2 E(N) = 0.15020000000 00 IZ LOWER BOUND UPPER BOUND 0.12351750-01 IN ROSBAK N = 2 E(N) = 0.15020000000 00 IZ LOWER BOUND UPPER BOUND 0.13251750-01 P413 1	15 2 0.00275462(15 1 10 2 1 0 0.00275462(2.53613.22402 1.0767482526	
IN ROSBRK N = 2 E(N) = 0.50000000-01 IZ LOWER BOUND UPPER BOUND OCCUR(12) 400 0-0 **EVA C		
IN ROSBRK N = 2 E(N) = 0.5000000000-01 12 LOWER BOUND UPPER BOUND OCCUR(12) 400 0.0 1.100000000 00 1.10000000 00 1.10000000-01 1.100000000 00 1.10000000 00 1.10000000 00 IN ROSBRK DP = 0.1257665D 00 -0.8174825D-01 IN ROSBRK N = 2 E(N) = 0.15020000000 00 IL LOWER BOUND UPPER BOUND OCCUR(12) 400 0.0 12 LOWER BOUND UPPER BOUND OCCUR(12) 400 0.0 12 LOWER BOUND UPPER BOUND OCCUR(12) 400 0.0 13 LOWER BOUND UPPER BOUND OCCUR(12) 400 0.0 14 LOWER BOUND UPPER BOUND OCCUR(12) 400 0.0 15 LOWER BOUND UPPER BOUND OCCUR(12) 400 0.0 16 LOWER BOUND UPPER BOUND OCCUR(12) 400 0.0 17 LOWER BOUND UPPER BOUND OCCUR(12) 400 0.0 400 0.0 1 LOWER BOUND UPPER BOUND OCCUR(12) 400 0.0 400 0	2,410365 TO 00	
IZ LOWER BOUND UPPER BOUND 0CCUR(12)	2 E(N) =	
NTAIR N	L.0755000 00 1.10680450 00 1.00-05 00 00 00 00 00 00 00 00 00 00 00 00 0	
IN ROSBRK DP = -0.1257665D 00 -0.01748250-01 2.2845992D 00 9.13251750-01 IN ROSBRK N = 2 E(N) = 0.1502000000 00 IL LINER BUNNO UPPER BOUND OCCUR(12) 400 0.0 -2 -1.00000000 01 1.10000000 00 1.0738426D 00 0.0 **FW4 F 0.0 -1.10000000 01 1.07550000 00 9.13251750-01 0.0 400 0.0 **FEV4 F = 0.0 **F	14 2 1 10 2 1 1 0.0000463012 2.41036.56978 0.9950000000	
E(N) = 0.13251750-01 E(N) = 0.15020000000 00 1 1.007550000 00 1.07384260 00 0.0 1.10000000 00 1.07384260 00 0.0 1.107550000 00 9.13251750-01 0.0 1.07550000 00 9.13251750-01 0.0 1.07550000 00 1.07384260 00 0.0 X = 2.28459920 00 9.13251750-		
LOWER BOUND UPPER BOUND OCCUR(12) 1.00000000 01 1.0755000 06 9.13251750-01 0.0 1.10000000 00 1.07384260 00 0.0 1.10000000 00 1.07384260 00 0.03251750 1.00000000 01 1.0755000 00 9.13251750-01 0.0 1.00000000 01 1.0755000 00 9.13251750-01 0.0 1.00000000 01 1.0755000 00 1.07384260 00 0.0 1.100000000 00 1.07384260 00 0.0	8	
134ER BOUND UPPER BOUND DCCUR(12) 0.0 1.10000000 00 1.07354260 00 0.0 1.10000000 00 1.07384260 00 0.0 1.00000000 01 1.07550000 00 9.13251750-01 0.0 0.0 1.00000000 01 1.07550000 00 1.07384260 00 0.0 F = 0.0 X = 2.28459920 00 9.13251750-01	2 E(N) = 0.1502000000	
1.00000000 01 1.07550000 00 9.13251750-01 0.0 0.0 0.0 F = 0.0 X = 2.28459920 00 9.13251750-	-1.90999900 0PPER BOUND OCCUR(12) -1.90999900 01 -1.97559009 00 -9.13251750-01 0.0 0.0 1.10900000 00 1.07384260 00 0.0	
	1.00000000 01 1.07550000 00 9.13251750-01 0.0 0.0 0.0 F = 0.0 X = 2.28459920 00 9.13251750-01	
	2 -1-00000000 01 -8-21926570-01 9-13251750-01 6-34028750-03	

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#444 1 - 1 10 - 2 - 2 - 2 - 2 - 2 - 2 - 2 - 2 - 2 -	
IN ROSBAK DP =	
2.29459920 00 9.13251750-01	
IN ROSBIK N = 1 E(N) = 0.100000000-01	
IZ LOWER SOUND UPPER BOUND OCCURILI2) PENALTY -2 -1.0000000 01 -0.21924578-01 9.13251750-01 0.0 400 0.0 400 0.0 400 0.0 400 0.0 400 0.0	FMALTY 750-03
MTRIA NSTAG WSUCC 0.0083402875	
Pitt i = 1 10 2 2,2945991553 0,9132517476	
IN ROSBRK DP =	
2.3245 99.20 00 9.13251 750-01	
N) = 0.30000000000000000	
12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY PENALTY 12 LOWER BOUND UPPER BOUND OCCUR(12) 8.34028750-03-2-1.00000000-01 8.219265750-03-1.10000000 00 1.09167500 00 0.0 400 0.0 400 0.0 9.13251750-01	FHAL TY 1750-03-
IN ROSBRK DP = 0.0-01-01-01-01-01-01-01-01-01-01-01-01-01	
P+11 1 = 1 10 2 2 9.13251 750-01	
ND UPPER BOUND ACCURITZ) ND UPPER BOUND ACCURITZ) 01 8.21926570-01 9.13751750-01 8. 1.10000000 00 1.13911040 00 1. 1.10000000 02 K = 2.41459920 00 9.13251	PENALTY 34028750-03 52962250-03 750-01

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2.4145991553	0.9132517474	
IN ROSERK DF .	10-0000000-01	4
2.32459920 00	9.23251750-01	
IN ROSORK N = 2 E(N) =	0.1000000000000000000000000000000000000	
12 LOWER BOUND UPP 400 0.0 1.100	UPPER BOUND OCCUR(172) PENALTY 1.10000000 00 1.08821850-01 1.02467940-02 1.10000000 00 1.08821860 00 0.0	
PLIT I - 1 10 2 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	7474	
IN ROSARK DP	-0.50000000-02	
2,32459920 00	9.08251750-01	
(N ROSDAK N . 2 E(N) .	-0* 5000 0000000-02	
12 10MER BOUND UPPER 400 0.0 1.1000	UPPER BOUND OCCUR(12) PENALTY 6.2192457D-01 9.08251750-01 7.4520358D-03 1.10000000 00 1.09345000 00 0.0 0.03 X 2.33450020 00 /,09351750-01	
P(1) 1 - 1 TD 2 15.3245991553	0.0074520358	
IN ROSBAK DP -	-0-19000000-01	
2.3245920 00	6. 93251 750-01	
IN ROSBER N = 2 EEN? .	-0.1500000000-01	
400 0.0 1.1000 400 0.0 1.1000	UPPER BOUND OCCUR(12) PENALTY 6.2192657D-01 8.93251750-01 5.08728060-03 1.10900000 00 1.09896270 00 0.0 60-03 X 2.33459920 00 6.93361740-01	
PHI 1 - 1 TO 2	0.0050872806	
IN ROSBRK DP -		

-0-45000000-01

IN ROSBAR OF THE CONTRACTOR OF	
0.52407140-02 0.65165830-02	
1	
TH ROSENK H 1 - E(N) 0.1 808 8800 00 - 61	
2 -1.00000000 01 8.21926570-01 8.56768330-01 1.21394810-03 400 0.0 1.133215460-03 x = 2.31935820 00 8.56788330-01	
WERTA H MS TAG MSUCC	
0.8	
I C.26204.870-02 -0.42582920-02	
P(1) 1 - 1 TO 2 - 4.43993440-01	
2 -1.000C0000 01 8.21926570-01 8.43993460-01 4.86947330-04 400 0.0 1.10000000 00 1.12041890 00 4.16895310-04 #FEV* F # 9.03842640-04 X = 2.32721940 00 8.43993460-01	
NTRIA N NSTAG MSUCC	
0.0	
IN ABSON. OF0.12774879-01	
P(1) 1 = 1 TO 2	
IN ROSSNE N = 1 - E(N) = - 0.1500000000-01	
2 -1.00000000 01 8.21926570-01 8.31216580-01 8.63414200-05 400 0.0 0.0 1.10000000 00 1.13039570 00 9.23894700-04 9.23894700-04	
MTRIA H MSTAG MSUCC	

2.35489950 00 8.61026620-01	
IN ROSBAR N = 2 EIN)0.3250000000-01	
12 LOWER BOUND UPPER BOUND OCCURILE) PENALTY 2 1.0000000 01 0.21926578-01 0.10206529-01 1.52001399-03 400 0.0 1.10000000 00 1.12778270 00 7.71878000-04 9FEVA F 2.30040130-03 X = 2.35489650 00 0.61026620-01	
IN ROSBIN DP -	
2.31338020 00 8.35476870-01	
IN ROSBAR N = 2 E(N) = 0.1625000000-01	
12 LOWER BOUND UPPER BOUND OCCURILIZI PENALTY 2 -1.00000000 01 0.21926578-01 0.35476878-01 1.83610630-04 -400 0.0 1.110000000 00 1.11696560 00 2.87831450-04	
300	
Z.3133801948 0.8354768727 IN ROSBAK DP =	
++++ 1 + 1 +0 + 2 + 1 + 1 + 1 + 1 + 1 + 1 + 1 + 1 + 1	
12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 2 -1.00000000 01 8.21926570-01 8.09927120-01 0.0 1.10000000 00 1.10752800 00 5.66711810-05 460 0.0 460 4.	
HIRITA N MSTAG MSUCC 0.0000566712 P411 1 = 1 TD 2 2.2718618519 0.699271232	
IN ROSSIN, OP	

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-1.0000000 01 8.2192651	10-01 7.33277870-01 0.0 10-01 7.33277870-01 0.0
2 -1.0000000 01 8.21926570-01 7.332 406 0.0 1.0000000 00 1.087 8FEV F = 0.0 X = 2.14730680	7.33277670-01 0.0 1.08749250 00 0.0 730680 00 7.33277870-01
KAED - 0 ITERM1	
12 LOWER BOUND UPPER BORND 1400 6.0 1.10000000 00 1.10000000 00 1.0000000 00 1.0000000 00 1.0000000 00 1.00000000	-01 7.33277870-01 5.37696440-03 00 1.08749250 00 0.0 2.14730680 00 7.33277870-01
1 1 1 0 0 0	0.0053789644
2.1473068231 0.733277674	
IN ROSBIE DF - CO.O.	
2.1973080 00 1,33277870-01	
IN NOSBIK N = 1 E(N) - 0.1000000000-01	10-00000
12 LOWER BOUND UPPER BOUND 4-00-01 1-10000000 00 1-00-00-01 1-10000000 00 1-00-00-01 1-10000000 00 1-00-00-01 1-10000000 00 1-00-00-00-00 1-00-00-00-00-00 1-00-00-00-00 1-00-00-00-00 1-00-00-00-00-00-00-00-00-00-00-00-00-00	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0
MENIA N METAG MEUCC	0.0053769644
2.1573066231 0.77	1.40
IN ROSBRK OF - 0.0	
2.1873080 00 7.33277870-01	10-0
IN ROSBRK N = 1 E(N) = 0,30000000000-01	10-00000
800MD UPPER BOUND 00 01 6.59950090-01 1.10000000 00 39196010-03 X = 2.18	UND DCCURILZ) PENALTY -01 7.33277876-01 5.37696440-03 00 1.10385950 00 1.48956770-05 2.18730880 00 7.33277870-01
PILL I TO 2	0.0053918601

2-15730680 00 7-43277870-01	
E(N) =	
LOWER BOUND UPPER BOUND OCCUR(12) PENALTY -1.00000000 01 6.59950090 01 1.08694540 00 0.0 -1.01000000 00 1.08694540 00 0.0	
1 = 1 10 = 2 2.1573068231 0.7432778747	
IN ROSBIK DP = -0.5000000-02	
7411 1 1 10 2 2.15730680 00 7.28277870-01	
IN ROSBAK N = 2 E(N) = -0.5000000000-02	
LOWER BOUND UPPER BOUND OCCURITZ) PENALTY -1.00000000 01 6.59950090 01 7.28277870 01 0.0 F. 4.66868650 03 X = 2.15730680 00 7.28277870 01 A H H37AG N3UCC	
PLIA I = 1 TO 2 0 1 0.004668665 2.1573068231 0.7282778747 IN ROSBNY DP = 0.1282778747	
00 68 90	
IN RDSBRK N = 2 E(N) = -0.1500000000-01	
LOWER BOUND UPPER BOUND OCCUR(12) PENALTY -1.00000000-01 6.59950090-01 7.13277878-01 2.84385299-03 0.0 1.10000000 00 1.10067440 00 4.54806690-07 1.10000000 00 1.1006740 00 7.13277870-01	
PILL 1 = 1 TO 2 0.7132778747	
IN ROSBRK DP -	

430 0.0 450 0.0 #FEVA F = 6.1584755D-04 X = 2.15730680 00 6.682772D-01	is p
-	
2.1573068231 0.6662778747	
IN ROSBEK DP0.18500000 00	9
2.15730680 00 5.33277870-01	11
IN RDSBAR N . 2 E(N)0.1350000000 00	18
12 LONER BOUND UPPER BOUND OCCUR(12) PERALTY 2 1.0000000 01 6.59950090 01 5.33277070-01 0.0 400 0.0 1.10000000 00 1.20667330 00 1.13791910-02 96584 F x 1.13791910-02 X x 2.15730680 00 5.33277870-01	
MINIS 1 10 2 0 0.0113791914 PILL 1 - 1 TO 2 0 0.5332778747	
C MATRIX FAOM GRAN 0.1520571843 -0.0883716977 -0.9883715977 -0.1520571843	
C MATRIX FROM ROSBRK	
1+520-01 -9+880-01	
PILL 1 A 1 TO 2 0. 0006138476 PLIL 1 A 1 TO 2 2.1573068231 0. 6682778747	
IN ROSBAK UP = -0.49418570-02	
2.158067 ID 00 6.63336 02D-01	
IN RUSSEK N = 1 E(N) = 0.5000000000-02	
12 LOWER BOUND UPPER BOUND UCCURIEZ PENALTY 2 -1.00000000 01 6.5995090-01 6.63936020-01 1.14645150-05 400 0.0 0.0 1.1265650 00 6.95194040-04	

1				
1		20-05		
RESSER N = 1 ETN = -0.2500C0000-32	1	10-01		
I	1 E(N) -	2000000-32		
	BOUND UPPER BOUND	6.70748809-01 1.12189649 00	94536 94536	
2.1569266801 0.6707488039 2.1569266801 0.674127880-02 2.15578630 00 6.78161590-01 2.15578630 00 6.78161590-01 1.90606080 01 1.10000000 00 1.11750930 00 3.06 0.0 6.38234270 04 8 2.15578630 00 5.78161590 0.0 6.38234270 04 8 2.1557862930 00 5.78161590 0.0 6.38234270 04 8 2.1557862930 00 5.78161590 0.32122080-01 0.49418580-02 0.32122080-01 0.49418580-02 0.32122080-01 0.49418580-02 0.32122080-01 0.49418580-02 0.1557862912 0.49418580-02 0.1557862910 01 0.49418580-02 0.1557862910 01 0.49418580-02 0.1557862910 01 0.49418580-02 0.1557862910 01 0.49418580-02 0.1557862910 01 0.49418580-02 0.1557862910 01 0.49418580-02	I I I I	0990965000		
IN ROSBAK DF =	2.1569266801 0.6	98039		
IN RDSBAK N = 1 E(N) = -0.750000000-02	1	7860-02		
### CCCURITY ##	00	590-01		
### BOUND UPPER BOUND C. 784615790-01 3-31 3-31 3-31 3-31 3-31 3-31 3-31	E(N) =	20-000000		
11	80UNO 80 01	6.78161598-01 1.11750930 00 578630 00 6.78	9+31658899-04 3.06576380-04 61590-01	
1 10 2 2.1557862512 0.6781615916 2.1557862512 0.49418580-02 0.32122080-01 2.18904889 00 0.19418580-01 2.18904889 00 0.32500000000-01 0.32500000000-01 0.32500000000-01 0.32500000000000000000000000000000000000	H HSTAG NSUEC	00005382353		
6.7569066D-01 6.7569066D-01 UPPER BURNO 00CUR(12) 6.59950090-01 6.75690660-01 1.10000000 00 1.13405520 00 1.1 SUCE 0 3 X 2.18904888 00 6.756906	2.1557862512 0.6	61 591 6		
60 6.7569066D-01 E(N) = -0.3250000000D-01 WD UPPER BUND 0CCUR(12) 1.1000000D 00 1.1340552D 00 1.1 5.2210 03 K = 2.1890468B 00 6.756906 6. MSUCE	1	8580-02		
2 EIN) = -0.32500000000-01 80UND UPPER BOUND 0CCURITZ) 2.4 8000 01 6.59950090-01 6.75690660-01 2.4 1.10000000 00 1.13405520 00 1.1 40752210-03 K 2.18904880 00 6.756906 NSTAG NSUCE	8	10-0990		
LOWER BOUND UPPER BOLNO 0000000 2.4 -00000000 01 0.59950090-01 0.7569060-01 2.4 -0000000 01 1.10000000 00 1.13405520 00 1.1 -0 1.40752210 03 K 2.18904880 00 6.756900	2 E(N) =	10-00000000		
N NSTAG NSUCE	LOWER SCUND UPPER SD -1.00000000 01 6.59950090 0.0 1.10000000	6.75690668 1.13405520	4-6	
2.1890487603 0.67	12 2.1890487603 0.67	U 0.0014075221 56906624		

The course bound 0.599.0090-01 0.682778	## #058## # 2 - E(#) 6.1625000000-01	
NTRIA N NSTAG NSUCC 0.0003325336 PIII 1 170 2 0.0003325336 PIII 1 170 2 0.4822772747 PIII 1 170 2 0.4822772747 PIII 1 170 2 0.4875800000000000000000000000000000000000	-1.00000000 01 6.5950090-01 6.68277870-01 6.00000000 01 11622290 00 2 F = 3.32533440-04 X = 2.14086560 00 6.6827	
	14 N NSTAG NSUCC 13 2 1 10 2 0.0003325336 1 - 1 10 2 0.6682778747	
	-	
TE		
MTRIA M NSTAG NSUCC 0.0000015187 2.0026825188 0.6406650869 1.0026825188 0.6406650869 1.0026825188 0.6406650869 1.0068088 0.6406650869 0.6406650869 0.6406650869 0.6406650869 0.6406650869 0.6406650869 0.6406650869 0.6406650869 0.6406650869 0.64066699 0.6406699 0.6406699 0.6406999 0.6406999 0.6406999 0.6406999 0.6406999 0.6406999 0.6406999 0.6406999 0.6406999 0.6406999 0.6406999 0.6406999 0.6406999 0.6406999 0.6406999 0.6406999 0.64069999 0.64069999 0.64069999 0.64069999 0.64069999 0.64069999 0.64069999 0.64069999 0.64069999 0.64069999 0.64069999 0.640699999 0.640699999 0.640699999 0.640699999 0.6406999999 0.64069999999999999999999999999999999999		
IN ROSBAR DE	1 - 1 TO 2 0.64	
1. 1. 1. 1. 1. 1. 1. 1.		
N ROSBRK N = 2 E(N) = 0.14625000000 00 11	1 - 1 TO 2 00	
# LUNER BOUND UPPER BOUND 0.3862672 500 0.0 1.0000000 01 5.59950000-01 6.3862672 500 0.0 1.0000000 01 1.10000000 00 1.0692420 5 1.00000000 01 1.10000000 00 1.0692430 660 0.0 17688 01 1.10000000 00 1.0692430 660 0.0 17688 01 1.10000000 00 1.0692430 660 0.0 17688 01 1.10000000 00 1.0692430 660 0.0 17688 01 1.10000000 00 1.0692430 660 0.0 17688 01 1.10000000 00 1.0692430 660 0.0 17688 01 1.10000000 00 1.0692430 660 0.0 17688 01 1.10000000 00 1.0692430 660 0.0 17688 01 1.10000000 00 1.0692430 660 0.0 17688 01 1.100000000 00 1.0692430 660 0.0 17688 01 1.10000000 00 1.0692430 660 0.0 17688 01 1.100000000 00 1.0692430 660 0.0 17688 01 1.100000000 00 1.0692430 660 0.0 17688 01 1.100000000 00 1.0692430 660 0.0 17688 01 1.100000000 00 1.0692430 660 0.0 17688 01 1.100000000 00 1.0692430 660 0.0 17688 01 1.100000000 00 1.0692430 660 0.0 17688 01 1.100000000 00 1.0692430 660 0.0 17688 01 1.100000000 00 1.0692430 660 0.0 17688 01 1.100000000 00 1.0692430 660 0.0 17688 01 1.100000000 00 1.0692430 660 0.0 17688 01 1.100000000 00 1.0692430 660 0.0 17688 01 1.100000000 00 1.0692430 660 0.0 17688 01 1.100000000 00 1.0692430 660 0.0 17688 01 1.100000000 00 1.0692430 660 0.0 17688 01 1.00000000 00 1.0692430 660 0.0 17688 01 1.000000000 00 1.0692430 660 0.0 17688 01 1.000000000 00 1.0692430	- EINI -	
TZ LUNER BUND UPPER BUND OCCUR 2 -1.0000000 01 5.5950090-01 5.362672 500 0.0 1.1000000 00 1.0692430 KAED - 0 17ERM - 1	1.00000000 01 6.59950090-01 6.38626720-01 0.0	
12 LUMER BOUND UPPER BOUND OCCUR 400 0.0 1.00000000 01 5.599509000 00 1.0692430 1.10000000 00 1.0692430 KRED - 0 17ERM - 1.04813320 00	F = 0.0 X = 1.94813320 00	
KAEB 0 TERM	-1.00000000 01 4.59950000 05 38626720-01 7.0	
FOURT BOCKET - HOPE ANIMA	0-17ERM -	
5.74764050-01 6.3862672	-1.00000000 01 5.74764050-00 6.38626720-01 4.07844090-03	
F = 4.07844090-03 X = 1.94813320 00	F = 4.07844090-03 X = 1.94813320 00	

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1 0.1000000-01	0.0	PR officer is not retained to the second
4 5	720-01 0000000-01 0000000-01	4 B 4
F = 4.0784409	1050-01 6-36 2000 00 1-01 = 1.9581332	₩ 15 12 12 12 12 12 12 12 12 12 12 12 12 12
1 ROSBAR DP - 110 2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	0.00	
MK N = 1 - F(1)	t) = -0.3000000000-01 -UPRER BOUND00CURTIZI PENALTY 5.74764050-01 6.38626720-01 4.07844090-03 1.10000000 00 1.07849090 00 0.0 10.000000 00 1.07849090 00 0.0	1 2 - 1
NTRIA N NSTAG NSUCC 3 1 1 0 2 P(1) 1 = 1 T0 2	0_00407844.09	- I am Miller in
1 0.9000000-01 1 0.9000000-01	0.0	1
IR ROSDRK N = 1 E(N) = 0. IE	## 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	

0.10000000-01

- 14 ROSBAK N 2 - FIM) 8.18880000000-61	
2 -1.00900000 01 5.74764050-01 6.48626720-01 5.45569440-03	· · · · · · · · · · · · · · · · · · ·
F - 5.45569440-03 X - 1.98813320 00 6.48	
PILL I = 1 TO 2 PILL I = 1 TO 2	
1 0.0	
P(1) 1 = 1 TO 2	
Z -1.00000000 01 5.74764050-01 6.33626720-01 3.46481420-03	
20-03 X = 1.98813320 00	
MTRIA M MSTAG MSUCC 0.0034648143 4 1 0 2 4 1 0 2 1.0041331590 0.4334267237	
IN ADSEAR OF0.1500000-01	
Piti I = 1 TO 2 1.00013320 00 4.10026720-01	
10-C00000000000000000000000000000000000	
2 -1.00000000 01 5.74764050-01 6.18626720-01 1.92393400-03	
A N NSTAG NSUCC 0.0019239340	
I 0.0 -0.4500000-01	
MIT 1 = 1.10 2	

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1		
	Z NSTAG NSUCC	
IN ROSSME DF =	1.9881331590 0.57	
	ROSAN DF -	
IN ROSENK N = 2 E(N) = -0.1350000000 00 IN ROSENK N = 2 E(N) = -0.1350000000 00 IN ROSENK N = 2 E(N) = -0.1350000000 00 IN ROSENK N = 2 E(N) = -0.1350000000 00 IN ROSENK N = 1 E(N) = -0.1350000000 00 IN ROSENK N = 1 E(N) = 0.2000000000 00 IN ROSENK N = 1 E(N) = 0.2000000000 00 IN ROSENK N = 1 E(N) = 0.2000000000 00 IN ROSENK N = 1 E(N) = 0.2000000000 00 IN ROSENK N = 1 E(N) = 0.2000000000 00 IN ROSENK N = 1 E(N) = 0.2000000000 00 IN ROSENK N = 1 E(N) = 0.20000000000 00 IN ROSENK N = 1 E(N) = 0.20000000000 00 IN ROSENK N = 1 E(N) = 0.20000000000 00 IN ROSENK N = 1 E(N) = 0.20000000000 00 IN ROSENK N = 1 E(N) = 0.20000000000 00 IN ROSENK N = 1 E(N) = 0.20000000000 00 IN ROSENK N = 1 E(N) = 0.20000000000 00 IN ROSENK N = 1 E(N) = 0.20000000000 00 IN ROSENK N = 1 E(N) = 0.20000000000 00 IN ROSENK N = 2.20000000000 00 IN ROSENK N = 2.20000000000 00 IN ROSENK N = 2.20000000000 00 IN ROSENK N = 2.2000000000000 00 IN ROSENK N = 2.200000000000 00 IN ROSENK N = 2.	1.98813320 00	
II LOWER BOUND UPPER BOUND OCCURIIZ OCCURII	2 E(N) =	
##### # WSTAG WSUCC 0.436626727 00 6.36626 ##### # WSTAG WSUCC 0.436626727 00 4.36626 1.9881331590 0.4366267237 0.5240974257 0.62240974274 0.62242140 0.62462140 0.624622140 0.624622140 0.624622140 0.624622140 0.62462140 0.6246242140 0.624622140 0.6246242140 0.6246242140 0.6246242140 0.62462424140 0.62462424140 0.62462424140 0.6246	LOWER BOUND UPPER BOUND OCCURITY	
C MATRIX FROM GRAM C. MATRIX FROM GRAM C. MATRIX FROM GRAM C. MATRIX FROM GRAM C. MATRIX FROM GRAM C. MATRIX FROM ROSBRX 5.24D-01 -8.52D-01 4.7814 1. 9681331590 0. 5736.26727 IN ROSBRK CP - 0. 20000000000000 IIN ROSBRK CP - 0. 20000000000000 IIN ROSBRK CP - 0. 20000000000000 IIN ROSBRK CP - 0. 20000000000000 IIN ROSBRK N - 1 F(N) - 0. 2000000000000 III ROSBRK N - 1 F(N) - 0. 20000000000000 2 -1.00000000 01 5.7476.4050-01 2 -1.000000000 01 5.7476.4050-01 2 -1.000000000 01 5.7476.4050-01 2 -1.000000000 01 5.7476.4050-01 3 -1.000000000 01 5.7476.4050-01 456993560-01 47814 1. 99861510 00 5.856993560-01 47814 1. 99861510 00 5.856993560-01 47814 47817 47814	0.0 1.1920542D 00 8.4739739D-03 X - 1.9881332D 00 4.36626	
C MATRIX FROM GRAN -0.8516583167 -0.8516583167 -0.8540974257 C MATRIX FROM ROSBRK 5.240-01 -8.520-01 -5.240-01 -8.520-01 -5.240-01 -8.520-01 -5.240-01 -8.520-01 -5.240-01 -8.520-01 -5.240-01 -8.520-01 -6.11 - 1.00 -6.10 -8.520-01 -6.11 - 1.00 -6.10 -8.520-01 -6.11 - 1.00 -6.10 -8.520-01 -6.10 -8.10 -8.10 -6.10 -8.10 -8.10 -6.10 -8.10 -8.10 -6.10 -8.10 -8.10 -6.10 -8.10 -8.10 -6.10 -8.10 -8.10 -6.10 -8.10 -8.10 -6.10 -8.10 -8.10 -6.10 -8.10 -8.10 -6.10 -8.10 -8.10 -6.10 -8.10 -	FA IF NSTAG NSUCC	
C MATRIX FROM GRAM -0.6546974257 -0.6516583167 -0.5240974257 C MATRIX FROM ROSBRK 5.240-01 -8.520-01 -0.5240974257 C MATRIX FROM ROSBRK 5.240-01 -8.520-01 -0.5240974257 C MATRIX FROM ROSBRK 5.240-01 -8.520-01 -0.5240974257 -0.524097427 -0.5240974257 -0.5240974257 -0.5240974257 -0.5240974257 -0.5240974257 -0.5240974257 -0.5240974257 -0.5240974257 -0.5240974257 -0.5240974257 -0.5240974257 -0.5240974257 -0.5240974257 -0.5240974257 -0.5240974257 -0.5240974257 -0.5240974257 -0.5240974257 -0.524097427 -0.5240977 -0.524097427 -0.524097427 -0.524097427 -0.524097427 -0.524097427 -0.524097427 -0.524097427 -0.524097427 -0.524097427 -0.524097427 -0.524097427 -0.524097427 -0.524097427 -0.524097427 -0.524097427 -0.524097427 -0.524097427 -0.524097427 -0	1.9881331590 0.43	
S.240-01 -8.520-01		
######################################	C MATRIX FROM ROSBRK	
	5.240-01 -8.520-01 -8.520-01 -5.240-01	
IN ROSBAK EF O. 17033170 01 IN ROSBAK EF I. 9786151D 00 5.5659356D-01 IN ROSBAK N . I F(N) 0.2000000000-01 IZ LOWER SCHWO UPPER BOLWO OCCUR(12) 2 -1.00000000 01 5.7475405D-01 5.5659356D-01 400 0.0	THE RESTANCE	
5.56593560-01 5.56593560-01 W) = 0.2000000000-01 UPPER BOLND	1.9881331590	
5.5659356D-01 W) = 0.2000000000-01 UPPER BOUND 0CCUR(12) 5.7476405D-01 5.5659356D-01 1.10000000 00 1.12242140 00 5.0272 50-04 X = 1.99861510 00 5.5659356D-	10.00.00	
MD UPPER BOLND OCCURITZ) 01 -5.74764050-01 5.56593560-01 0.0 1.10000000 00 1.12242140 00 5.0272 21050-04 x = 1.99861510 00 5.56593560-		
1.00000000 01 5.74764050-01 5.56593560-01 0.0 0.0 1.10000000 00 1.12242140 00 5.0272 7 5.02721050-04 x = 1.99861510 00 5.56593560-	E(N) -	
	LOWER SOUND UPPER BOLND DCCUR(12) -1.00000000 01 -5.7476405D-01 5.5659356D-01 0.0 0.0 1.10000000 00 1.1224214D 00 5.0272	

5.82143	•33
IN RUSSAK N . 1 EIN0.100000000-01	
LOBER 1.000000	
##### ################################	
IN NOSEM, DP	
1.96716930 00 6.07693060-01	
IN NOSBAL M . 1 E(N)0. 3000000000-01	
12 LOWER ROUND UPPER BOUND DECURITZ) PENALTY -2-L-000000C.J-01-5-74784950-01-6-07693060-01-1-08431940-03 400 G.0 1.100000000 00 1.08720430 00 0.0 460 F. 1.08431940-03 X - L-06716930 00 4.07693060-01	
##### # ##############################	
IN ABSENK OF O. LTO331 70-01	
2.010571 ID 00 5.99176470-01	
1201	
#### # ##### #########################	
IN ACCORD OF THE CO. DELICED OF	

PENS F - 1.06937370-05 X -	70-05 X = 1.96905270 00 5.73626720-03	
HTRIA N HSTAR MSUCC		
1 2 1 110	0.0000186937	
1.9690527371	0.5736267237	*
IN ROSBRK OF	0.2854-0750-01	* * * * * * * * * * * * * * * * * * *
1.92753440 00 5.4	5.4807697D-01	
IN ROSBAK N * 2 E(N) = 0.	0.4875000000-01	
12 LOWER SOUND UPPER BOUND 400 0.0 1.10000000 00 0.0 1.10000000 00 0.0 1.10000000 00 0.0 0.	50-01 5.48076978-01 0.0 PENALTY 50-01 5.48076978-01 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0	
===	0.0000405374	
1.9275343941 0.	0.5480769742	
0.0000000000	-1.0000000000	
C MATRIX FROM ROSBAK		
-1.000 00 -4.070-15 1.770-15 -1.000 00		
PILI I - 1 TO 2 0.5	0.0000186937	**
IN ROSBIK DP - 0-5000000-020-1	-0.08643500-17	
1.974052 70 00 5.7	5.73626720-01	
IN RUSBRK N = 1 E(N) = -0.50000 12 LUNER BOUND UPPER BOUND 2 -1.00000000 01 5.74764050-01 400 0.0	-0.5000000000-02 R BOUND DCCUR(12) PENALTY 4050-01 5.73626720-01 0.0 0000 00 1.10574470 00 3.30019210-05	

IN ADSBAK OF =	
1. 96655270 00 5. 73626720-01	
IN ROSBAK N = 1 E(N) = 0.2500000000-02	
12 LOWER BOUND UPPER BOUND OCCURATED PENALTY 400 0.0 1.31391170-05 L 1.0000000 00 1.1036.24.80 00 1.31391170-05	
#FRIA WSTAG WSUCE 0.0000131391 PILL 1 = 1.10 2 0.5736.267237	
IN ROSBAK DF = 0.13294530-14	
P415 1 - 1 TO 2 - 5.73426720-01	
IN ROSBAK N * 1 E(N) * 0.7500000000-02	
12 LOWER BOUND UPPER BOUND OCCURITY) PENALTY 450 0.0 1.10600000 00 1.10157520 00 2.48115010-06	
MRIA N NSTAG NSUCE 0. 0000024812	
IN ADSSAIK DF .	
1.93655270 00 5.73626720-01	
IN ROSBRK N = 1 E(N) = 0.2250000000-01	
12 LOWER BOUND UPPER BOUND OCCURITZ) PENALTY -2 -1.00000000 01 -5.7474-4050-01 5.73426720-01 0.0 1.10000000 00 1.09584820 00 0.0 1.10000000 00 1.09584820 00 6.73426730-01	
2 -1.00000000 01 5.74764050-01 5.73626720-01 0.0 PENALTY 400 0.0 1.10000000 00 1.09584820 00 0.0 eFEV* F = 0.0 X = 1.93655270 00 5.73626720-01	
UPPER BOIND OCCURETY	
2 -1.0000000 01 5.16244050-01 5.73624720-01 3.29047620-03	

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WSUCC	
1.9365527371 0.5736.267237	
IN ROSBAK DP	
1.94655270 00 5.73626720-01	
IN ROSORK N . 1 E(N) - 0.1000000000-01	
12 LOWER BOUND UPPER BOUND OCCUR(12) 400 0.0 1.09831540 00 1.10000000 00 1.09831540 00 4650 5 3.30047430 03 X 1.04455370 00 6.734	3.2904762D-03
1 0 0 0032904762 1 10 2 0 0 0 5736267237	
IN ROSARK DP . 10-00-0000-01-0-0-0-0-0-0-0-0-0-0-0-0-0	
1.97655270 00 5.73426720-01	
IN ROSBAK N = 1 E(N) = 0.300000000-01	
12 LOWER SOUND UPPER BOLNO GCCUR(12) 1 400 0.0 1.10646700 00 1.10646700 00 4 1.0646700 00 1.10646700 00 4 1.076570 00 5.7342	3.29047620-03 4.1822099D-05 26720-01
" WIRTH WATAG WSUCC 0.0033322983	
IN ROSBRK OF *	
### # 1 TO 2 5.83626720-01	
IN ROSBRK N - 2 FEM3 - 0.1000000000-01	
12 LOWER BOUND UPPER BOUND OCCUR(12)	PENALTY 0.0

Takan di kacamatan kacaman palebah kacamatan

2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	1.9465527371	0.5836267237	
	ROSBAK DP	-0.5000000-02	
In totake w = 2 etw 0.500000000-02 PERALTY PER	1.946557 TO 00	5.68626770-01	
	Z ECH)	-0* 200000000-05	
	1.00000000 01 0.0	6000 00 000000000000000000000000000000	
	1 72	0.0027424821 0.5686267237	
	IN ROSANK OF -	0-1 #00 0000-01	
12 120 12 12 12 12 12 12	111 1 - 1 TO 2 TO 00	5, 53426 720-01	
#### # #\$727371 0.553426727 PENALTY ***********************************	~	-0.1900000000-01	
IN NOSAK N - 2 E(N)0.45000000-01 IN NOSAK N - 2 E(N)0.450000000-01 IN NOSAK N - 2 E(N)0.4500000000-01 II LOWER BOWND UPPER BOWND OCCURILLY 2 -1.0000000 01 5.1624020-01 12 LOWER BOWND UPPER BOWND OCCURILLY 2 -1.0000000 01 1.10000000 01 1.12293350 01 1.00055410-03 4504	0 00000000 1-0	5.53626728-01 1.10642305 00	
IN NOSBNK DF0.45000000-01 1.94455270 00 5.0862672D-01 IN ROSBNK N - 2 E(N)0.4500000000-01 2 -1.00000000 01 5.162405D-01 5.0862720-01 400 0.0 1.10000000 00 1.13299330 00 1.088554 PEUT 1.00000000 01 5.162405D-01 5.0862720-01 400 0.0 0.0 1.10000000 00 1.13299330 00 1.088554	1.12	91.02	
1.94455270 00 5.08626720-01 1.04655270 00 5.08626720-01 1.0468 8040 00 00 00 00 00 00 00 00 00 00 00 00		-0.48000000-01	
LOWER SOUND UPPER BOUND OCCURILLY PE 1.000000000 01 5.08626720-01 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0	1.946552 TO 00	S.08626 72D-01	
1.0000000 01 5.124-035-01 5.00424720-01 0.0 0.0 1.10000000 00 1.13299330 00 1.088554 0.0 1.0000000 00 1.13299330 00 1.088554	~	-0-4500 00000 00-01	
2 0 2 0 2 0 2 0 0 2 0 0 0 0 0 0 0 0 0 0	10 00	8000 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	
	17	3	

	UPPER BOUND DCCURITZ!	
SUCC 0.174.24.723.7 -0.174.24.723.7 -0.174.24.723.7 -0.174.24.723.7 -0.152.057184.3 -0.4983.71497.7 -0.4983.7 -0.	F = 1.48575630-02 X = 1.94655270 GO	
##### FROM SOAM	1. 1 TO 2	
######################################		
	- 1-520-01 -1-520-01 -9-880-01 -1-520-01	
1	NSUCC P. P.	
1		
#TRIA N NSTAG NSUCC 0. 501264050-01 5.03684877-01 5.0368487-01 5.	20-00000000000-05	
HTRIA N NSTAG WSUCC P(1) 1 = 1 TO 2 1.9473430230 0.50 1	1.00000000 01 5.16264050-01 5.0368487F-01 0.0000000 01 1.13607640 00 0.00 1.13607640 00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	
14 ROSERR DF - 100 - 03 P(1) I = 1 TO 2 1,94617269 90 1W 1758RR W = 1 E(W) =	- 1 TO 2 1-9473130730 0.50	
P(1) 1 = 1 TO 2 1,94617269 00 1W x 758RK W = 1 - E(W) =		
IN x 15868 W = 1 - E(W) =	1.94617260 00	
	IN x 15868 W = 1 - E(W) =	

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1.9461725941 0.5110976530	
Pill I = 1 T0 2 1.04503230-00 5-10510440-01	
+H ROSBAR H 1 - E(N) 75000000000-02	
2 -1.00000000 01 5.16264050-01 5.18510440-01 5.04626510-06 400 0.0 0.0 1.10000000 00 1.12693370 00 7.25424610-04 4FEV* F - 7.30470860-04 X - 1.94503220 00 5.18510440-01	
-	
P(1) I = 1 TO 2 1-9450321452 0-5185104407	
I -0.34212870-02 0.22238360-01	
Pill 1 - 1 TO 2 1-94161090 00 5.40748800-01	
## ROSONE # = - 1 - E1#1 =0.225000000-01	
2 -1.00000000 01 5.16264050-01 5.40748800-01 5.99503110-04 400 0.0	
MTRIA N MSTAG MSUCC	
0.84	
1 0.32122 060-01 0.49418580-02	
P(1) 1 = 1 TO 2 1-977154-20 00 5-23452300-01	
## ROSSKK N =- 2 - E(N) = -0.3250000000-01	
16 LOWER DOUND UPPER DOUND OCCURRATED PENALTY 2 -1.00000000 01 5.16364050-01 5.23452300-01 5.16709070-05 400 0.0 1.16260240-03	
MTRIA N MSTAG MSUCC.	

1.92897110 00 5.16039510-01	
0000000 01 5.16264	
##### # #\$740 #\$000 14 2 1 10 2 1411 1 = 1 10 2 1.9289711251 0.5160395115	
IN ROSSHX DF = -0.74127880-02	
1. 1. 1. 10 2 5.08626720-01	
	4
12 LOWER BOUND UPPER BOLND OCCUR(12) PENALTY -2-1-0000000 01 5-10-264050 01 5-00626720-01 0.0 400 0.0 1-10000000 00 1-1158538D 00 2-5134317D-04 FEVE F = 2-5134317D-04 K = 1-8007880D 00 5-00626720-01	
##### # #\$T## #5UCC 0.0002513432 ###################################	
IN ROSBAK DP = -0.22238360-01	
1.73623860 00 4.86388360-01	
80UND UPPER 009-01 5.16264 1.10000 55784160-05-X	
MIRIA N MSTAG MSUCC 0.0000455784 16 2 1 3 0.0000455784 P411 1 = 1 10 2 0.4063883605	
IN ROSBRK OF -	

C MATRIX FROM GRAM C MATRIX FROM GRAM -0.1051526103 -0.10515261		12 LOWER BOUND 2 -1-00000000 01 400 0.0	R BOUND UPPER BOUND DCCUR(12) PENALTY HOCOD 01 5.16264950-01 4.19673270-01 0.0 1.23055520-02 1.2000000 00 1.21093040 00 1.23055520-02 1.23055520-02 1.3005050-02 1.3005050-01	
C MATRIX FROM ROSERR -9.94-01 1.050-01 -1.050-01 1.050-01 -9.94-15-09-68 -9.94-10-01 1.050-01 -9.94-10-01 1.050-01 -9.94-10-01 1.050-01 -9.94-10-01 1.050-01 -9.94-01 1.050-01 -9.94-10-01 -9.94-10-01	C MATRIX FROM ROSBRX -0.1051526103	14 % WSTAG		
C MATRIX FROM GRAM -0.1051526103 -0.1051526103 -0.1051526103 -0.1051526103 -0.1051526103 -0.1051526103 -0.1051526103 -0.1051526103 -0.1051526103 -1.1072 -	C MATRIX FROM GRAM -0.0944560968 -0.1051526103 -0.9944560968 -0.1051526103 -0.9944560968 -0.1051526103 -0.9944560968 -0.0000455784 -1.105-01 1.005-01 -1.105-01 1.005-01 -1.105-01 1.005-01 -1.105-01 1.005-01 -1.105-01 1.005-01 -1.105-01 1.005-01 -1.105-01 1.005-01 -1.105-01 1.005-01 -1.105-01 1.005-01 -1.105-01 1.005-01 -1.105-01 1.005-01 -1.105-01 1.005-01 -1.105-01 0.005-01 -1.105	1	0.4196732709	
C MATRIX FROM ROSBRK -9.940-01 1.050-01 -1.050-01 -9.940-01 -1.050-01 -9.950-01 -1.150-01	C MATRIX FROM ROSBRK -9.940-01 1.050-01 WITHIN WISTAG WSUEE 0.0000455784 1.7362386441 0.486383605 IN ROSBRK DP 0.486384120-01 IN ROSBRK N = 1 E(N) = -0.500000000-02 IN ROSBRK N = 1 E(N) = -0.5000000000-02 IN ROSBRK N = 1 E(N) = -0.5000000000-02 IN ROSBRK DP 0.000000000 01 1.10680730 00 4.86914120-01 WITHIN WISTAG WSUEE 0.000000000 00 1.10680730 00 4.86914120-01 IN ROSBRK DP 0.000000000 01 1.10680730 00 4.86914120-01 IN ROSBRK DP 0.000000000 01 1.10680730 00 4.86914120-01 IN ROSBRK N = 1 E(N) = 0.26288150-03 FILL LOWER SOUND UPPER SOUND 02 0.10673340 00 4.86120-01 IN ROSBRK N = 1 E(N) = 0.26288150-03 FILL LOWER SOUND 02 0.10000000 01.10673340 00 4.86120-01 WHALA W WSTAG WSUEE 0.0000000 01.10673340 00 4.86120-01 FILL LOWER SOUND 03 0.10000000 01.10673340 00 4.86120-01 FILL LOWER SOUND 03 0.10000000 01.10673340 00 4.86120-01 FILL LOWER SOUND 03 0.10000000 01.10673340 00 4.86120-01 FILL LOWER SOUND 04000 00 01.10673340 00 4.86120-01 FILL LOWER SOUND 05 0.10000000 01.10673340 00 4.86120-01 FILL LOWER SOUND 05 0.10000000 01.10673340 00 4.86120-01 FILL 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	MATRI		
WTRIL	WTAIL	MATRI	-0.9944560968	I I
WTRIF	WTMIN	-9.940-01 1.050-01		
IN ROSBRX DP = 1 E(N) = -0.55950000000-02 IN ROSBRX N = 1 E(N) = -0.5000000000-02 IZ : GMER BGUND UPPER BGUND GCCUR(12) 0.0 III : T41210970 00	IN ROSBAX DP =	1	9.0	
IN ROSBRK N = 1 E(N) = -0.5000000000-02 12	IN ROSBAK N = 1 E(N) = -0.500000000-02 12			11
IN ROSBRK N = 1 E(N) = -0.5000000000-02 12	IN ROSBRK N = 1 E(N) = -0.5000000000-02	1.74121090	1	
12 10WER BOUND UPPER BOUND OCCURILL21 O.0	12 COWER BOUND UPPER BOUND LEGISTICAL 0.0 4.00 0.0 1.10000000 01 1.1060013D 00 4.65 4.00 0.0 1.10000000 00 1.1060013D 00 4.65 4.00 1.100000000 0.0 0.0000463393 0.00 0.0000463393 0.00 0.0000463393 0.00 0.000000000 0.00 0.00000000			
NTRIA N NSTAG NSUCE 0.0000463393 1	NTRIA N NSTAG NSUCE 0.0000463393 1.7412109246 0.4669141236 1.7412109246 0.4669141236 1.7412109246 0.4669141236 1.7412109246 0.26288150-03 1.74121092246 0.26288150-03 1.73375250 0.2568060606000-02 1.73375250 0.2568060606000-02 1.26264050-01 1.0673340 0.4612540 0.46	00000000 01 0.0	50-01 4.86914120-01 0.0 50-01 1.10680730 00 4.6 00 00 1.10680730 00 4.869141	
IN ROSBRK DP = -0.26288150-03 -10.24861440-02 -1. 13375250 00	BRK UP =	1	0.4	
IN RYSBIK N = I E(N) = 0.25080600000-02 IL LOWER BOUND UPPER BOWN 0CCUR(12) -2 -1.00060000-01 -5.16264050-01 -4.86125480-01 0.0 400 0.0 1.10000000 00 1.100673340 00 4.55	IN RYSORK N = 1 E(N) = 0.2500000000-02 IL LOWER BOUND UPPER BOUND 0CCUR(12) 2 -1.00000000-01 -5.16264050-01 4.86125480-01 400 0.0 1.10000000 00 1.1067334D 00 4.554000000 00 0.1067334D 00 4.5125400000000000000000000000000000000000	IN ROSBAK DP		
IN RTSBRK N = I E(N) = 0.25000000000-02 12 LOWER BOUND UPPER BOUND 0CCUR(12) -2 -1.000000000 01 -5.16264050-01 4.86125480-01 0.0 400 0.0 I.100000000 00 1.1067340 00 4.5540000 0.0	IN RTSBRK N = I E(N) = 0.25080608080-02 IZ LOWER BOUND UPPER BOUND 0CCUR(IZ) 2 -1.00808080-01 5.16264850-01 4.861254880-01 0.0 400 0.0 1.10873340 00 4.861254 NTRIA N NSTAG NSUCE 0.0080453382	1.73375250 00	4.86125480-01	
	4 N NSTAG NSUCG 0. 0000453382	12 LOWER BOUND 12 -1.00000000-01 100 0.0	8040 0CCURITZI 50-01 4.86125480-01 0.0 00 00 1.10673340 00 4.55	

1.7337525039	0.4861254790	
IN ROSBRK DP = 1 0.74584210-02	-0.78864460-03	
1.72629410 00	4. 85336630- 31	7
IN ROSBRK N = 1 E(N) =	0.7500000000-02	
12 LOWER BOUND UPPE 2 -1.00000000 01 5.1626 4.00 0.0 1.1000	R BOUND UPPER BOUND OCCUR(112) PENALTY 0000 01 5.16264050-01 4.85336830-01 0.0 4.51708210-05 1.10000000 00 1.10672090 00 4.51708210-05 4.51708210-05 X - 1.72629410 00 4.86336830-01	
######################################	0.0000451708	
IN ROSBAK DP = 22375260-01	-0-23659340-02	1
1.70391880 00	4.82970 900-01	
IN ROSBRK N = 1 E(N) =	0.2250000000-01	
12 LOWER BOUND UP.YER BOUND 400 0.0 1.10000000 00 00 1.10000000 00 00 00 00 00 00 00 00 00 00 0	ER BOUND OCCUR(112) PENALTY (44050-01 4=82970900-01 0=0 x 1.10705250 00 4.97382120-05 x 1.70391880 00 4.82970900-01	
	0.0000497382	
IN ROSBRK DP = 1106740-01	0-10503940-00	
1,73,40080 00	3,80297410-01	
IN ROSBRK N = 2 EIN) =	0.1056250000D 00	
R BOUND 90000 01	UPPER BOUND OCCUR(III) PENALTY 5.1626405D-01 3.8029741D-01 0.0 1.1000000D 00 1.1610685D 00 3.7293587D-02 190-03 X 1.7374008D 00 3.8029741D-01	
22 2 2 2 0 0 0 1 1 1 1 1 1 1 1 1 1 1 1 1	0. 00.372.935.87	
1	0.3802974092	

- #M ROSBRK N =- 2 - E(N) = -0.52812500000-01	
1.0000	
074070 00	
MINIA N NSTAG NSUCC 0.0004662359 PILI I = 1 TO 2 1.2207402109 0.5372545470	
1 0.27766860-02 -0.2625960-01	
P(1) 1 = 1 TO 2 1. 728070080 JO 4. SP074.080-01	
10-08	
* F * 3.54807460-04 X =	
HERIA H HSTAG HSUCC 0.0008548075 PILI I = 1 TO 2 0.450076781	
I -0.1383430-02 0.13129930-01	
P(1) 1 = 1 TO 2 1.724805 TO 00 4.8444 740-01	
IN ADSENT II - 2 E(N)0.13203125000-01	
12 - LOWER BOUND UPPER BOUND OCCURITZE PENALTY 2 -1.00000000 01 5.16.26.4050-01 4.98466760-01 0.0	
- X 90-054	
MTRIA N NSTAG NSUCC U 0.0000010631	
I -0.41650290-02 0.39389780-01	
P(1) 1 = 1 TO 2	

										***			TY 033		
4.64235870-04 X = 1.72074070 00 5.37854550-01 WS746 WSUCC 0.0004662359	0.5378565470	0.7292538977			0 0.000010631	U. 4984667626	0.3616247B-02	5.02113030-01	- 0.500000000000000000000000000000000000	S.16264050-01 5.02113030-01 0.0 1.10000000 00 1.09941200 00 0.0 X = 1.72148450 00 5.02113030-01	\$.16264050-01 5.02113030-01 0.0 1-10000000 00 1.09941200 00 0.0 x = 1.72148450 00 5.02113030-01	-1	4.51901730-01 5.02113030-01 2.52117500-03 1.10000000 00 1.09941200 00 0.0 00-03-X = 1.72148450 00 5.02113030-01	0.5021130321	646
WENT F . 4.64235870-0	1.7207407109	C MATRIX FROM GRAN -0.4843431970 0.7292538977	C MATRIX FROM ROSBRK	7.200-01 7.290-01	24 1 3 3 4 15 16 15 15 2	1.7249057401	IN ROSBAK DF .	1.7214050 00	IN ROSBRK N - 1 FEN)	12 1.046R BOUND 5 45 46 64 64 64 64 64 64 64 64 64 64 64 64	2 -1.00000000 01 5	KRED . 0 ITERM .	12 LOWER BOUND UPPER 400 0.0 1.1000 4EEV* F = 2.52117500-03.X	1 10 2 7214845241	10 000000001 10 01 0000 01

2.52117500-03 12 LOWER BOUND UPPER BOUND OCCUR(12) P. WILTY 400 0.0 0.0 1.10000000 00 1.10142290 00 2.02451730-06 6694 F. 2.52318950-03 X - 1.74148450 00 -5.02113030-01 3-42540100-03 12 LOWER BOUND UPPER BOUND DCCURIL2) PENALTY 2 -1.00000000+ 01 4.51901730-01 4.97113030-01 2.04406190-03 400 0.0 1.10104770 00 3.41404860-06 4.97113030-01 12 LONER BOUND UPPER BOUND OCCUR(12) PER 2 -1.00000000 01 4.51901730-01 5.02113030-01 2.5211750 30 0.0 1.09972720 00 0.0 1.10000000 00 1.09972720 00 0.0 12 LOWER BOUND UPPER BOUND OCCUR(12) PEN 400 0.0 1.0000000 01 4.51401730-01 5.12113030-01 3.4254014 400 0.0 1.09558000 00 0.0 46EVR F. 3.62540100-03 X 1.73148450 00 5.12113030-01 0.0025211750 0. 0025231995 0.0036254010 0-1000000000000000 -0.50000000000-02 0.5021130321 5.02113030-01 0. 5021 130321 5-12113030-01 0.5121130321 4.97113030-01 1 EIN) -IN ROSORK N - 1 EINS -IN ROSORK N = 2 ECH) -IN ROSBRK N = 2 EENS -IN ROSBAK DP -1.76140450 00 1.7314845241 1.7614845241 1.73146450 00 1.73148450 00 1-7314845241 HILL 1 - 1 10 2 Pill 1 - 1 TO 2 - - 1111 1 - 1 10 2 IN ROSPRE DP -IN ROSBRK N -IN ROSBAK DP . Ī 1-243

130001110000	
1 0.0 -0.15000000-01	
P(1) 1 = 1 TO 2 	
12 -1.00000000 01 4.51901730-01 4.82113030-01 9.12722840-04 400 0.0 -1.10000000 00 1.10839670 00 7.05040830-05 •FEV* F = 9.83226920-04 X = 1.73148450 00 4.82113030-01	
HTRIA H NSTAG NSUCC	To a superior
9.4	
1 0.0 -0.4500000-01	
P(1) 1 = 1 TO 2 1e73148450 004.37113030-01	
TH ROSEKK N = -2 -E(N) =0.+500000000-01	
2 -1.00000000 01 4.51901730-01 4.37113030-01 0.0 400 0.0 -1.10000000 00 1.12973109 00 8.83934780-04 *FEV* F = 8.83934780-04 X = 1.73148450 00 4.37113030-01	AR. T.
MTRIA N NSTAG NSUCC	
0-43	
1 0.0 -0.1350000 00	
P(1) I = 1 T0 2 1.73140450 00 3.02113030-01	
IN ROSBIK N = 2 - E(N) = -0.1350000000 00	
12 LOWER SOUND UPPER SOUND UCCUR(112) C PENALTY 2 -1.00000000 01 4.5190173D-01 3.0211303D-01 0.0 400 0.0 -1.10000000 00 1.2089216D 90 1.1863926D-02 9FEV* F = 1.1863926D-02 X = 1.7314845D 00 3.0211303D-01	
NTRIA N NSTAG NSUCC	

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4 : z

	0.1520571843 -0.9883716977	-0.088371.6677 -0.1520571843	
1.52D-01 -9.88D-01 1.52D-01 1.52D-02 1.52D-03 1.52D-02 1.52D-03 1.52D-02 1.52D-03 1.52D-02 1.52D-03 1.52D-03 1.52D-03 1.52D-03 1.52D-03 1.52D-03 1.52D-03 1.52D-03 1.52D-03 1.52D-03 1.52D-03 1.52D-03 1.5DD-03			
NEAL			
IN ROSBRK DP = 1 E(N) = 0.44418580-02 IN ROSBRK N = 1 E(N) = 0.5000000000-02 IZ LOWER BOUND UPFER BO	1.12	934	
IN ROSBAK N = 1 E(N) = 0.500000000-02 IZ LOWER BOUND UPPER BOUND 0.00217117B-01 400 0-0.0000000-0-1.45190173B-0-1 400 0-0.0000000-0-1.12224480 00 1.1322972D 00 1 RESPECTOR 1.10000000 00 1.1322972D 00 1 RESPECTOR 1.10000000 00 1.1322972D 00 1 RESPECTOR 1.10000000 00 1.1322972D 00 1 IN ROSBAK DP = 1 E(N) = 0.24709280-02 IN ROSBAK N = 1 E(N) = -0.25000000000-02 IZ LOWER BOUND UPPER BC.ND 0.4284618D 00 8-10000000 0.0 1.1284618D 00 8-100000000 0.0 1.1284618D 00 8-10000000 0.0 1.1284618D 00 8-1000000000 0.0 1.1284618D 00 8-10000000000 0.0 1.1284618D 00 8-10000000000 0.0 1.1284618D 00 8-100000000000 0.0 1.1284618D 00 8-100000000000 0.0 1.1284618D 00 8-1000000000000000000000000000000000	IN ROSBRK DP =	-0-491.8580-02	
IN ROSBRK N = 1 E(N) = 0.5000000000-02 IZ LOWER BOUND UPPER BOUND 1.3229720 00 11.32270 0	1,73244 80 00	4.32171170-01	
IZ	1 E(N)	0.500000000-02	
NYRIA	-1.00000000-01 0.0	30UND 0CCUR(12) 90-04 4.32171178-04 0 00 00 1.13229720 00 1.	
IN ROSSBAK DP =	- ALTH 1 - 1 - 1 - 1 - 1 - 1 - 1	0.0010431075 0.4221711736	
IN ROSBAK N = 1 E(M) = -0.2500000000-02 IZ LOWER BOUND UPPER BCAND DCCUR(12) 2 -1.00000000 01 4.3958396D-01 0.0 400 0.0 0.0 1.10000000 06 1.1284618D 00 8.1007 FEVA F E 1.0074280-04 E 1.7311044D 00 4.39583960 WIRLA N NSTAG NSUCE 0.0008100744 1.7311043811 0.4395839613 IN ROSBAK DP = 1 FD 2 0.7412788D-02 PHI I = 1 FD 2 0.7412788D-02	=-	0-24709290-02	
IN ROSBAK N = 1 E(N) = -0.2500000000-02 IZ LOWER BOUND UPPER BC.ND DCCUR(IZ) 2-1.00000000 01 -4.51001440-01 4.39583960-01 0.0 400 0.0 400 0.0 1.10000000 06 1.1284618D 00 8.1007 FEVA F E 1.0074300 04 X 1.7311044D 00 4.395839613 IN ROSBAK DP = 0.74127880-02 P(1) 1 = 1.70 2 1.7311043811 0.4395839613 P(1) 1 = 1.70 2 P(1)	1.73110440 00	4.3958396D-01	
NTRIA N NSTAG WSUCE 10 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	IN ROSBAK N = 1 E(12 LOWER BOUND 2 -1.00000000 01 400 0.0	000000-02 0CCUR(12) 4-39583460-01 1,12846180 00 8-1007	,
IN ROSBRK DP 1 -0.11404.290-02 +111 1 - 1 - 10 - 2 1.72996400 00 4	MERTA 10 PHI 1 - 1	3	
1.72996400 00	E.	0.74127880-02	
	1.72996400	4.46996750-01	
The state of the s	Z -1.00000000 01 4.5190	4.51901730-01 4.44996750-01 0.0 PENALTY	

11 1 2 0.0006105559	
1.7299639523 0.4469967490	
IN ROSORK DF =	
### # # 100 \$ 4.69235110-01	
IN RDSBAK N = 1 F(N) = -0.2250000000-01	
IZ LOWER BOUND UPPER BOUND DCCUR(IZ) PENALTY 2 -1.0006000 01 1.51001730-01 4.69235110-01 3.00446180-04 400 0.0 400 0.0 400 0.0 400 0.0 400 0.0	
HTRIA - H HSTAG WSUCC 0.0004950560	
P(11 1 - 1 TO 2 0.4692351122 0.4692351122	
IN RDSBRK OF	
++++ + - 1 +0 -2 -0 5.35950200-01	
IN ROSBRK M = 1 F(N) = -0.6750000000-01	
12 LOWER SOUND UPPER BOUND OCCUR(112) PENALTY 2 -1.00000000-01 -1.10000000 00 1.08615440 00 0.0 400 0.0 5 7.04414580-02 X 1.71427880 00 5.35050200-01	
NTRIA N NSTAG WSUCC 0.0070641458 13 1 1 3 0.0070641458 P411 1 = 1 TO 2 0.5359502018	
IN ROSDRIK DP = 0-49418580-02	
#### 1 TO 2 4.74176970-01	
IN ROSBRK W = 2 E(N) = -0.3250000000-01 IZ LOWER BOUND UPPER ROLNO CCCUR(12) +.96186400-04 2 -1.60600000 01 +.51961730-01 4.74176970-01 4.96186400-04 400 0.0 1.10000000 00 1.11403310 00 1.96928630-04	

1.754647458 0.4741769707	
1	
IN ROSSEK N = 2 E(N) = 0.1625000000-01	*
12 LOWER SOUND UPPER BOUND OCCUR(12) PENALTY 2 -1.00000000 01 -4.51801739-01 4.46744180-01 2.20892540-04 400 0.0 1.10000000 00 1.11431470 00 2.04909300-04	
3966	
IN ROSBAK OF0.74127880-02	
1	7
- (%)	
12 LOWER BOUND UPPER BOLND DCCUR(:2) PENALTY -2 -1.0000000 01 4.51401730-01 4.59551400-01 5.54975300-05 400 0.0 1.10000000 00 1.11703110 00 2.90059280-04 6544 F 3.45554810-04 X 1.44228850 00 4.58151400-01	
##### - # #\$746 #\$WC6 0.0003455568 16 2 1 2 0.0003455568 #### 1 = 1 TD 2 0.4973513953	
IN ROSBIN DP = -0.22238340-01	w. 24-419 () to 1
1.51774910 00 4.37113030-01	
IN ROSBRE N = 2 E(N) = 0.14625000000 00	
12 LOWER BOUND UPPER BOUND OCCURII2) PENALTY 2 -1.0000000 01 4.5190130-01 4.3713030-01 0.0 1.10000000 01 1.13979020 00 1.58326200-03 FEVE F 1.58334300 01 X 1.5177410 00 4.37113030-01	
##### ################################	
C MATRIX FROM GRAN -0.9520265613 -0.3060091804 0.3060091804 -0.9520285613	

C MATRIX FROM 405 BAK	
3.060-01 -9.520-01 3.060-01 -9.520-01	
WIRIA W MSTAG WSUCC 0.0003455568	
-	
IN NOSBRK DF = -0.4972 6490-02	
1.67776900 00 4.54378750-01	6
IN ROSBAK N = 1 E(N) = -0.1625000000-01	
12 LOWER SOUND UPPER BOLMD OCCURILZ! PENALTY 2 L. 0000000 01 L.513015000 01 L.51370750 01 6.13541430 04 400 0.0 1.10000000 00 1.11909750 00 3.64716.200-04 66VA F. 3.70851820-04 X = 1.67776900 00 4.54378750-01	
1	
1.6777569694 0.4543787461	
IN MOSBAK DF	
1.65456330 00 4.61837720-01	
IN ROSSBAK N = 1 E(N) = 0.8125000000-02	
12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 2 1.00000000 01 1.5100130-01 1.41837720-01 9.87239170-05 400 0.0 1.10000000 00 1.11616500 00 2.61306260-04 4FEVE F = 3.60030130-04 X = 1.65456330 00 4.61837720-01	
**	
1.65456.32732 0.4618377199	
IN ROSBAK DF	
1.66616610 00 4.58108230-01	
IN ROSBRK N = 1 FIN) = -0.4062 5000000-02	
12 LOWER BOUND UPPER BOUND (ICCUR(12) PENALTY 2 -1-00000000 01 1-51001730 01 1-55100730 05	

1 222				7581430-04 1716200-04 10-01						0-01 0.0 0 00 9.78870190-04 4.27167300-01		1000				PENALTY 9890950-04 9979550-04 00-01
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	1	LOWER BOUN 0.0 0.0 F = 3.5461	IN NOSONK N . 2 EIN)0.4062500000-02	4.50373000-01 4.51401730-01 4.51401730-01 4.51401730-01 4.6147505000000-02 4.61475050-01 4.61475050-01 4.61475050-01 4.61475050-01 4.61475050-01 4.61475050-01 4.61475050-01 4.61475050-01 4.61475050-01 4.61475050-01 4.61475050-01 4.61475050-01 4.61475050000-02 4.61475050000-02 4.61475050000-02 4.614750170-01 4.56174420-01 4.56174420-01 4.56174420-01 4.56174420-01 4.56174420-01 4.66554450 00 4.66174420-01 4.66554450 00 4.66174420-01 4.66554450 00 4.66174420-01 4.66554450 00 4.66174420-01 4.66554450 00 4.66174420-01 4.66554450 00 4.66174420-01 4.66554450 00 4.66174420-01 4.66554450 00 4.66174420-01 4.66564650 00 4.66174420-01 4.665646159
4:		23157 00250 7	LOWER BOUND UPPER BOUND UCCURILZ! PENALTY 1.000000000 0 1.11584770 00 2.51150510-04 2 2 2 0 0.0003526384 1.6674092836 0.4619758490 1.66554450 00 4.56174420-01 1.66554450 00 4.56174420-01 1.66554450 00 1.11834010 00 3.3559920-04 1.000000000 01 4.51901730-01 4.56174420-01 1.000000000 01 4.51901730-01 4.56174420-01 1.000000000 01 4.51901730-01 4.56174420-01 1.000000000 01 4.51901730-01 4.56174420-01 1.000000000 01 4.51901730-01 4.56174420-01 1.000000000 01 4.51901730-01 4.56174420-01 1.000000000 01 4.51901730-01 4.56174420-01 1.000000000 01 4.51901730-01 4.56174420-01 1.0000000000 01 4.51901730-01 4.56174420-01	26 2 2 0 1 = 1 to 2 1.6655445401 0.45
LOWER BOUND UPPER BOUND OCCUR(12) PENAL -1.000000000 01 4.51901730-01 4.56174420-01 1.82559920- 0.0 1.11834010 00 3.35559990- F = 3.54615920-04 X = 1.66554450 00 4.56174420-01 H H3746 W30CC 0.0003546159	LOWER BOUND UPPER BOUND DCCUR(IZ) PENAL -1.000000000 01 4.51401730-01 4.56174420-01 1.82559320- 0.0 1.11834010 00 3.35559990- F = 3.54615920-04 X = 1.66554450 00 4.56174420-01		UPPER BOUND OCCUR(12) PENALTY 4.51901730-01 4.61975850-01 1.01467900-04 1.10000000 00 1.11584770 00 2.51150510-04 40-04 X = 1.66740930 00 4.61975850-01 0.4619758490 4.56174420-01	2 E(N) -
LOWER BOUND	LOWER BOUND UPPER BOUND DCCUR(172) PENAL-L-000000000-01 4.51401730-01 4.56174420-01 1.82559320-0.0 1.100000000 00 1.11834010 00 3.35559900-0.0 F. 3.54615920-04.X 1.66554450 00 4.56174420-01	2 E(N) -	UPPER BUMNO DECUR(12) PENALTY 4.51901730-01 4.61975850-01 1.01487900-04 1.10000000 00 1.11584770 00 2.51150510-04 10-04 X - 1.46740939 00 4.61975850-01 0 0 0.0003526384 0.4619758490	
#.5617442D-01 UPPER BOUND	## 56174420-01 WPPER BOUND	:	LOWER BOUND UPPER BOUND OCCUR(12) 1.0000000H-01 4.51901730-01 4.61975850-01 1.0146790D-04 0.0 1.1000000H-01 4.51901730-04 F- 3.52630410-04 X - 1.66740930 00 4.61975850-01 2 2 2 0 0.0003526384 1.6674092836 0.4619758490	ADSBRK OF
4.56174420-02 4.56174420-01 UPPER BOUND 0CCUR(12) 4.51901730-01 1.10000000 00 1.11834010 00 3.36359990- 20-04 x = 1.66554450 00 4.56174420-01 50-04 x = 0.0003546159	4.56174420-01 4.56174420-01 4.56174420-01 UPPER BOUND		LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 1.00000000	1.6674092836
## 1	# - 6.19758490 4.5617442D-01 4.5617442D-01 UPPER BOUND		-1.00000000	2 2 100 2 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
### 0. 1938080-02 4.56174420-01 4.56174420-01 4.56174420-01 UPPER BOUND	### 0.4619758490 4.56174420-01 4.56174420-01 4.56174420-01 4.56174420-01 4.51901730-01 4.51901730-01 4.56174420-01 1.100000000	0.46 0.46 4.561		LOWER SOUND UPPER BOUND OCCURIIZ) PENALTY -1.0000000FH 01 4.51901730-01 4.61975850-01 1.014877908-04 0.0 1.110000000 00 1.11584770 00 2.51150510-04 F = 3.52630410-04 X = 1.66740930 00 4.61975850-01
UPPER BOUND	UPPER BOUND OCCURITZ! PENALTY **SINDITAD-01 4.61975859-01 1.01467908-04 1.10000000 00 1.11584770 00 2.51150510-04 10-04 X = 1.66740930 00 4.61975850-01 0.4619758490 4.56174420-01 UPPER BOUND OCCURITZ! PENALTY 4.51901780-01 4.56174420-01 1.82559320-04 1.10000000 00 1.11834010 00 3.36359990-04 20-04 X = 1.66554450 00 4.56174420-01	WPER BOUND OCCURITY 4.51901730-01 4.51901730-01 4.51901730-01 4.51901730-01 4.51901730-01 4.56174420-01 4.56174420-01 4.56174420-02		
###	### ### ##############################	#.61975850-01 WPFR BOWN	1.46740930 00 4.61975850-01	MOSBIN, DF - 0.38174140-02
### ### ##############################	#. 61975850-01 #. 61975850-01 UPPER BUUND	### ### ##############################		 1.4636797967
0.4503 730009 4.61975 850-01 4.61975 850-01 4.51901730-01 4.61975 850-01 1.10000000 00 1.11584770 00 2.51150510-04 10-04 X - 1.65740930 00 4.61975 850-01 0.461975 8490 4.56174420-01 4.56174420-01 4.56174420-01 4.56174420-01 4.56174420-01 3.36359990-04 50-04 X - 1.66594450 00 4.56174420-01 900000000 00 1.11834010 00 3.36359990-04 50-04 X - 1.66594450 00 4.56174420-01 9000	0.101110 00 4.61975850-01 4.61975850-01 1.10000000 00 1.11584770 00 2.51150510-04 1.10000000 00 1.11584770 00 2.51150510-04 1.10000000 00 1.11584770 00 2.51150510-04 1.0000000 00 1.11584770 00 2.51150510-04 1.0000000 00 1.1158470 00 4.61975850-01 4.56174420-01 4.56174420-01 1.10000000 00 1.11834010 00 3.36359990-04 1.10000000 00 1.11834010 00 3.36359990-04	# # # # # # # # # # # # # # # # # # #		2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2
### ### ##############################	### ### ##############################	### 0. 2031 2500 000-02 ### 0. 2031 2500 000-02 ################################		0.0 1.1208627D 00 4.352
1.100.0000 00 1.12086270 00 4.35252260-04 6.4407382523 6.4503730009 4.61975850-01 4.61975850-01 4.61975850-01 4.61975850-01 4.61975850-01 4.61975850-01 4.61975850-01 4.61975850-01 4.61975850-01 4.61975850-01 4.61975850-01 4.61975800000-02 4.61975800000000 4.61975800000000 4.61975800000000 4.56174420-01 4.56174420-01 4.56174420-01 4.56174420-01 4.6619758000000000 4.6619758000000000000000000000000000000000000	1.100.0000 00 1.12086270 00 4.35252260-04 0-0.4. 1.46367930 00 4.50373000-01 4.41975850-01 4.41975850-01 4.41975850-01 4.41975850-01 4.5190139-01 4.61975850-01 0.4003526384 0.4003526384 0.4019758490 4.56174420-01	1.100.0000 00 1.12086270 00 4.35252260-04 6-04 4 - 1.44347900 00 4.50373000-01 0.4903730009 4.41975850-01 4.41975850-01 4.41975850-01 1.10000000 00 1.11584770 00 2.51150510-04 1.10000000 00 1.11584770 00 2.51150510-04 1.10000000 00 1.11584770 00 2.51150510-04 1.10000000 00 1.11584770 00 2.51150510-04 1.10000000 00 1.11584770 00 2.51150510-04 1.10000000 00 1.1158470 00 4.61975850-01 0.10318080-02 4.56174420-01 4.56174420-01	1.100.0000 00 1.12086270 00 4.352 60-04 X - 1.46347980 00 4.50373000 0 0.0004352523 0.4503730009	-1.000000000 01 4.51401730-01 4.57373000-01 0.0
4.51901780-01 4.51901780-01 4.51901780-01 4.51901780-01 4.51975850-01 4.51975850-01 4.51975850-01 4.51975850-01 4.51975850-01 4.51975850-01 4.51975850-01 4.51975800000-02 4.51901780-01 4.51901780-01 4.511975800000-02 4.511901780-01 4.55174420-01 4.55174420-01 4.55174420-01 4.56174420-01	### ### ##############################	4.5171442D-01 4.51401730-01 4.51401730-01 4.51401730-01 4.61475850-01 4.61475850-01 4.51401730-01 4.61475850-01 4.51401730-01 4.51401730-01 4.51401730-01 4.514142D-01 4.5517442D-01 4.5517442D-01 4.5517442D-01 4.5517442D-01 4.5517442D-01	4.51901730-01 4.50373000-01 0.0 1.10000000 00 1.12086270 00 4.3523 60-04 X - 1.46347980 00 4.50373000 0.4503730009 4.61975850-01	. (4) .
#1 * 0.8125000000-02 **SINOTER BOWNO 0.1.1208627D 00 4.3525226D-04 **SINOTER BOWNO 0.1.1208627D 00 4.3525226D-04 **SINOTER BOWNO 0.1.1208627D 00 4.3525226D-04 **SINOTER BOWNO 0.2.51150900-04 **SINOTER BOWNO 0.2.5115081D-04 **SELT442D-01 **	#1 * 0.8125000000-02 UPPEP BOWN	UPPER BOUND CCCURILZ! PENALTY 4.51401730-01 4.59373000-01 1.10000000 01 1.12086270 00 4.35252260-04 1.10000000 01 1.12086270 00 4.35252260-04 4.61975850-01 4.61975850-01 4.61975850-01 4.51901730-01 4.61975850-01 4.5174420-01 4.56174420-01 4.56174420-01 4.56174420-01 4.56174420-01	WPPEP BOUND OCCURITY 0.0 4.51401730-01 4.50373000-01 0.0 1.10040000 00 1.12086270 00 4.3525 60-04 x - 1.46347980 00 4.50373000 506 0.0004352523 6.38674140 02	

400 0.0 1.10000000 00 1.11709020 00 2.92073240-04	1
12	
1.6664769119 0.4590751370	
IN RISSRX DF 0.28007130-02	
1.66740930 00 4.61975850-01	
IN ROSSAK N = 2 E(N) = -0.30468750000-02	
12 LOWER BOUND UPPER BOUND OCCUR(12) PENALTY 400 0.0 0.0 1.01487908-04 1.01487908-04 400 0.0 1.11584770 00 2.51150510-04 400 0.0 4.61775850-01	
MTAIR N HSTAG NSUCC 0.0003526384	
C MATRIX FROM GRAM 0.00778214965 -0.0659716692 0.9978214965	
C MATRIX FROM ROSBRK 9.980-01 6.600 42 -6.600-02 9.980-01	
"	
IN ROSSPIX OF - 0.20268250-02 - 0.13400500-03	
11.66445010 00 4.59209140-01	1
IN ROSORK N * 1 E(N) * -0.20312500000-02	
12 LOWER BOUND UPPER BOUND OCCUR(12) 5.33982860-05 400 0.0 0.0 1.11705860 00 2.90994680-04	

LOWER BBUND UPPER BI 1.00000000 01 4.51901731 0.0 1.1000000	
NTRIA N NSTAG NSUCC 0.0003432836 P(I) I = 1 TO 2 0.4583004208	
I 0.1675062D-04 0.2633531D-03	
P(1) 1 = 1 TO 2 1.67054720 00 4.5905.0480-01	
IN ROSDAK W = -2 - E(N) 0.25590625000-09	
12 LOWER BOUND UPPER BOUND GECURALIZA 2 -1.00000000 01 4.5190173D-01 4.5906.048D-01 5.1247721D-05 400 0.0 1.10000000 00 1.11704120 00 2.9108332D-04 9FEV* F = 3.4233105D-04 X = 1.6705473D 00 4.5906048D-01	
NITALA N NSTAG NSUCC 0.0003423311 PILI I = 1 TO 2 0.4890404802	
IN 805BRK DP - 0.76005930-03	
P(1) I = 1 T0 2 1.67059760 00 4.59820540-01	
	1 + 0 F 0 F 0 F 0 F 0 F 0 F 0 F 0 F 0 F 0
2 -1.00000000 01 4.5190173D-01 4.5982054D-01 6.2707562D-05 400 0.0 1.10000000 00 1.1167351D 00 2.8006228D-04 *FEV* F = 3.4276984D-04 X = 1.6705976D 00 4.5982054D-01	
NTRIA N NSTAG MSUCC 0.0003427698 P(1) 1 = 1 TO 2 0.4598205395	
C_MATRIX_FRBH_GRAM 0.9999935171 0.0036007997 -0.0036007997 0.9999935171	
C MATRIX FROM ROSDRK -1.000 00 3.600-03 -3.600-03 1.000 00	

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P(I) I = 1 TO 2 1.6705473123 0.4590604802	1.0
IN ROSBRY DP = -0.73141240-05	All the second s
P(1) 1 = 1 TO 2 1.67257850 00 4.59053170-01	4 44
IN ROSBRK N = -1 - F(N) = 0.2031250000-02	Ī
12 LOWER BOUND UPPER BOUND OCCUR(112) PENALTY 2 -1.00600000 01 4.5190173D-01 4.5905317D-01 5.1143054D-05 400 0.0	7.
MTRIA N MSTAG MSUCC U	T IZ
P(1) I = 1 T0 2 1-6725785491 0-4590531661	
1 0.60937100-02 -0.21942370-04	
P(1) I = 1 TO 2 1+678672-30-604-59681220-61	
IN RUSBRK W = -1 -E(N) = 0.66937500000-02	
12 LOWER BOUND UPPER BOUND OCCURILY PERMITY 2 -1.00000000 01 4.5190173D-01 4.59031220-01 5.0829696D-05 400 -0.01.10000000 00 -1.11706550 00 -2.91231380-04 *FEV* F = 3.4206108D-04 X = 1.67867230 00 4.59031220-01	
MTRIA N MSTAG NSUCC U 0.0003420611	
P(1) 1 = 1 TO 2 1.6786722596 0.4590312237	
IN ROSCINK DP = 0.45713280-06 0.12695230-03	
P(1) 1 = 1 T0 2 1,67257900 00 4,59180120-01	
1N-ROSBAK N = -2 - £1N}-=0.12695312500-03	
12 LOWER BOUND UPPER BOUND DCCUR(12) PENALTY 2 -1.00000000 01 4.51901730-01 4.59180120-01 5.29749540-05 400 0.0 -1.10000000 00 1.11699980 00 2.88993880-04 9FEV* F = 3.41968830-0^ X = 1.67257900 00 4.59180120-01	
NTRIA N NSTAG NSUCC	
P(1) 1 = 1 TO 2 1-4725190043 0-4501801184	

#0474 #0474 #0474 #0474 #0474 #0474	*DATA				
A 0.625 -0.75 0.625 -1.75 2241-8 1 CCCUR(2) 1.0 IN L 2241-8 1 IRC 400 AMULT 1.0					
ARIABLE FIBONACCI E 3 ICON411 -) IREF 5.25 IPROC 1 LIMIT 1 ALOW - 10.0 UP-1	DATA CALON OLD CONTROL DATA LAED OLD CONTROL CALON OLD C				

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GHE VARIABLE FIBONACCI EXAMPLE	2241-8
12 LOWER BOUND UPPER BOUND OCCUR(12) 400 0.0 1.28850480 01 X = -2,36074270 00	1.28850480 01
12 LOWER BOUND UPPER BOUND OCCUR(12) 400 0.0 1.08133460 00 x = 2,36074270 00	PENALTY 813346D DO
12 LOWER BOUND UPPER BOUND DCCUR(12) 400 0.0 7.71791120 00 7.7179 4FEV* F = 7.71791120 00 X = 5.27851460 00	PENALTY 1791120 00
12 LOWER BOUND UPPER BOWND OCCUR(12) 400-0-0-0-0-0-0-0-0-0-0-0-0-0-0-0-0-0-0	2-30135300 00
12 LIWER BOUND UPPER BOUND 0CCUR(12) 400 0.0 2.35939890 00 X = 3.47480110 00	2.3593989D 00
12 LOWER BOUND UPPER BOUND OCCUR(12) 400 0.0 1.06761460 00 X = 1.67108750 00	1.06761460 00
12 LOWER BOUND UPPER BOUND OCCUR(12) 400 0.0 1.35467780 00 1.35467780 00 1.3546 9FEV* F = 1.35467780 00 X = 1.24668440 00	1.35467780 00
IZ LOWER BOUND UPPER BOUND OCCUR(12) 400 0.0 1.00253290 00 1.00253290 00 1.0025	PENALTY 1.00253290 00
IZ LOWER BOUND UPPER BOUND 0CCUR(IZ) 400 0.0 0.0 0.0 0.0 1.00569900 00 1.0056	1-00569900-00
12 LOWER BOUND UPPER BOUND OCCUR(12) +00 0.0 1.01801180 00 X = 1.83023870 00	PENALTY 1.0180118D-00
12 LOWER BOUND UPPER BOUND DCCUR(12)	PENALTY 3007040 00
12 LOWER BOUND UPPER BOUND OCCUR(12) 400 0.0 0.0 0.0 0.0 1.00112570 00 1.0011 *FEV* F = 1.00112577. 00 X = 2.04244030 00	1.00112570 00
12 LOWER BOUND UPPER BOUND OCCURIIZ) 1.0000	PENALTY 1.0000704D 00
KRED = -1 ITERM = 0	

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IMPUT CARDS READ DATA-BND-0F-JOB		19414		
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IEF2851	SYS68312.T174300.RP002.P2542F.GOSET	PASSED
1EF2851 1Ef 2851	SYS68312, T174300, RP 002, P2542F, GOSET - VOL. SER. NOS - SSSSSS.	DELETED
16F2851	SYSOUT	SYSOUT
IEF2851 IEF2851	SYS68312-1174300, RP002, P2542F R0000005	DELETED
IEF2051	SYS68312-7174306-RP002-P2542F-R0000006	DELETED
1EF2851 1EF2851	SY 568 312. T	DELETED
1672801	K 1847EAKL .P.254ZF	
1		
1		

11/07/68	******KZ40].1' ,X	HOURS, 019.25 MINUTES, START TIME = 17.42.57 NES = 000029 MES = 012617		,	
ASP-J08-NO 0042	//PZ542F JGB (<143E510010,00000,116,030,LCRRAINE,************************************	ELAPSED TIME ON MAIN = SYI = 00.321 HOURS, 019.25 DDNAME = SYSMSG PRINTED ON PRI , LINES = 000029 DDNAME = SYSMUT PRINTED ON PRI , LINES = 012617	LINES DUTPUT FOR THIS JOB = 012646 NO CARD BUTPUT FOR THIS JOB.	INPUT STARTED AT 17.25.36, COMPLETED AT 17.33.28 PRINT STARTED AT 18.04.05, COMPLETED AT 18.17.15 SYSTEM TURNARDUND TIME = 00.91.79	

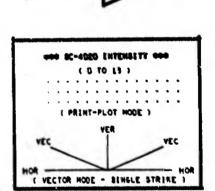
HATCH IS

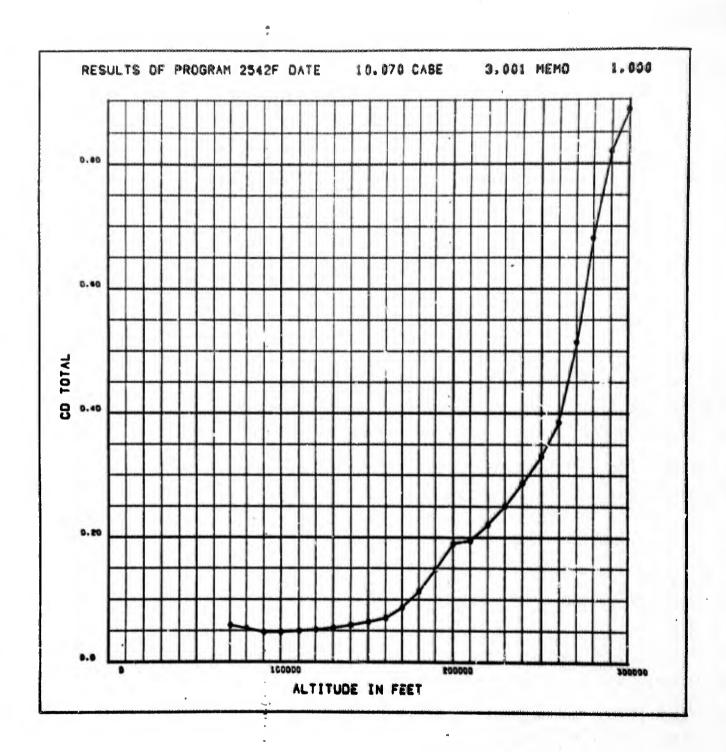
APPENDIX 4

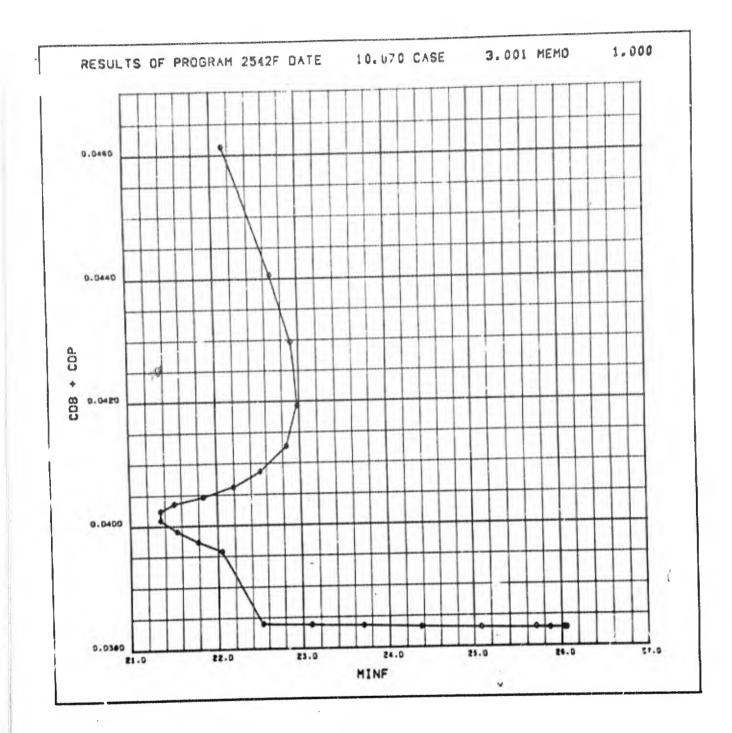
PLOTS FROM SAMPLE PROBLEMS

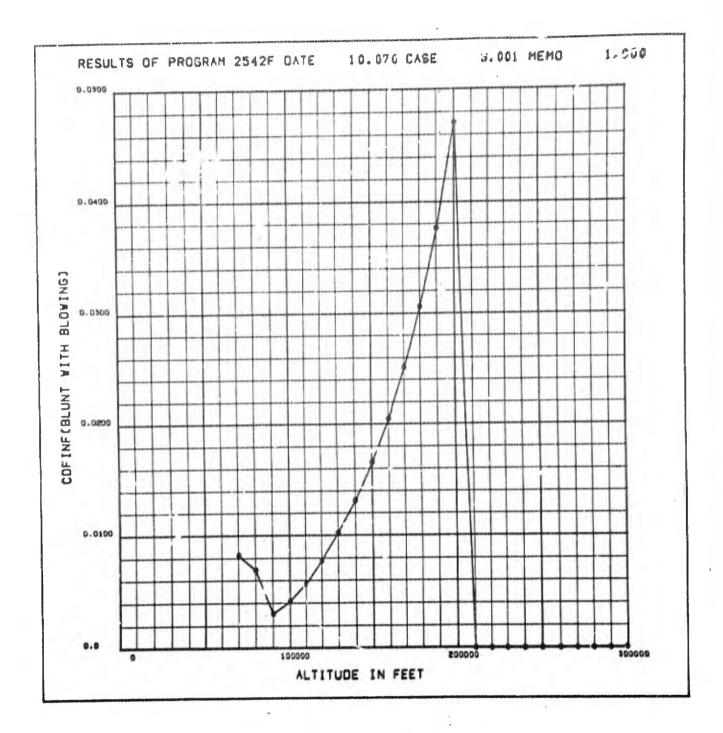


LORAYNE K210 BIN 2542F 7 NOV 1968

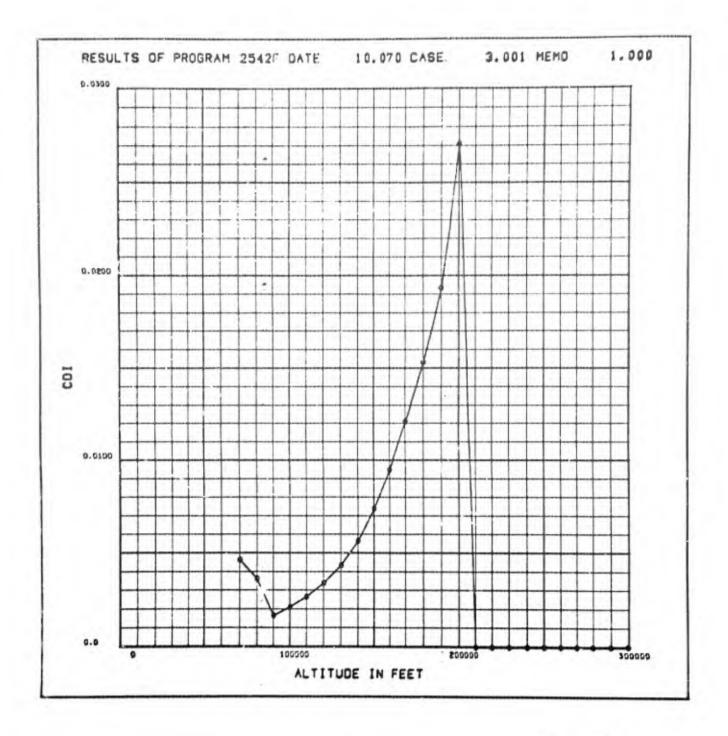




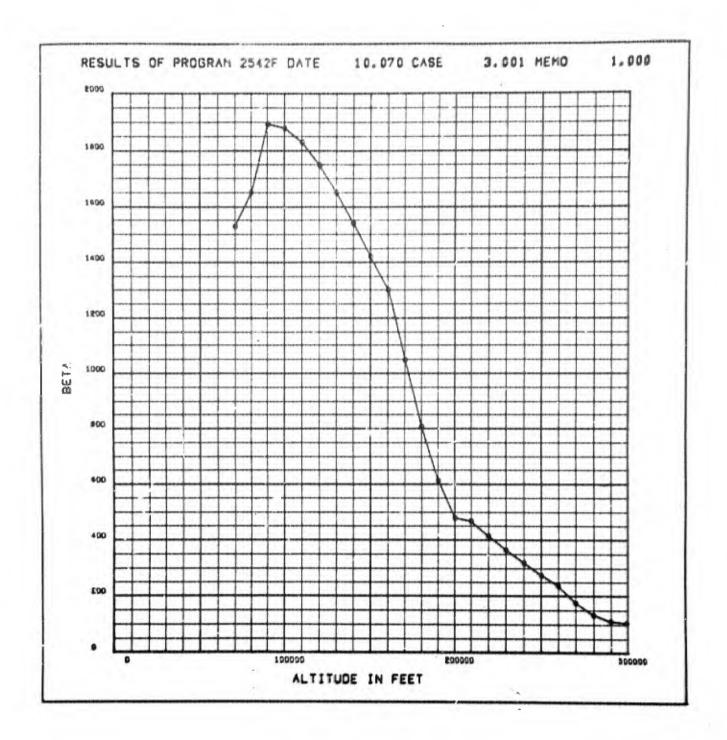




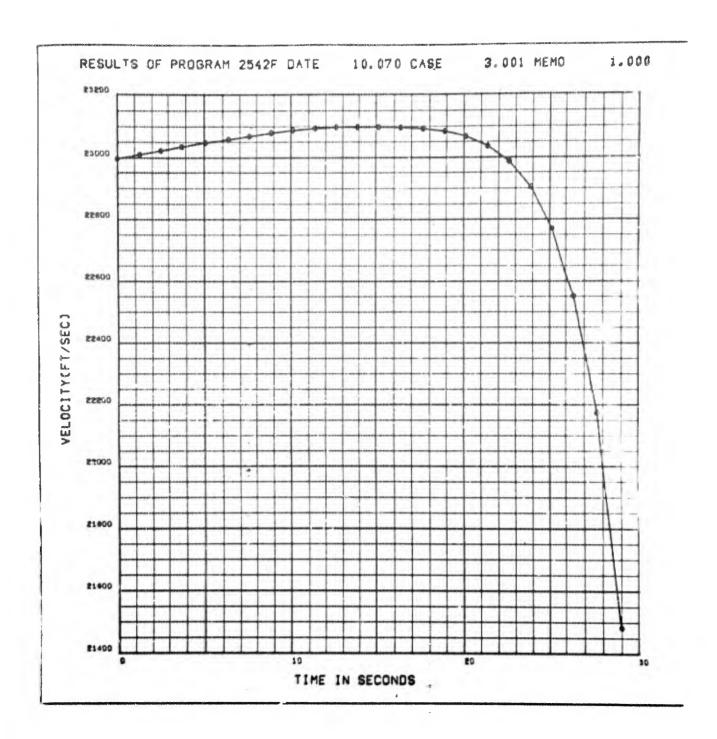
FRAME 4



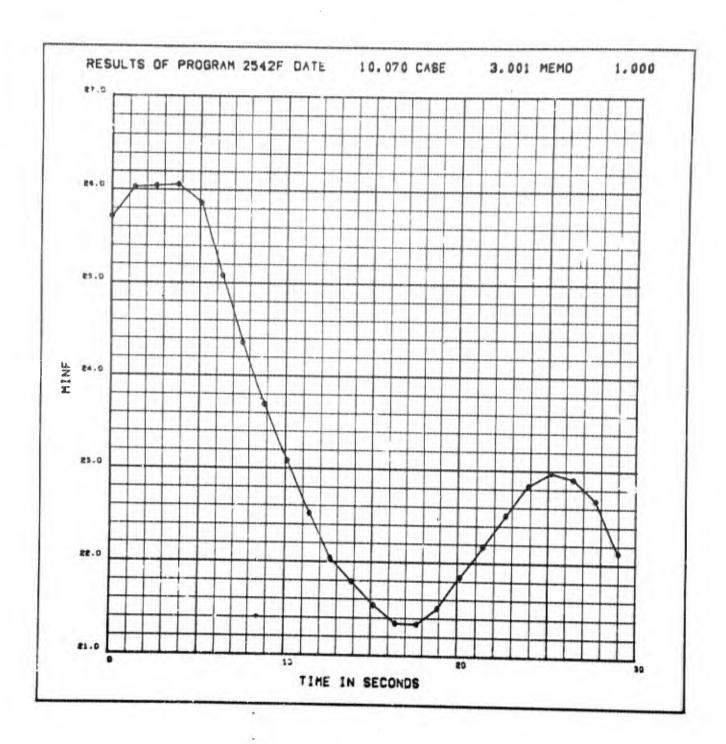
FRAME 5



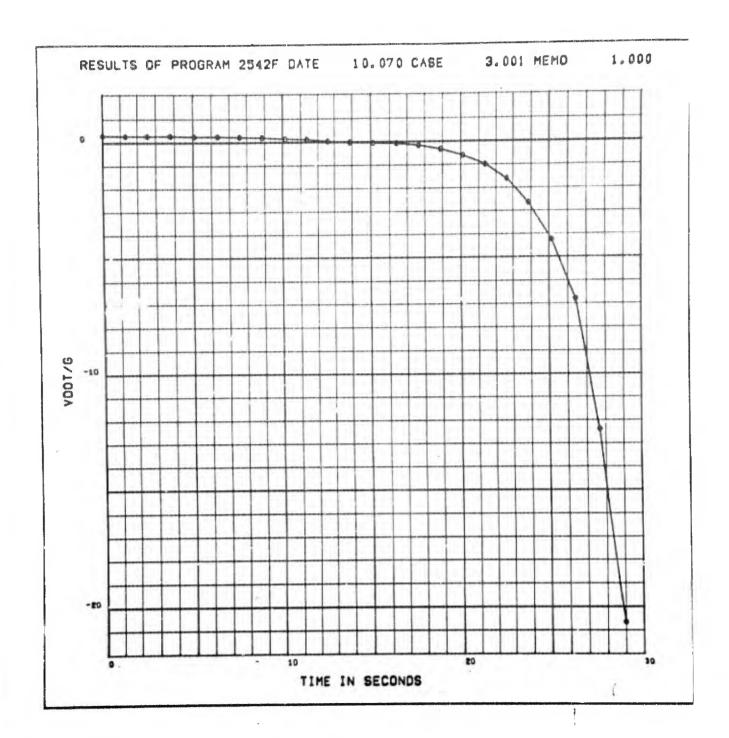
FRAME 6



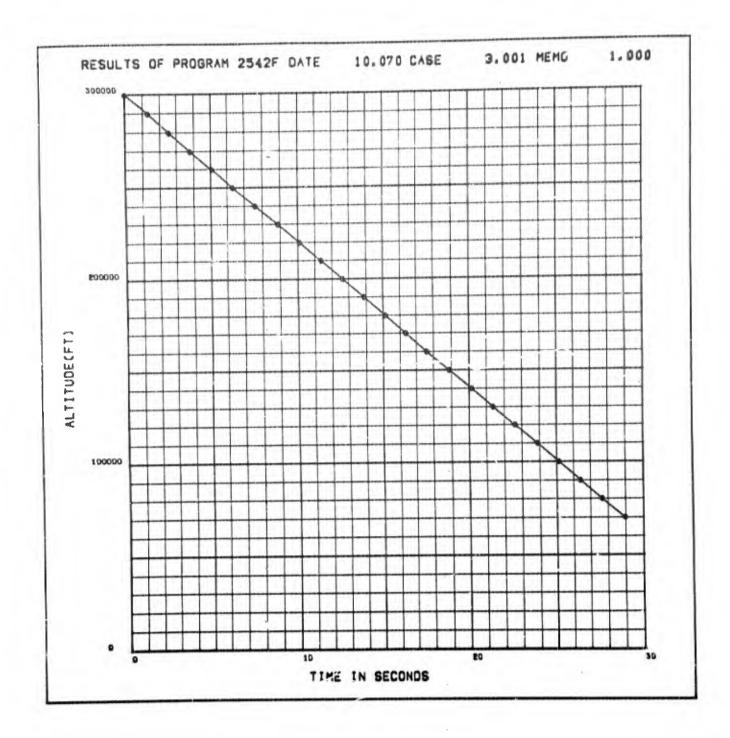
FRAME 7



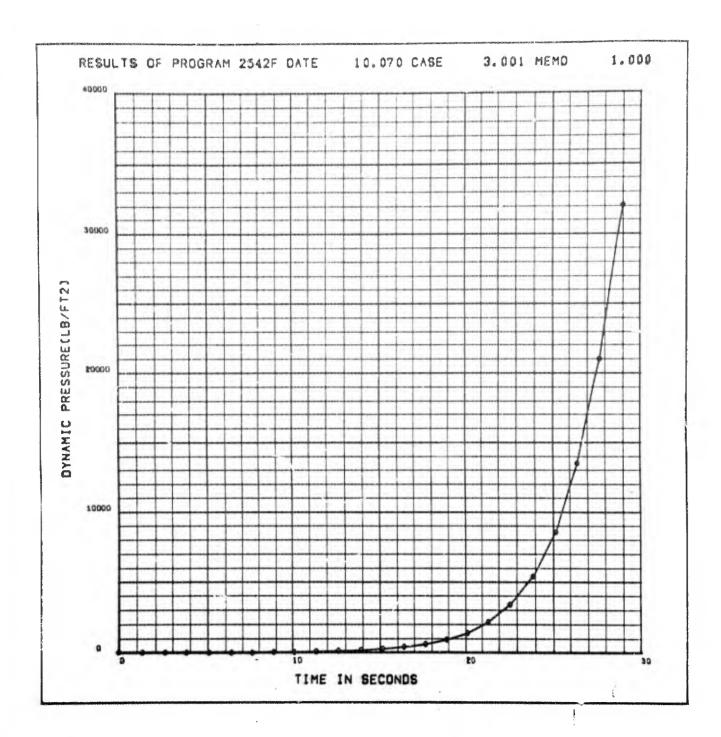
FRAME 8



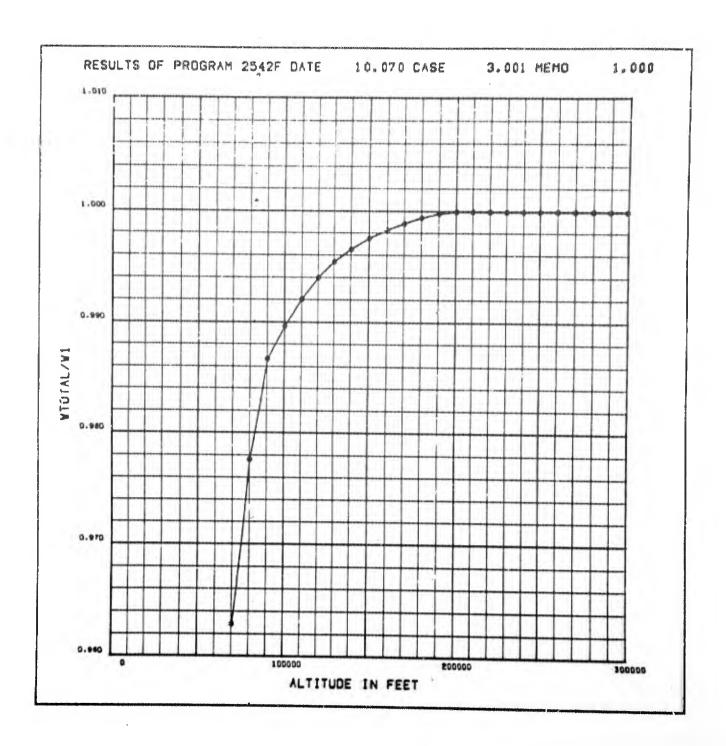
FRAME 9



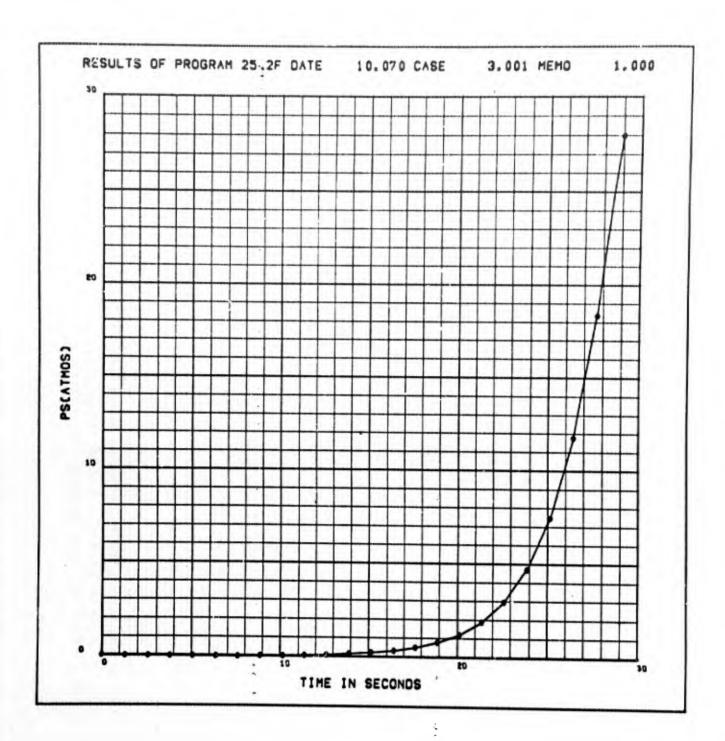
FRAME 10



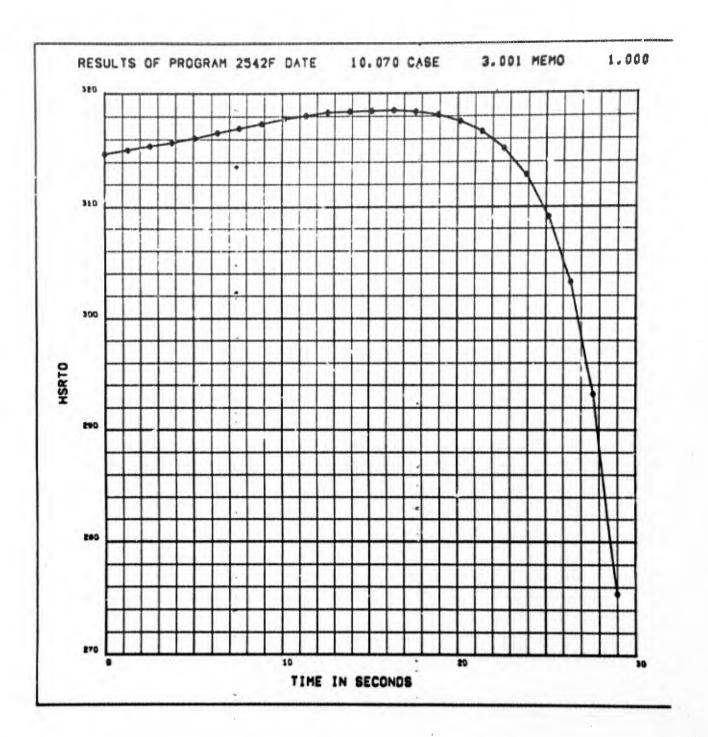
FRAME 11



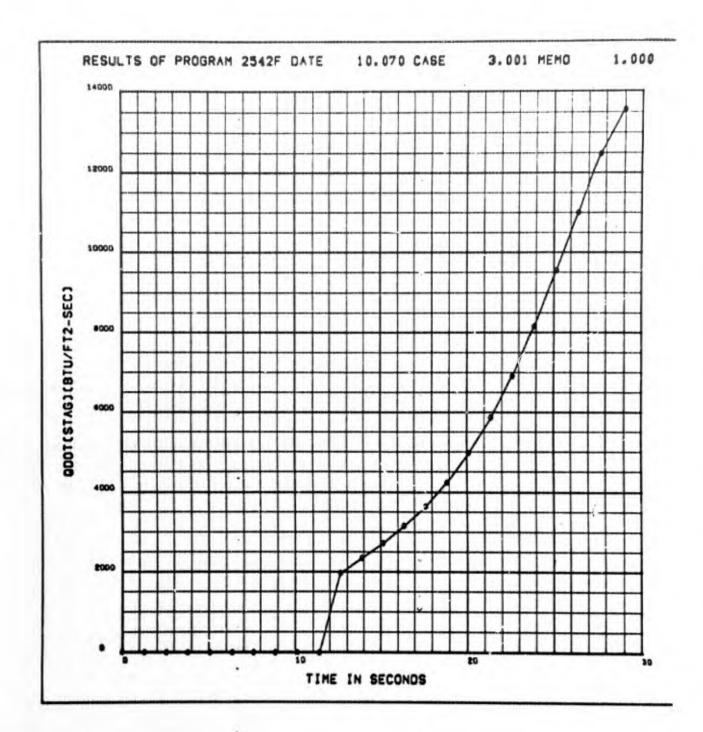
FRAME 12



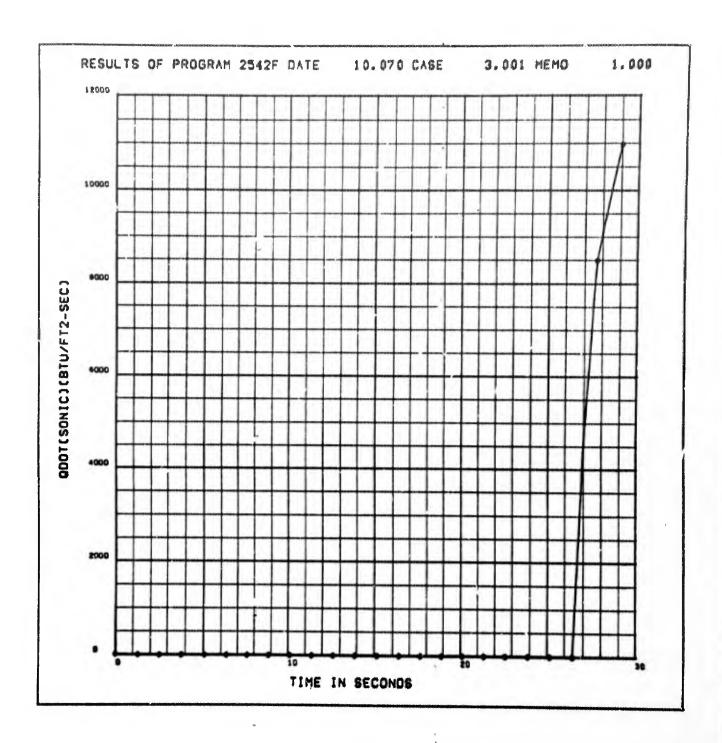
FRAME 13



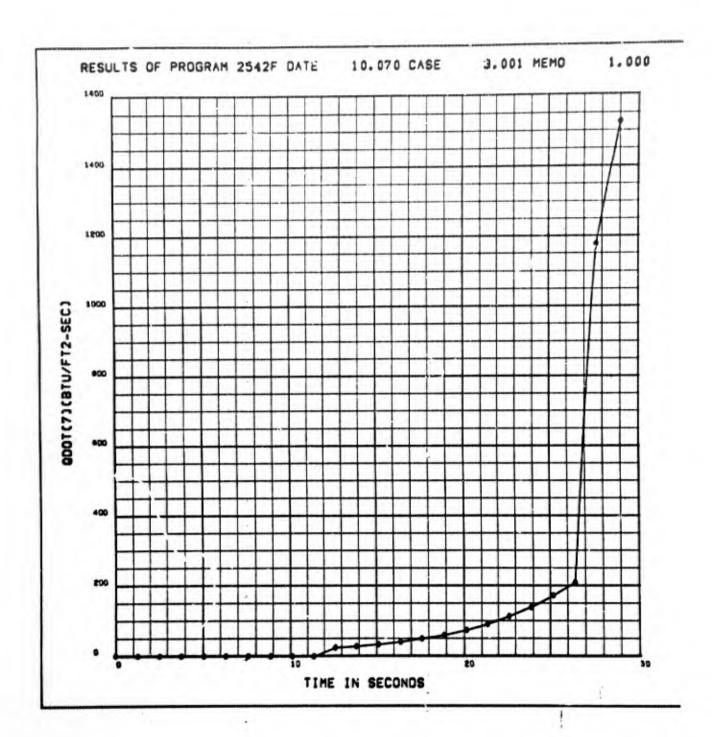
FRAME 14



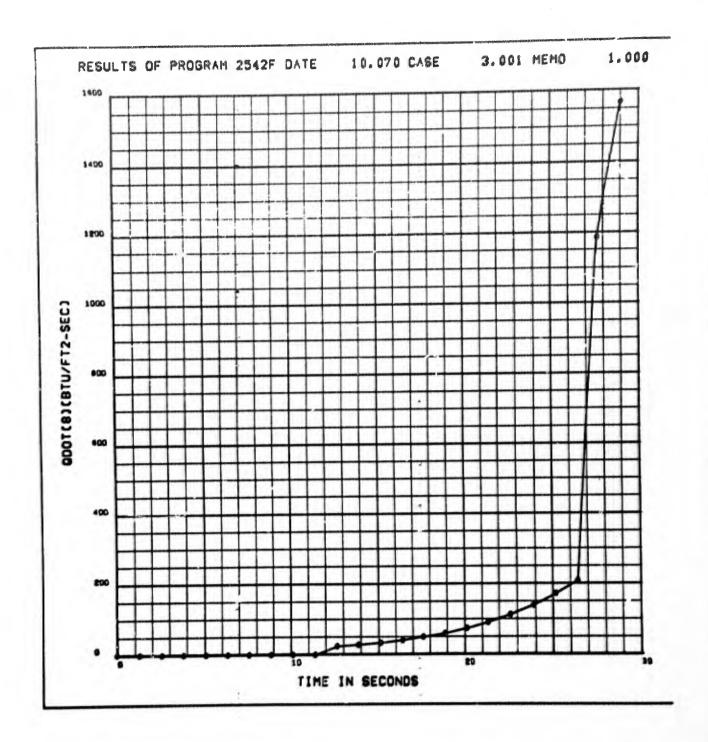
FRAME 15



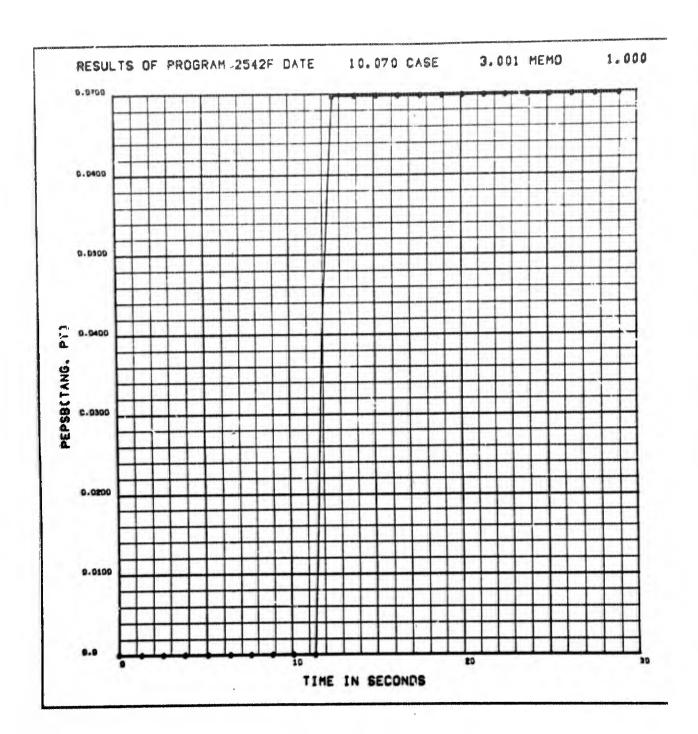
FRAME 16



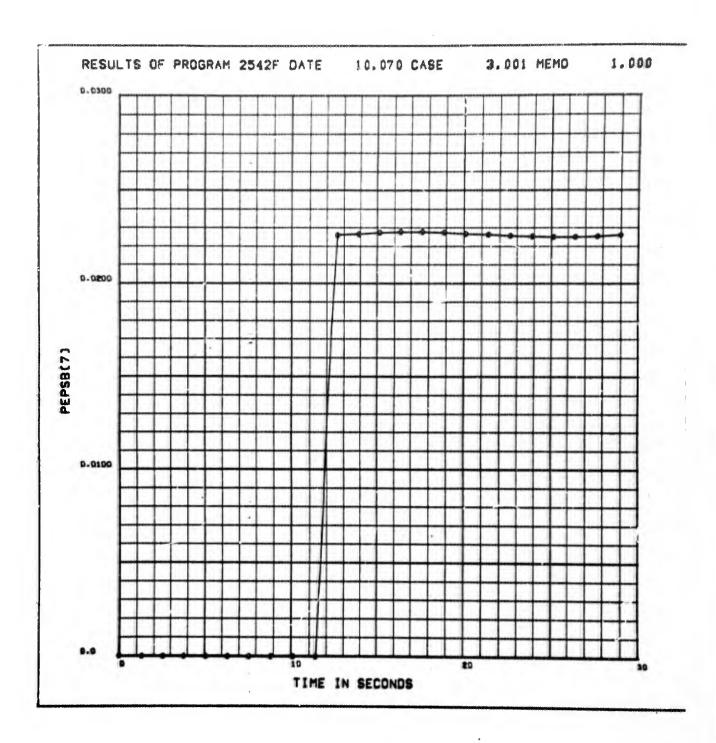
FRAME 17



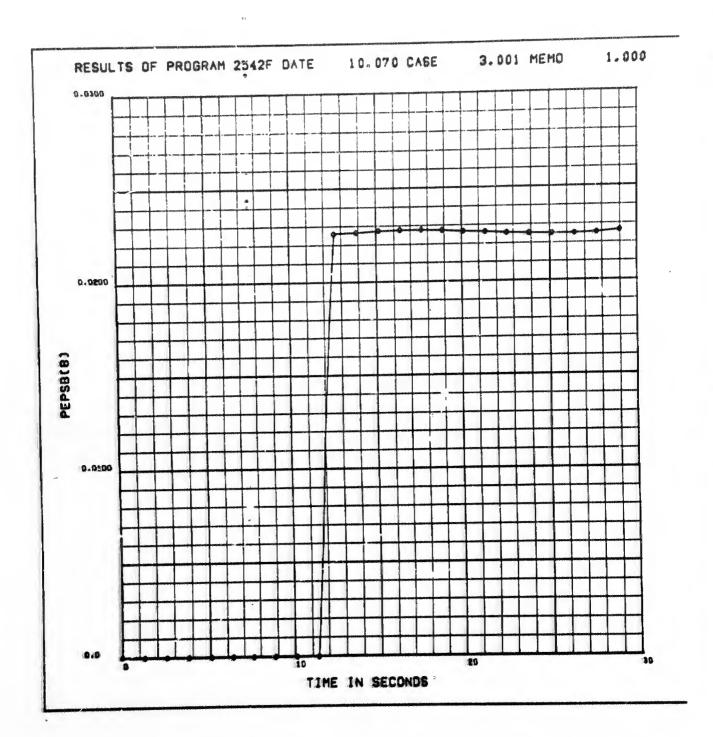
FRAME 18



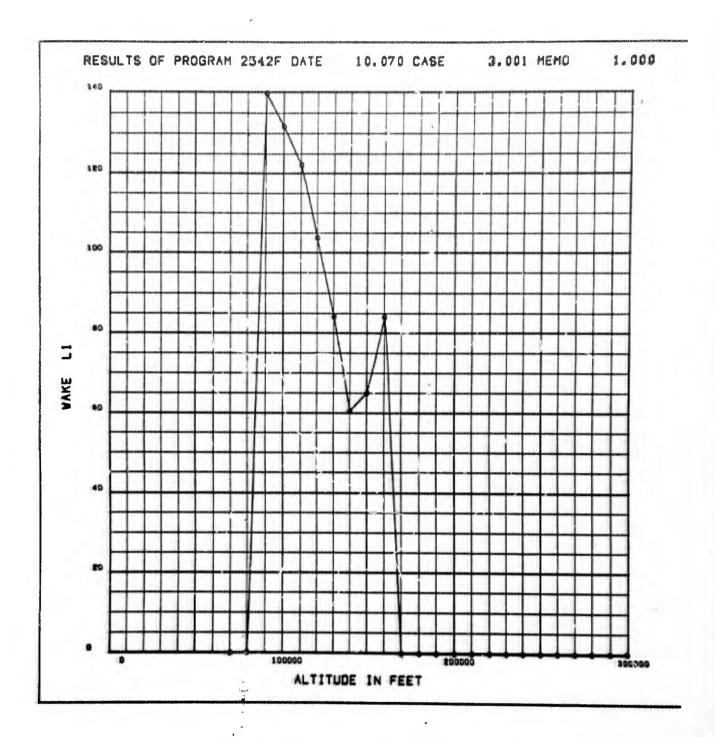
FRAME 19



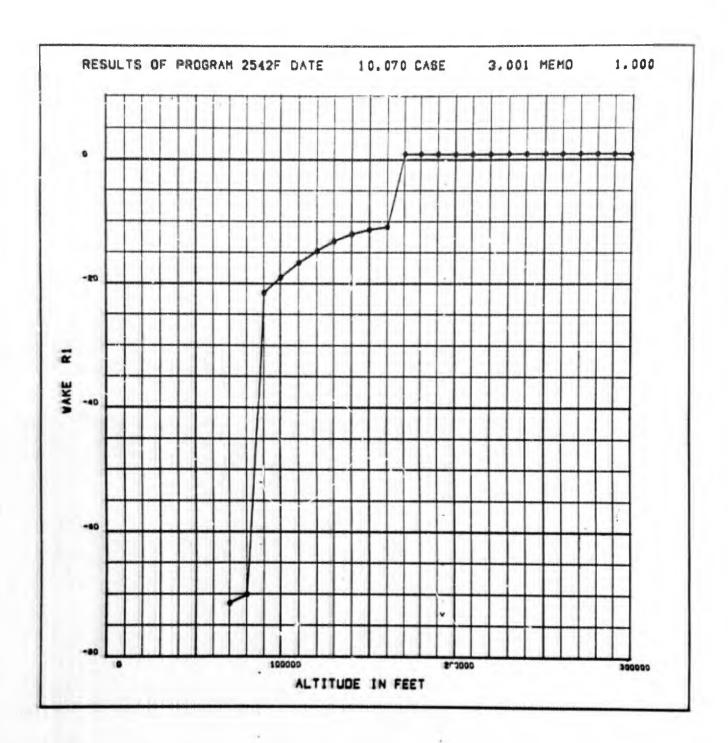
FRAME 20



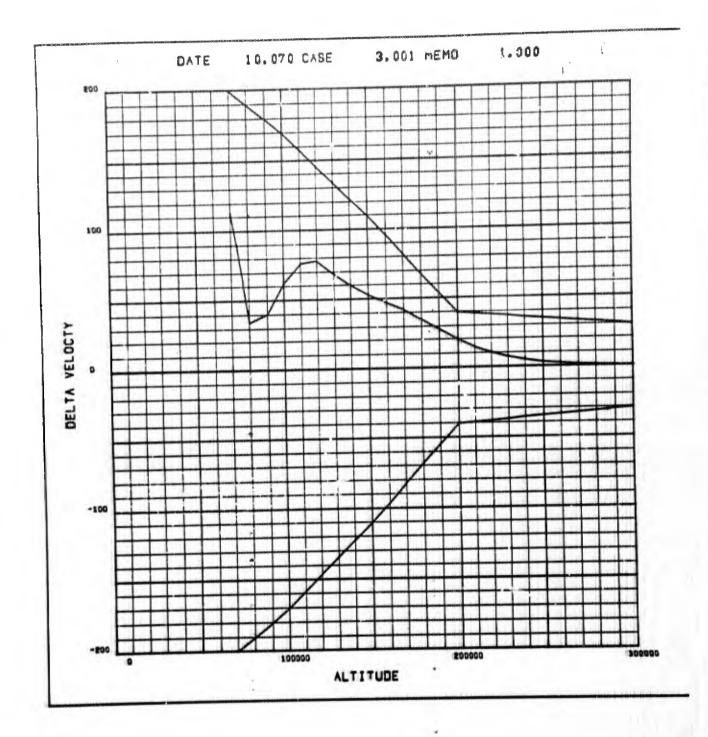
FRAME 21



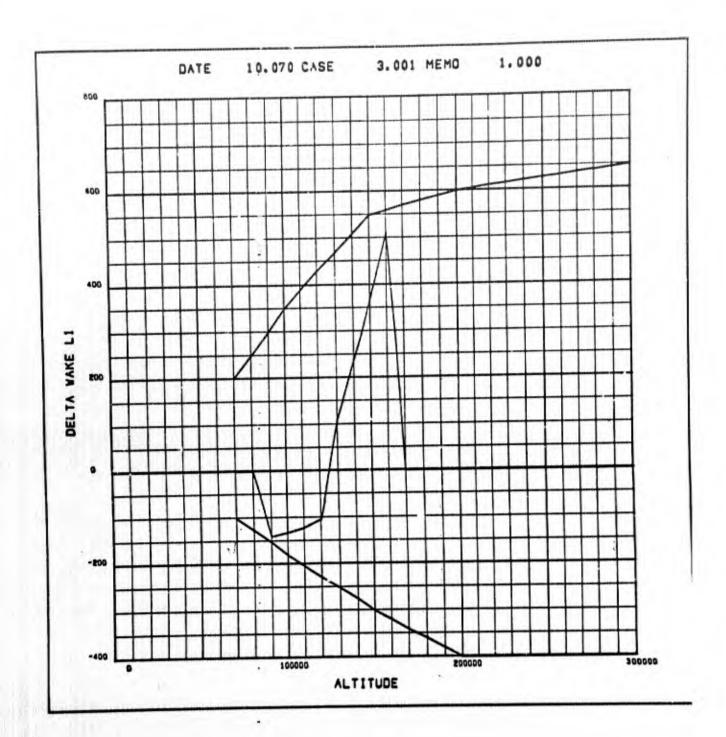
FRAME 22



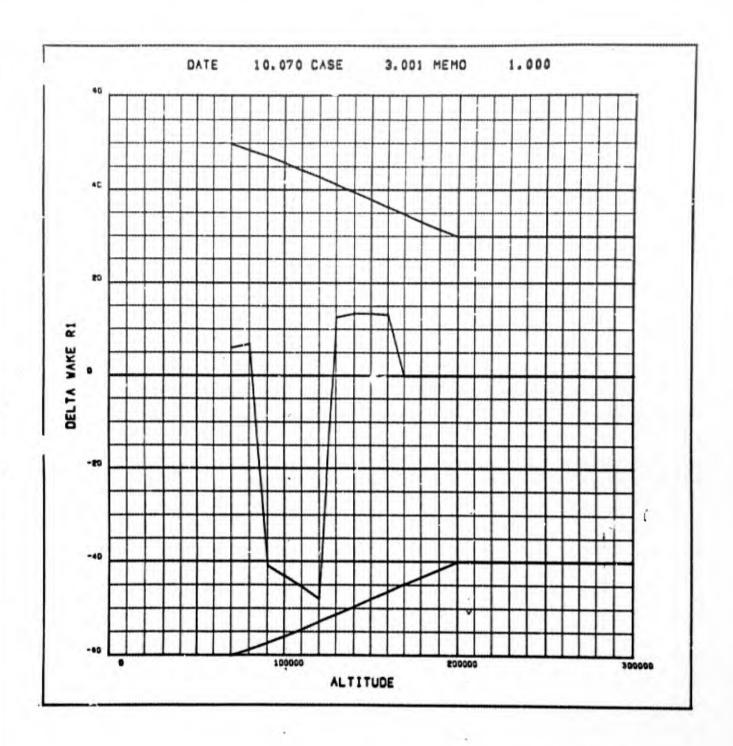
FRAME 23



FRAME 24



FRAME 25



FRAME 26

